

Instability

Solution conservation laws may be unobservable due to instabilities. Small disturbances invariably present.

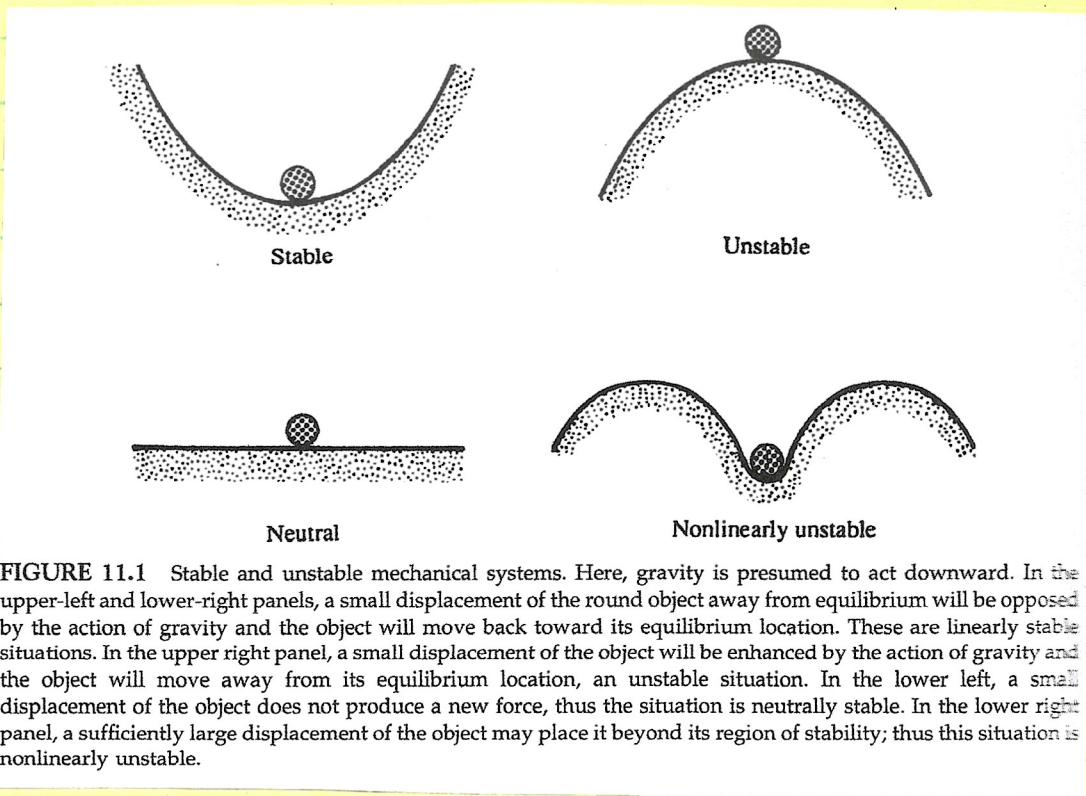


FIGURE 11.1 Stable and unstable mechanical systems. Here, gravity is presumed to act downward. In the upper-left and lower-right panels, a small displacement of the round object away from equilibrium will be opposed by the action of gravity and the object will move back toward its equilibrium location. These are linearly stable situations. In the upper right panel, a small displacement of the object will be enhanced by the action of gravity and the object will move away from its equilibrium location, an unstable situation. In the lower left, a small displacement of the object does not produce a new force, thus the situation is neutrally stable. In the lower right panel, a sufficiently large displacement of the object may place it beyond its region of stability; thus this situation is nonlinearly unstable.

Smooth laminar flows are only stable to small disturbances when certain conditions are satisfied: e.g., channel flow $Re < Re_{crit}$ or stratified shear flow $Ri < Ri_{crit}$. Otherwise, disturbances grow and change character of the original flow. Sometimes, disturbances grow to finite amplitude and reach a new steady state equilibrium, which may also become unstable and evolve to another steady state equilibrium condition and so on.

In the limit, flow becomes superposition of a variety of interacting disturbances with random phases is turbulent flow.

