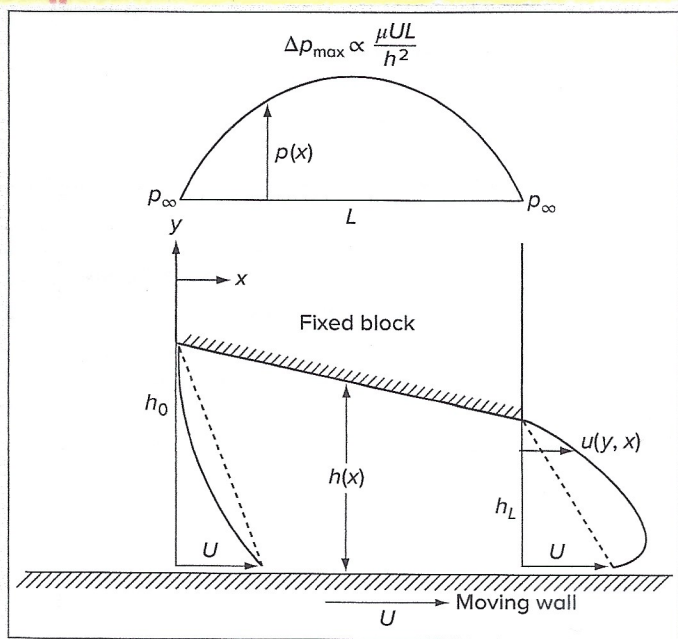


Lubrication Theory (also applicable porous media, filtration, adhesion, biological, ^{manufacturing} flow)

Lubrication is friction reduction of two bodies in near contact is generally accomplished by a viscous fluid moving through a narrow but variable gap between the two bodies with one or both moving (Reynolds, 1886)



Assume $2D \frac{\partial^2}{\partial z^2} = 0$ & Stokes flow such that inertia negligible
 $\epsilon \mu u_x \ll \mu u_y$
 $\epsilon \frac{\partial^2 u}{\partial z^2} \ll \mu \left(\frac{U}{h}\right)^2$
 $\frac{\epsilon UL}{\mu} \left(\frac{h}{L}\right)^2 \ll 1$ Re can be large $U/h/L$

$U = 10 \text{ m/s}, L = 4 \text{ cm}, h = 0.1 \text{ mm}$ Small
 SAE 50 lubricating oil $\nu = 2.7 \times 10^{-4} \text{ m}^2/\text{s}$
 $Re_L = 570$ but $Re_L \left(\frac{h}{L}\right)^2 = 0.004 \ll 1$ OK
 FIGURE 3-48 Low Reynolds number Couette flow in a varying gap: To maintain continuity, the gap pressure rises to a maximum and superimposes Poiseuille flow toward both ends of the gap.

$$\epsilon = h/L \ll 1 \text{ at } Re \text{ moderate}$$

Like Stokes flow since Re fairly small and inertia negligible. Usually quasi-steady.

Like BL flow in that $\frac{\partial p}{\partial y} = 0$ & $\frac{\partial p}{\partial x}$ important

However, the pressure & viscous stress scale differently than Stokes or BL flow.

Reynolds Equation for Beary Theory

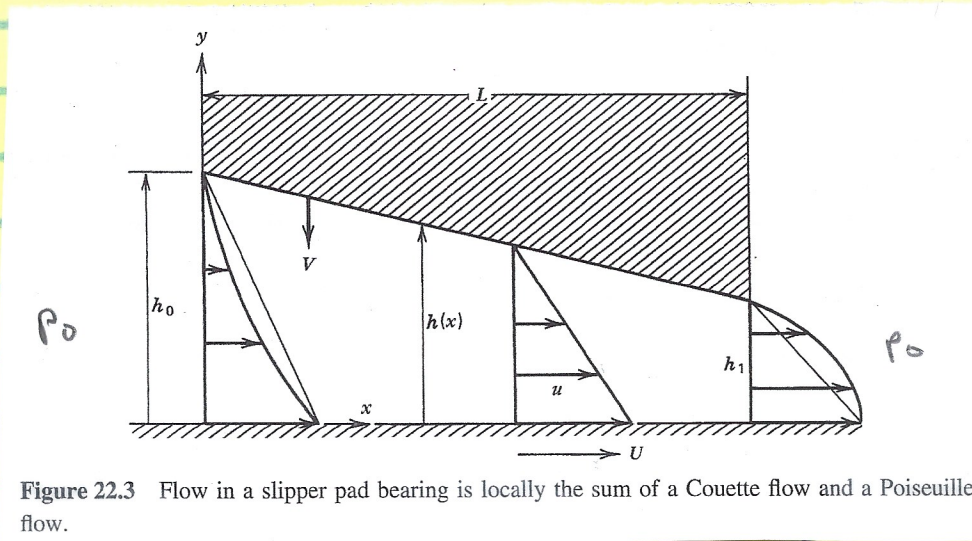


Figure 22.3 Flow in a slider bearing is locally the sum of a Couette flow and a Poiseuille flow.

Events:

1. Moving wall sweeps fluid into narrowing passage due to viscous shear forces, which induces Couette flow $u = Vy/h$ and $Q_c = \frac{1}{2}Vh$
2. Continuity requires $Q = \text{const}$; therefore, p_x required that induces Poiseuille u component that redistributes fluid & maintains $Q = \text{const}$

In general $h(x, t)$ and for simplicity upper wall vertical motion only V with mag be $v(t)$, lower wall $U = \text{const}$, and allow for w due p_z

$$x, z \sim L \quad y \sim h_0 \quad \epsilon \sim h_0/L \quad u, w \sim U \quad v \sim \frac{U h_0}{L} \quad p \sim \frac{\mu U L}{h_0^2}$$

$$y\text{-momentum} \quad \lim_{h_0/L \rightarrow 0} \frac{\partial p}{\partial y} = 0 \Rightarrow \frac{\partial p}{\partial y} = 0 \text{ ie } p = p(x, z)$$

x- and z-momentum equations yield quasi-steady equations (See Appendix)

$$0 = -p_x + \mu u_{yy}$$

$$0 = -p_z + \mu w_{yy}$$

$$u(y=0) = 0 \quad v(y=0) = 0 \quad w(y=0) = 0$$

$$u(y=h) = 0 \quad v(y=h) = 0 \quad w(y=h) = 0$$

Partial integration over y of uppermost BC:

$$u = \frac{1}{2\mu} p_x (y^2 - y^2) + (1 - y/h) 0$$

$$w = \frac{1}{2\mu} p_z (y^2 - y^2)$$

Combination Couette & Poiseuille flow
function pressure gradient, h, at 0

Need to determine pressure distribution
that will support the load with
the bearing.

The Reynolds equation for pressure
is derived by integrating the
continuity equation over the y-direction.

$$\frac{d}{dt} \int_{A(t)} f(x,t) dx = \int_A \frac{\partial f}{\partial t} dx + \frac{dh}{dt} f(x,t) - \frac{dh}{dt} f(x,t)$$

$$\int_0^h u_x dy + \int_0^h \omega_z dy = - \int_0^h u_y dy = -V = + \frac{\partial h}{\partial t}$$

$$\frac{\partial}{\partial x} \int_0^h u dy = \int_0^h \frac{\partial u}{\partial x} dy + \frac{dh}{\partial x} u(y=h) - \frac{dh}{\partial x} u(y=0) = \int_0^h \frac{\partial u}{\partial x} dy$$

$$\bar{u} = Q_x/h$$

$$\bar{\omega} = Q_z/h$$

$$\frac{\partial}{\partial z} \int_0^h \omega dy = \int_0^h \frac{\partial \omega}{\partial z} dy + \frac{dh}{\partial z} \omega(y=h) - \frac{dh}{\partial z} \omega(y=0) = \int_0^h \frac{\partial \omega}{\partial z} dy$$

$$u(y) = \frac{1}{2\mu} P_x (y^2 - yh) + (1 - y/h) \sigma$$

$$= a (y^2 - yh) + \sigma - \sigma y/h$$

$$\int_0^h [a(y^2 - yh) + \sigma - \sigma y/h] dy$$

$$= \left[\frac{a y^3}{3} - \frac{a h y^2}{2} + \sigma y - \frac{\sigma y^2}{2h} \right]_0^h = \frac{a h^3}{3} - \frac{a h^3}{2} + \sigma h - \frac{\sigma h}{2}$$

$$= -\frac{a h^3}{6} + \frac{\sigma h}{2}$$

$$\omega(y) = \frac{1}{2\mu} P_z (y^2 - yh) = b (y^2 - yh)$$

$$b \int_0^h (y^2 - yh) dy = b \left[\frac{y^3}{3} - \frac{y^2 h}{2} \right]_0^h = -\frac{b h^3}{6}$$

$$\frac{\partial}{\partial x} \left[-\frac{a h^3}{6} + \frac{\sigma h}{2} \right] + \frac{\partial}{\partial z} \left[-\frac{b h^3}{6} \right] = -\frac{\partial h}{\partial t}$$

$$\frac{\partial}{\partial x} [-a h^3 + 3\sigma h] + \frac{\partial}{\partial z} [-b h^3] = -6 \frac{\partial h}{\partial t}$$

$$\frac{\partial}{\partial x} \left[-\frac{1}{2\mu} P_x h^3 + 3\sigma h \right] + \frac{\partial}{\partial z} \left[\frac{1}{2\mu} P_z h^3 \right] = -6 \frac{\partial h}{\partial t}$$

$$\frac{1}{\mu} \left[\frac{\partial}{\partial x} (h^3 \frac{\partial P}{\partial x}) + \frac{\partial}{\partial z} (h^3 \frac{\partial P}{\partial z}) \right] = 6\sigma \frac{\partial h}{\partial x} + 12 \frac{\partial h}{\partial t}$$

Reynolds equation lubrication in channel

$h(x, t)$ with lower wall moving speed U .

$p(x, z, t)$ found as f (geometry motion walls).

Once pressure known proportion Poiseuille or Couette in the velocity profiles (u, w) .

Slipper Bearing

Non-dimensional variables:

$$p^* = \frac{p - p_0}{\mu U L / h_0^2} \quad x^* = x/L \quad h^* = \frac{h}{h_0} = 1 - Ax^*$$

$$\frac{\mu U L}{h_0^2} p^* + p_0 = p$$

$$A = \frac{\alpha L}{h_0} = \frac{h_0 - h_1}{h_0}$$

Assume 1D flow, i.e., $\frac{\partial z}{\partial x} = 0$

$$w = 0$$

$$\alpha = \frac{h_0 - h_1}{L}$$

$$\frac{\partial z}{\partial t} = 0 \quad \text{i.e. } v = 0$$

$$\frac{\partial}{\partial x} \left(h^3 \frac{\partial p}{\partial x} \right) = 6 \sigma \mu \frac{\partial h}{\partial x}$$

$$h^3 \frac{\partial p}{\partial x} = 6 \sigma \mu h + C$$

$$C = -6 \sigma \mu h_m$$

where $h_m = h \left(\frac{\partial p}{\partial x} = 0 \right)$

i.e. where Poiseuille flow = 0

$$h^3 \frac{\partial p}{\partial x} = 6 \sigma \mu (h - h_m)$$

$$h^3 = h_0^3 h^{*3}$$

$$p_x^* \frac{h_0^3 h^{*3}}{h_0^2} \frac{\mu U L}{h_0^2} = 6 \sigma \mu (h_0 h^* - h_0 h_m^*)$$

$$\frac{\partial p}{\partial x} = \frac{\partial x^*}{\partial x} \frac{\partial p}{\partial p_0} \frac{\partial p_0}{\partial x^*}$$