Issues are: (1) process of laminar to turbulent transition; & (2) prediction transition

Linear stability theory:

- (1) infinitesimal disturbances imposed on particular flow to determine grow or decay
- (2) perturbations assumed small enough that linearization acceptable, i.e., quadratic & higher order terms neglected (of perturbation variables & their derivatives).
- (3) limited initial behavior i.e. does not predict transition just instability
- (4) real flow may be stable linear theory but unstable larger (non linear) disturbances

In spite limitations many successes:

- (1) Thermal convection in layer fluid
- (2) Tollmien-Schlichting waves in viscous BL
- (3) Taylor vortices

all confirmed experiments

Method of Normal Modes

Sinusoidal disturbances imposed on basic state (background, initial, or equilibrium state).

Assume basic state parallel x or $f(z)$ only
i.e. $\underline{U} = U(z)\hat{e}$

Real part
to obtain
physical
quantities

$$u = (x, y, z, t) = \hat{u}(z) \exp(i\alpha x + imy + \sigma t) \\ = \hat{u}(z) \exp[ik(\hat{e}_k \cdot \underline{x} - ct)]$$

$\hat{u}(z)$ = complex amplitude

$$\underline{k} = (\alpha, m, 0) \quad \hat{e}_k = \underline{k}/|\underline{k}| \quad k = |\underline{k}| = \text{real} > 0$$

σ = temporal growth rate = $\sigma_r + i\sigma_i$

c = complex wave speed = $c_r + ic_i$

Two forms useful for stationary or
traveling waves, respectively. Behavior
examined for all possible $\underline{k}$. If σ_r or c_i
> 0 for any $\underline{k}$ the system is unstable
at this $\underline{k}$ otherwise stable.

$\sigma_r < 0$ or $c_i < 0$ Stable

$\sigma_r = 0$ or $c_i = 0$ neutrally Stable

$\sigma_r > 0$ or $c_i > 0$ unstable

Each mode accounted separately as
linearly implies modes do not interact.
The method leads to eigenvalue problem.

Boundary between stable & unstable called marginal state. Two types:

(1) $\sigma_i = c_r = 0 \Rightarrow$ stationary motion i.e. cellular convection / secondary flow

(2) $\sigma_i \neq 0$ or $c_r \neq 0 \Rightarrow$ transverse oscillations of growing amplitude
called overstability or oscillatory mode

Neutral stability & marginal state both have $\sigma_r = c_i = 0$; however, marginal state also is border line between stable & unstable.

In many cases, stability criterion found by simply setting σ_r or $c_i = 0$ without proof that represent border line condition.

Sinusoidal transverse waves

$$y(x,t) = A \cos \left[\frac{2\pi}{\lambda} (x - ct) \right]$$

x = horizontal

z = vertical = $y(x,t)$

A = amplitude $\begin{matrix} + \text{ crest} \\ - \text{ trough} \end{matrix}$

= wave form

λ = wave length

k = wave number rad/m

c = phase speed

$$2\pi(x - ct)/\lambda = \text{phase}$$

$$\lambda = \text{distance between crests} \quad T = 2\pi/\omega = \lambda/c$$

= period i.e. Δt

f = cyclic frequency = T^{-1} Hz between passages

$\omega = 2\pi f$ = radian frequency rad/s crests

$$y(x,t) = A \cos(kx - \omega t) \quad \omega = 2\pi \times \frac{kc}{2\pi} = kc$$

Speed wave crests $y = A \Rightarrow$ phase = $2n\pi$

n = integer

$$\frac{2\pi}{\lambda}(x_{\text{crest}} - ct) = 2n\pi = kx_{\text{crest}} - \omega t$$

$$x_{\text{crest}} = (\omega/k)t + 2n\pi/k \quad \Delta x_{\text{crest}} = (\omega/k)\Delta t$$

$$\lambda = (\omega/k)T$$

$$= (\omega/k) \times 1/c$$

phase speed specifies the

travel speed of constant-phase

wave features such as troughs

or crests.

$$c = \omega/k$$

$$= 2\pi f / \frac{2\pi}{\lambda}$$

$$= \lambda f$$

3D generalization

$$\gamma = a \cos(lx + ly + mz - \omega t) = a \cos(\underline{k} \cdot \underline{r} - \omega t)$$

$$c = \omega/k \quad \underline{c} = (\omega/k) \hat{\underline{k}}$$

$$\hat{\underline{k}} = \underline{k} / |\underline{k}|$$

$$= \underline{k} / k$$

$$\underline{k} = (l, l, m)$$

$$|\underline{k}|^2 = l^2 + l^2 + m^2$$

$$\lambda = 2\pi/k$$

trace velocity:

$$c_x = \omega/l$$

$$c_y = \omega/l$$

$$c_z = \omega/m$$

all $> c = \omega/k$
as when $\neq 0$
all smaller k

thus c_x, c_y, c_z

not vector

components

but reflect that constant phase surfaces appear to travel faster along directions not coming $\underline{k}$