$$h^{*3} p_x^* = 6 (h^* - h_m^*)$$

$$= \frac{1}{L} \times \frac{\mu U L}{h_0^2} p_x^*$$

Rate at h_m^* $u = (1 - y/h)U = \frac{\sigma}{h} (h - y)$

$$h_m^* = \frac{2Q}{\sigma h_0} = Q^*$$

$$Q = \int_0^{h_m} u dy = \frac{\sigma}{h} \left[h y - \frac{y^2}{2} \right]_0^{h_m} = \frac{\sigma h_m}{2}$$

$$dp^* = \frac{G}{h^{*3}} (h^* - h_m^*) dx^*$$

$$= G (h^{*-2} - h_m^* h^{*-3}) dx^*$$

$$dp^* = G \left[(1 - Ax^*)^{-2} - h_m^* (1 - Ax^*)^{-3} \right] dx^*$$

$$x = 1 - Ax^* = h^*$$

$$dx = -A dx^*$$

$$= \frac{-G}{A} \left[x^{-2} - h_m^* x^{-3} \right] dx$$

$$\int x^n dx = \frac{x^{n+1}}{n+1} \quad n \neq -1$$

$$p^* = \frac{-G}{A} \left[x^{-1} / -1 - h_m^* \frac{x^{-2}}{-2} \right] + C$$

$$= 6A^{-1} \left(+h^{*-1} - \frac{h_m^*}{2} h^{*-2} \right) + C$$

$$h_0 h^* = h = h_0 (1 - Ax^*) = h_0 \left[1 - \frac{h_0 - h_1}{h_0} \frac{x}{L} \right]$$

$$h(0) = h_0$$

$$h = h_0 - (h_0 - h_1) \frac{x}{L}$$

$$h(L) = h_1$$

$$h^* = 1 = h = h_0 \quad \wedge \quad p = p_0 \quad \overset{0}{=} \quad p^* = 0$$

$$0 = 6A^{-1} \left(+1 + h_m^* / 2 \right) + C$$

$$C = 6A^{-1} \left(-1 + h_m^* / 2 \right)$$

$$p^* = 6A^{-1} \left[1 - h^{*-1} + \frac{h_m^*}{2} (-h^{*-2} + 1) \right]$$

$$= 6A^{-1} (h^{*-1} - 1) - 3A^{-1} h_m^* (h^{*-2} - 1)$$

$$p^+ = 0 \quad \text{at} \quad h^+ = 1 - A \quad \text{ie} \quad h = h_0 - (h_0 - h_1) \frac{x}{L}$$

$$0 = 6A^{-1} \left(\frac{1}{a} - 1 \right) - 3A^{-1} h_m^+ \left(\frac{1}{a^2} - 1 \right) \quad h(L) = h_1 \quad x^+ = 1$$

$$\frac{3h_m^+}{A} \left(\frac{1-a^2}{a^2} \right) = \frac{6}{A} \left(\frac{1-a}{a} \right)$$

$$h^+ = 1 - A, \quad p = p_0, \quad \gamma^+ = 0$$

$$h_m^+ (1-x)(1+a) = 2(1-x)a^2$$

$$(1-a)(1+a) =$$

$$h_m^+ = \frac{2a^2}{1+a}$$

$$1+a+a-a^2 = 1-a^2$$

$$(1-A)(1-A) = 1-2A+A^2$$

$$= \frac{2(1-A)}{1+1-A} = \frac{2(1-A)}{2-A}$$

$$p^+ = 6A^{-1} \left(\frac{1-h^+}{h^+} \right) - 3A^{-1} h_m^+ \left(\frac{1-h^{+2}}{h^{+2}} \right)$$

$$= 6A^{-1} h^+ (1-h^+) - 3A^{-1} h_m^+ h^{+2} (1-h^{+2})$$

$$h^+ = 1 - Ax^+$$

$$\frac{h^+ - h^{+2}}{A}$$

$$= 6A^{-1} h^+ \left[(1-h^+) - \frac{h_m^+ h^+}{2} (1+h^{+2}) \right]$$

$$\frac{1-h^+}{h^+ A} = \frac{Ax^+}{Ah^+}$$

$$= 6A^{-1} h^+ (1-h^+) \left[1 - \frac{h_m^+ h^+}{2} (1+h^{+2}) \right]$$

$$= \frac{x^+}{1-Ax^+}$$

$$p^+(x^+) = \frac{6x^+}{1-Ax^+} \left[1 - \frac{1-A}{2-A} \times \frac{2-Ax^+}{1-Ax^+} \right]$$

$$h^+ (1+h^{+2}) = h^+ + h^{+2}$$

$$h_m^+ = 1 - Ax_m^+ = 2 \frac{1-A}{2-A}$$

$$= \frac{1+h^+}{h^+} = \frac{2-Ax^+}{1-Ax^+}$$

$$h^+ = 1 - Ax^+ \quad \Rightarrow \quad h^{+3} \frac{p^+}{x^+} = \epsilon (h^+ - h_m^+)$$

but $h_m^+ = 2 \frac{1-A}{2-A}$ from 1st integration

$$\infty \quad 2 \frac{1-A}{2-A} = 1 - Ax_m^+$$

equation $p^+ = 0$ when $h^+ = h_m^+$

$$Ax_m^+ = 1 - 2 \frac{1-A}{2-A} = \frac{2-A-2(1-A)}{2-A} = \frac{2-A-2+2A}{2-A}$$

$$= \frac{A}{2-A} \quad \text{ie} \quad x_m^+ = \frac{1}{2-A}$$

$$1 - Ax_m^+ = 1 - \frac{A}{2-A} = \frac{2-A-A}{2-A} = \frac{2(1-A)}{2-A}$$

$$2 - Ax_m^+ = 2 - \frac{A}{2-A} = \frac{4-2A-A}{2-A} = \frac{4-3A}{2-A}$$

$$p_m^+ = \frac{6}{(2-A) 2(1-A)} \left[1 - \frac{1-A}{2-A} \frac{4-3A}{2-A} \frac{2-A}{2(1-A)} \right]$$

$$p_m^+ = \frac{3}{(1-A)} \left[1 - \frac{4-3A}{2(2-A)} \right] = \frac{3}{(1-A)} \left[\frac{2(2-A) - 4 + 3A}{2(2-A)} \right]$$

$4 - 2A - 4 + 3A = A$

$$p_m^+ = \frac{3A}{2(1-A)(2-A)}$$

$A=0$ walls //
 Couette flow $p = \text{const} = p_0$

$A > 0$ but very small

$$p_m^* = \frac{3A}{4(1-\frac{A}{2})(1-A)}$$

$$= \frac{3A}{4} (1-\frac{A}{2})^{-1} (1-A)^{-1}$$

$$= \frac{3A}{4} (1+\frac{A}{2})(1+A) \approx \frac{3A}{4}$$

$$x_m^* = \frac{1}{2} (1-\frac{A}{2})^{-1} = \frac{1}{2} (1+\frac{A}{2}) \approx \frac{1}{2}$$

$$\frac{dp^*}{dx^*} > 0 \quad x^* < x_m^* \quad \wedge \quad \frac{dp^*}{dx^*} < 0 \quad x^* > x_m^*$$

Poiseuille flow opposes Couette flow & vice versa

Process: fluid dragged into converging channel via viscous shear force piles up to create high pressure $x^* = \frac{1}{2}$ after which $\frac{dp^*}{dx^*}$ changes sign at pressure flow towards exit. $\frac{dp^*}{dx^*}$ between center at either end induces Poiseuille flow towards both ends of the bearing: subtracts Couette flow first half & adds second half

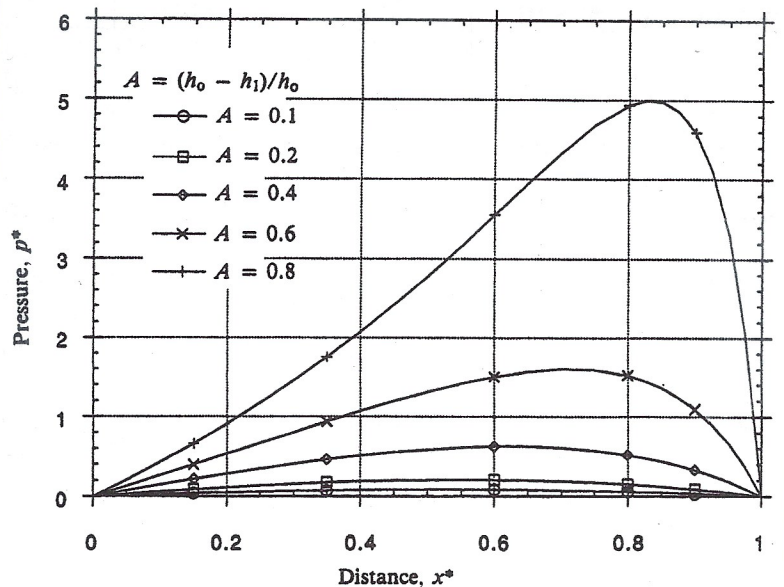


Figure 22.4 Pressure in a slipper pad bearing.

$$\frac{p_m - p_o}{\mu U L / h_o^2} = \frac{3V}{4} + O[A^2]$$

$$p_m - p_o = \frac{\mu U L}{h_o^2} \times \frac{3V}{4}$$

$$= \frac{3\mu U L}{4} \left(\frac{h_o - h_i}{h_o^3} \right) = \frac{3\mu U L}{4} \frac{1 - h_i/h_o}{h_o^2}$$

Shows importance h_o

SAE oil with $U = 10 \text{ m/s}$, $L = 4 \text{ cm}$, $h_o = .1 \text{ mm}$

$p_m - p_o$ of order $\frac{\mu U L}{h_o^2} \approx 2.5 \times 10^7 \text{ Pa}$ or 250 atm

ie very high force to slipper block, which enables it to support large load without block touching well.

Stokes flow is linear and so reversible.

If reverse wall motion $U < 0$ then $\Delta p < 0$ ie will cavitate and form vapor void in gap (G.I. Taylor film low Re hydrodynamics) so flow in expanding narrow gap may not support large loads & provide good lubrication. Issue for rotating journal bearing where gap contracts/expands as often leads to partial cavitation.

Squeeze film lubrication: viscous adhesion

wringing together smooth surfaces relatively

crankshaft bearing

power stroke piston causes $V(t)$

which dominates over hydrodynamic

journal-bearing effect, whereas

separating smooth surfaces by

pulling normal to increase gap

height is difficult, although

sliding is easy

Assume $V = -\frac{\partial h}{\partial t}$ and $U = 0$ Also ω and $\frac{\partial^2 h}{\partial x^2} = 0$

Take $x=0$ such that bearing part ends at $\pm L/2$

Reynolds pressure equation: $\frac{1}{\mu} \frac{d}{dx} \left(h^3 \frac{dp}{dx} \right) = 12 \frac{\partial h}{\partial t} = 12V(t)$

$$\frac{dp}{dx} = \frac{12\mu}{h^3} \frac{dh}{dt} x$$

$$u = \frac{1}{2\mu} \frac{\partial p}{\partial x} (y^2 - y^2) \text{ at } x=0 = 0$$

$$p - p_0 = \frac{12\mu}{h^3} \frac{\partial h}{\partial t} \frac{x^2}{2} \Big|_{-L/2}^x$$

at maximum at $x = \pm L/2$
Thus all flow into

in out gap must cross at ends

$$= \frac{12\mu}{h^3} \frac{\partial h}{\partial t} \left[\frac{x^2}{2} - \frac{L^2/4}{2} \right] = \frac{12\mu}{h^3} \frac{\partial h}{\partial t} \left[\frac{x^2}{2} - \frac{L^2}{8} \right] = \frac{12\mu}{h^3} \frac{\partial h}{\partial t} \frac{1}{2} \frac{L^2}{4} \left(\frac{x^2}{L^2} - 1 \right)$$

$$p - p_0 = -\frac{3\mu L^2}{2h^3} \frac{\partial h}{\partial t} \left[1 - \left(\frac{x}{L/2} \right)^2 \right]$$