If waves exist in fluid moving $\underline{U} \Rightarrow \underline{c}_0 = \underline{c} + \underline{U} \quad \underline{c} \cdot \underline{k} =$

$$\underline{c}_0 \cdot \underline{k} \Rightarrow \omega_0 = \omega + \underline{U} \cdot \underline{k} \quad \omega_0 = \text{observed} \quad \frac{\omega}{k} \times \frac{k^2}{k} = \omega$$

$\omega = \text{intrinsic (measured moving with the flow)}$

Doppler shifted

Say $\omega = 0$ but flow

hor. periodic x with $\lambda = 2\pi/k$ at wave at speed U then frequency at point is $\omega_0 = U k$.

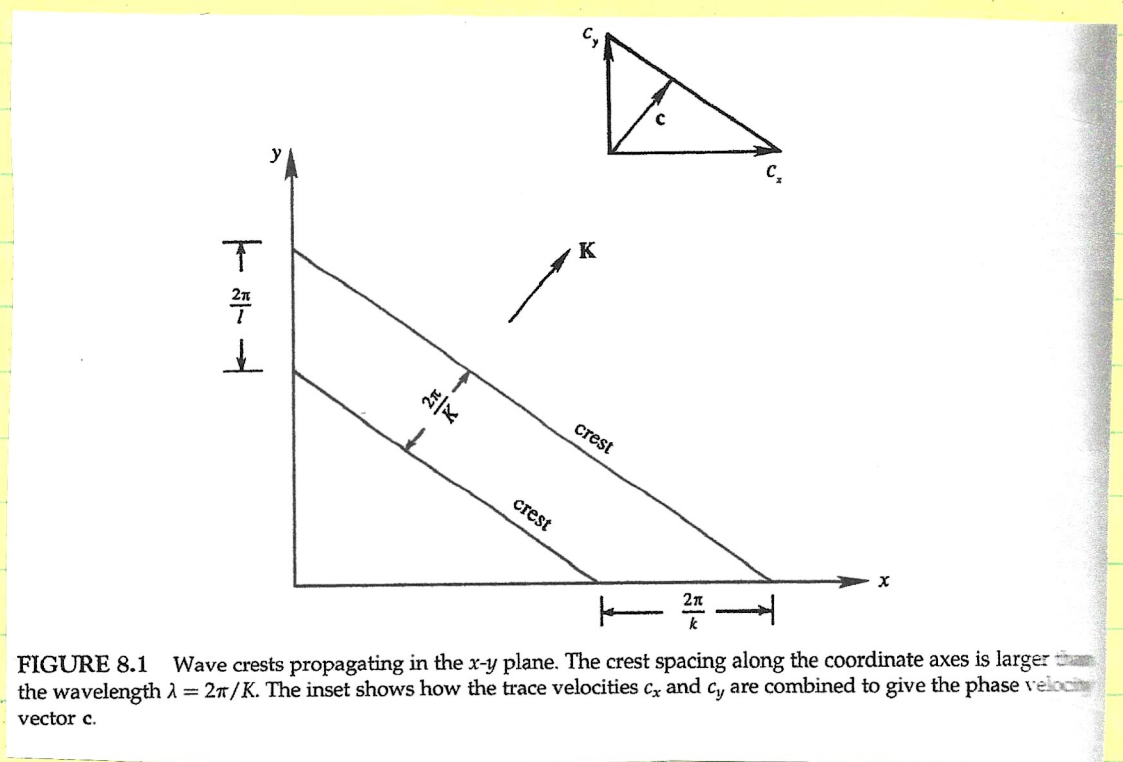


FIGURE 8.1 Wave crests propagating in the x - y plane. The crest spacing along the coordinate axes is larger than the wavelength $\lambda = 2\pi/k$. The inset shows how the trace velocities c_x and c_y are combined to give the phase velocity vector $\underline{c}$.