$p \propto h^{-3}$

very large pressures

$$= \frac{3\mu L^2}{2h^3} V \left[1 - \left(\frac{x}{L/2} \right)^2 \right]$$

$$W = \int_{-L/2}^{L/2} (p - p_0) dx \quad \text{load capacity per unit span}$$

$$= \frac{3\mu L^2}{2h^3} \int_{-L/2}^{L/2} \left[1 - \left(\frac{x}{L/2} \right)^2 \right] dx$$

$$x' = x/L/2$$

$$dx' = dx/L/2$$

$$\int_{-1}^1 (1 - x'^2) dx' = \left(x' - \frac{x'^3}{3} \right) \Big|_{-1}^1 = \frac{2}{3} - \left(-1 + \frac{1}{3} \right) = \frac{4}{3} \times \frac{L}{2}$$

$$L/2 dx' = dx$$

$$-2/3 = \frac{2L}{3}$$

$$W = \frac{3\mu L^2}{2h^3} \int_{-1}^1 (1 - x'^2) dx' \times \frac{2L}{2} = \mu \left(\frac{L}{h} \right)^3 \int_{-1}^1 (1 - x'^2) dx'$$

$$= -\mu \left(\frac{L}{h} \right)^3 \frac{dh}{dt}$$

$$\mu = \frac{\eta s}{m^2} \times \frac{m}{s} = \frac{\eta}{m}$$

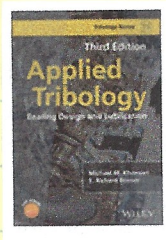
$$\int_{t_1}^{t_2} dt = -\frac{\mu L^3}{W} \int_{h_1}^{h_2} \frac{dh}{h^3}$$

$$\int h^{-3} dh = -\frac{1}{2} h^{-2}$$

$$\Delta t = -\frac{\mu L^3}{W} \left(-\frac{1}{2} h^{-2} \right) \Big|_{h_1}^{h_2}$$

$$\Delta t = \frac{\mu L^3}{2W} \left(\frac{1}{h_2^2} - \frac{1}{h_1^2} \right)$$

$\Delta t =$ time of approach for film gap to reduce from h_2 to h_1 . $\Delta t \rightarrow \infty$ as $h_2 \rightarrow 0$ i.e. takes infinite time squeeze out all the fluid!



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9.4 Generalization for planar squeeze film

For planar squeeze-film problems, the time of approach has the following form (Moore, 1993):

$$\Delta t = K \frac{\mu A^2}{W} \left(\frac{1}{h_2^2} - \frac{1}{h_1^2} \right) \quad (9.13)$$

where K is the shape function. Taking the time derivative of Equation (9.13), the surface approach velocity can be obtained:

$$V_s = \frac{1}{2K} \frac{h^3 W}{\mu A^2} \quad (9.14)$$

where A is the plate area and h is the squeeze-film thickness.

Forms of the function K for a series of planar squeeze-film geometries are shown in Table 9.1 and plotted in Figure 9.5 for convenience (Khonsari and Jang, 1997). Using either the table or the figure, one can readily evaluate K . Then, for a given load, W , Equation (9.13) gives the time of approach for the film thickness to drop from an initial h_1 to a final h_2 .

Example 9.1 Estimate the time of approach in a wet clutch system modeled as two rigid concentric annuli with $R_1 = 0.047$ m and $R_2 = 0.06$ m submerged in a lubricant with viscosity of $\mu = 0.006$ Pa s. Hydraulic pressure is $P = 1.25$ MPa, and the initial separation gap is $h_1 = 25 \times 10^{-6}$ m. Estimate the initial squeeze velocity and the length of time necessary for the film thickness to drop to $h_2 = 5 \times 10^{-6}$ m.

Table 9.1 Types of planar squeeze

Type	Ratio, r	Configuration	Constant, K
Circular section	—		$\frac{3}{4\pi}$
Elliptical section	$\mathfrak{R} = \frac{b}{a}$		$\frac{3\mathfrak{R}}{2\pi(1+\mathfrak{R}^2)}$
Rectangular section	$\mathfrak{R} = \frac{B}{L}$		$\frac{1}{2\mathfrak{R}} \left[1 - \frac{192}{\pi^5 \mathfrak{R}} \sum_{n=1,3,5,\dots} \frac{\tanh(n\pi\mathfrak{R}/2)}{n^5} \right]$
Triangular section	—		$\frac{\sqrt{3}}{10}$
Circular sector	$\mathfrak{R} = \frac{\alpha}{2\pi}$		$\sum_{n=1,3,5,\dots} \frac{24}{n^2 \pi^3 \mathfrak{R} \left[2 + \left(\frac{n}{2\mathfrak{R}} \right)^2 \right]}$
Concentric annulus	$\mathfrak{R} = \frac{D_i}{D_o}$		$\frac{3}{4\pi} \left[\frac{\ln \mathfrak{R} - \mathfrak{R}^4 \ln \mathfrak{R} + (1 - \mathfrak{R}^2)^2}{(1 - \mathfrak{R}^2)^2 \ln \mathfrak{R}} \right]$

The shape factor K for the concentric annulus can be evaluated from Table 9.1:

$$A = \pi (R_2^2 - R_1^2) = 0.0044 \text{ m}^2$$

$$\mathfrak{R} = \frac{R_1}{R_2} = \frac{0.047}{0.06} = 0.783$$

$$K = \frac{3}{4\pi} \frac{\ln \mathfrak{R} - \mathfrak{R}^4 \ln \mathfrak{R} + (1 - \mathfrak{R}^2)^2}{(1 - \mathfrak{R}^2)^2 \ln \mathfrak{R}}$$

$$= \frac{3}{4\pi} \frac{\ln(0.783) - (0.783)^4 \ln(0.783) + (1 - 0.783^2)^2}{(1 - 0.783^2)^2 \ln(0.783)} = 0.019$$

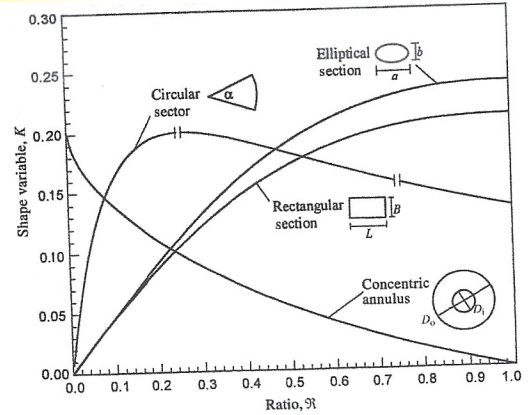


Figure 9.5 Variation of constant K with shape ratio \mathfrak{R}

Using Equation (9.13), the time of approach is

$$\Delta t = K \frac{\mu A^2}{W} \left(\frac{1}{h_2^2} - \frac{1}{h_1^2} \right) = K \frac{\mu A}{P} \left(\frac{1}{h_2^2} - \frac{1}{h_1^2} \right)$$

$$= \frac{0.019(0.006)(0.0044)}{1.25 \times 10^6} \left[\frac{1}{(5 \times 10^{-6})^2} - \frac{1}{(25 \times 10^{-6})^2} \right] = 0.02 \text{ s}$$

The squeeze velocity can be estimated using Equation (9.14):

$$V_s = \frac{1}{2K} \frac{h_1^3 W}{\mu A^2} = \frac{1}{2K} \frac{h_1^3 P}{\mu A}$$

$$= \frac{1}{2(0.019)} \frac{(25 \times 10^{-6})^3 1.25 \times 10^6}{(0.006)(0.0044)} = 0.02 \text{ m/s}$$

The time of approach predicted above is a representation of the first stage of the engagement duration while the clutch operates in the hydrodynamic regime. This engagement begins when pressure is applied hydraulically by means of a piston and hydrodynamic pressure is developed in the ATF as a result of squeeze action which supports most of the applied load. Since the surfaces are separated by a relatively thick film of fluid, behavior of the clutch is governed by the theory of hydrodynamic lubrication. This period lasts only 0.02 s.

During engagement, the fluid film thickness drops to the extent that surface asperities come into contact. As a result, contact pressure at the asperity level begins to support a major portion of the imposed load, significantly influencing the behavior of the wet

clutch. The film thickness is further reduced as the friction-lining material is compressed and deforms elastically. The surfaces are subsequently pressed together and 'locked' when their relative speed drops to zero. The timescale of the engagement process is typically of the order of 1 s, during which the squeeze action is of paramount importance. It follows, therefore, that in a typical engagement cycle, the lubrication regime undergoes a transition from hydrodynamic to mixed or boundary lubrication. This shift in the lubrication regime has important implications on the signature of the total torque, i.e. the combination of the viscous torque and contact torque. The total torque reaches a peak value when the film thickness drops to a minimum. After this peak value, the torque initially remains relatively flat for a short period of time and then begins to rise gradually as the relative speed between the clutch disks decreases. This increase in the torque is a direct consequence of the change in the coefficient of friction as a function of speed, in accordance with the Stribeck friction curve. The most interesting torque signature is a highly undesirable sudden spike or 'rooster's tail' toward the end of the engagement. Its occurrence can be predicted analytically (Jang and Khonsari, 1999) and can be treated to minimize its effect by altering the friction behavior.

Note that the friction material is rough, porous, and deformable. Also, the large disk diameters used in wet clutch systems may require consideration of the centrifugal forces. These elements can affect the engagement time and torque-transfer characteristics, as well as the temperature field in the ATF and on the surface of the separator. The interested reader may refer to Jang and Khonsari (1999) for detailed analysis of automotive wet clutch systems. More recent studies of wet clutches have included parametric analysis of variance and experimental results (Mansouri et al., 2001, 2002; Marklund et al., 2007).

Journal Bearing

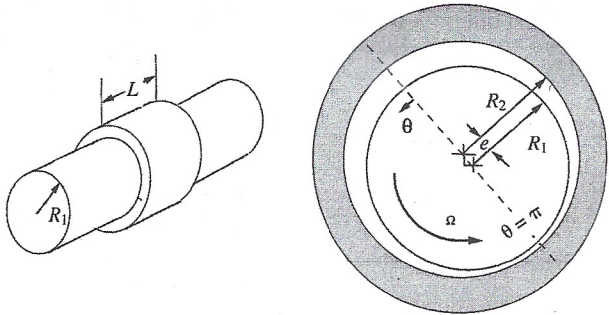


Figure 22.5 Journal bearing.

Rotating shaft, journal, radius R_1 , in bearing housing radius R_2 . $L/R_1 < 1.5 = \text{short}$
 $> 2 = \text{long}$
 aspect ratio

journal offset bearing distance

$e = \text{eccentricity}$

$c = R_2 - R_1 = \text{clearance}$

$\epsilon = e/c = \text{eccentricity ratio with min} = c - e$

$= \text{maximum change film}$ $\text{max} = c + e$

height

with the low such that drops fluid wide to narrow passage, which creates high pressure support journal. Amount of load determines e or ϵ .

Physics similar double slider. Convey / diverge pumps similar double slider with reversed motion expand half of since lubrication equations reversible solution is rel of pressure around $p = -p$ however large $-p$ implies cavitation. Hence, p around constant or other approximations.

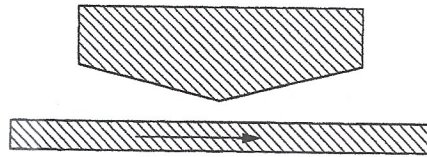
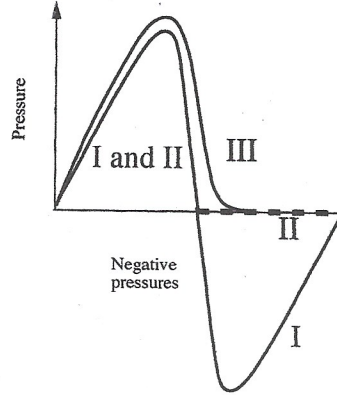


Figure 22.6 Similarity of journal bearing and double slider block. Pressure distribution I, full Sommerfeld; pressure distribution II, half Sommerfeld; pressure distribution III, Swift-Steiber.

Appendix

$$u_x + v_y + w_z = 0$$

$$\rho_t + u u_x + v v_y + w w_z = -\rho_x / \rho + \nu (u_{xx} + v_{yy} + w_{zz})$$

$$\rho_t + u u_x + v v_y + w w_z = -\rho_y / \rho + \nu (u_{xx} + v_{yy} + w_{zz})$$

$$\rho_t + u u_x + v v_y + w w_z = -\rho_z / \rho + \nu (u_{xx} + v_{yy} + w_{zz})$$

$$(x^*, z^*) = (x, z) / L \quad y^* = y / h_0 \quad t^* = \sigma t / L$$

$$(u^*, w^*) = (u, w) / U \quad v^* = v L / \sigma h_0 \quad p^* = p / \rho_0$$

$$\frac{\partial}{\partial t} = \frac{\partial}{\partial t^*} \frac{\partial t^*}{\partial t} = \frac{\sigma}{L} \frac{\partial}{\partial t^*} \quad \frac{\partial}{\partial x} = \frac{\partial}{\partial x^*} \frac{\partial x^*}{\partial x} = \frac{1}{L} \frac{\partial}{\partial x^*} \quad \frac{\partial}{\partial z} = \frac{1}{L} \frac{\partial}{\partial z^*} \quad \frac{\partial}{\partial y} = \frac{1}{h_0} \frac{\partial}{\partial y^*}$$

$$(u, w) = U (u^*, w^*) \quad v = \frac{\sigma h_0}{L} v^* \quad \rho = \rho_0 \rho^* \quad z = \frac{L}{\sigma} z^*$$

$$= \sigma z v^* \quad \epsilon = h_0 / L \ll 1$$

$$\frac{1}{L} \frac{\partial}{\partial x^*} \sigma u^* + \frac{1}{h_0} \frac{\partial}{\partial y^*} \frac{\sigma h_0}{L} v^* + \frac{1}{L} \frac{\partial}{\partial z^*} \sigma w^* = 0$$

$$\frac{1}{L} \left(\frac{\partial u^*}{\partial x^*} + \frac{\partial v^*}{\partial y^*} + \frac{\partial w^*}{\partial z^*} \right) = 0 \quad \text{is no change continuity}$$

$$Re_L = \frac{\sigma U L}{\mu}$$

$$\epsilon = h_0 / L$$

$$\epsilon^2 Re_L =$$

$$\frac{h_0^2 \rho U}{L \mu}$$

$$\frac{h_0^2 \rho U}{L \mu}$$

$$\frac{\sigma}{L} \frac{\partial}{\partial z^*} \sigma u^* + \frac{\sigma^2}{L} \left(u^* \frac{\partial u^*}{\partial x^*} \right) + \frac{\sigma h_0}{L} v^* \frac{1}{h_0} \frac{\partial}{\partial y^*} \sigma u^* + \sigma w^* \frac{1}{L} \frac{\partial}{\partial z^*} \sigma u^*$$

$$= -\frac{\rho_0}{\rho L} \frac{\partial p^*}{\partial x^*} + \nu \left(\frac{1}{L^2} \frac{\partial^2}{\partial x^{*2}} \sigma u^* + \frac{1}{h_0^2} \frac{\partial^2}{\partial y^{*2}} \sigma u^* + \frac{1}{L^2} \frac{\partial^2}{\partial z^{*2}} \sigma u^* \right)$$

$$\frac{\sigma^2}{L} \left(\frac{\partial u^*}{\partial z^*} + u^* \frac{\partial u^*}{\partial x^*} + v^* \frac{\partial u^*}{\partial y^*} + w^* \frac{\partial u^*}{\partial z^*} \right) = -\frac{\rho_0}{\rho L} \frac{\partial p^*}{\partial x^*} + \nu \left(\frac{\sigma}{L^2} \frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\sigma}{h_0^2} \frac{\partial^2 u^*}{\partial y^{*2}} + \frac{\sigma}{L^2} \frac{\partial^2 u^*}{\partial z^{*2}} \right)$$

$$\frac{\sigma^2}{L} \frac{\partial u^*}{\partial z^*} = -\frac{\rho_0}{\rho L} \frac{\partial p^*}{\partial x^*} + \mu \frac{\sigma}{L^2} \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{L^2}{h_0^2} \frac{\partial^2 u^*}{\partial y^{*2}} + \frac{\partial^2 u^*}{\partial z^{*2}} \right)$$

$$\Lambda = \frac{\mu \sigma L}{\rho_0 h_0^2}$$

$$\frac{\sigma U L}{\mu} \frac{\partial u^*}{\partial z^*} = -\frac{\rho_0 L}{\mu \sigma} \frac{\partial p^*}{\partial x^*} + \frac{\partial^2 u^*}{\partial x^{*2}} + \epsilon^{-2} \frac{\partial^2 u^*}{\partial y^{*2}} + \frac{\partial^2 u^*}{\partial z^{*2}}$$

$$\epsilon^2 Re_L \frac{\partial u^*}{\partial z^*} = -\frac{\rho_0 L h_0^2}{\mu \sigma L^2} \frac{\partial p^*}{\partial x^*} + \epsilon^2 \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial z^{*2}} \right) + \frac{\partial^2 u^*}{\partial y^{*2}}$$

$$\frac{\rho_0 h_0^2}{\mu \sigma L} = \Lambda^{-1}$$

$$\Lambda = \frac{\mu \sigma L}{\rho_0 h_0^2} = \frac{\text{mass force}}{\text{pressure}} \text{ term \#}$$

$\Sigma \ll 1$ Re_L moderate Δ in order unity

$$\infty \quad 0 = -\frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial y^2} \quad \text{drop } \tau$$

Similarly for $z \quad 0 = -\frac{\partial p}{\partial z} + \mu \frac{\partial^2 w}{\partial z^2}$

$$\frac{\rho \sigma}{L} \frac{\partial}{\partial t} \frac{\partial h_0}{\partial x} + \frac{\rho \sigma u^*}{L} \frac{\partial}{\partial x} \frac{\partial h_0}{\partial x} + \frac{\rho \sigma u^*}{L} \frac{\partial}{\partial x} \frac{\partial h_0}{\partial y} + \frac{\rho \sigma u^*}{L} \frac{\partial}{\partial x} \frac{\partial h_0}{\partial z} + \frac{\rho \sigma w^*}{L} \frac{\partial}{\partial z} \frac{\partial h_0}{\partial x}$$

$$= -\frac{\rho_0}{h_0} \frac{\partial p^*}{\partial y^*} + \nu \left(\frac{1}{L^2} \frac{\partial^2}{\partial x^{*2}} \frac{\partial h_0}{\partial x} + \frac{1}{h_0^2} \frac{\partial^2}{\partial y^{*2}} \frac{\partial h_0}{\partial x} + \frac{1}{L^2} \frac{\partial^2}{\partial z^{*2}} \frac{\partial h_0}{\partial x} \right)$$

$$\frac{\nu^* h_0}{L^2} \left[\frac{\partial^2 u^*}{\partial x^{*2}} + u^* \frac{\partial^2 u^*}{\partial x^{*2}} + w^* \frac{\partial^2 u^*}{\partial y^{*2}} + w^* \frac{\partial^2 u^*}{\partial z^{*2}} \right] = -\frac{\rho_0}{\rho} \frac{\partial p^*}{\partial y^*} + \nu \frac{\sigma}{h_0} \frac{\partial^2 u^*}{\partial y^{*2}} + \nu \frac{\sigma h_0}{L^2} \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial z^{*2}} \right)$$

$$\frac{\rho \sigma L}{\mu} \frac{D u^*}{D t} = -\frac{\rho_0 L^3}{\mu \nu h_0^2} + \frac{L^2}{h_0^2} \frac{\partial^2 u^*}{\partial y^{*2}} + \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial z^{*2}} \right)$$

$$\Sigma^+ Re_L \frac{D u^*}{D t} = \underbrace{-\frac{\rho_0 L^3}{\mu \nu h_0^2}}_{\sim -1} + \Sigma^2 \frac{\partial^2 u^*}{\partial y^{*2}} + \Sigma^4 \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial z^{*2}} \right)$$

$$\frac{-\rho_0 h_0^2}{\mu \nu L} = -1$$

$$\infty \quad 0 = -\frac{\partial p^*}{\partial y^*}$$

note $\Sigma^2 Re_L = 0.001$

t_f external time

Scale $z = \text{period}$

meso-scale oscillation,

$$Rh^2/\mu \sigma \ll 1$$

room temperature 3D-wireframe

with $v = 1 \times 10^{-4} \text{ m/s}$

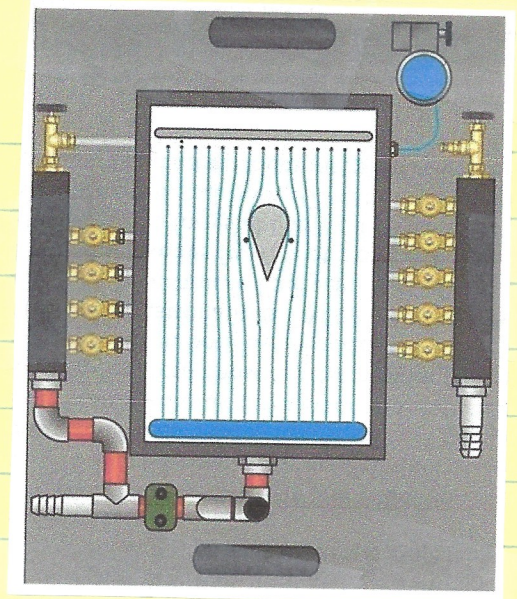
at $h_0 = 1 \text{ mm}$ $L = 25 \text{ cm}$

$\Delta T = 10 \text{ m/s}$

Hele-Shaw Flow

Table/cell = thin jet chamber,
constant h driven by p or g

Originally for study lubrication
by Hele-Shaw inventor
fluid clutch & variable
pitch propeller



2D viscous dominated flow with potential
flow streamlines. Wall shape of obstacle
can be used to visualize ideal flow patterns,
including singularities ie point sources/sinks.

Consider lubrication approximation with x, y
main flow scaled L , $z = \text{width scaled } h$, ϵ
 $\epsilon = h/L \rightarrow 0$.

$$0 = -p_x + \mu u_{zz} \quad 0 = -p_y + \mu v_{yy} \quad 0 = -p_z$$

$$z=0 \quad (u, v, w) = 0$$

$$z=h \quad (u, v, w) = 0$$

$$u = \frac{h^2}{2\mu} p_x \left(\left(\frac{z}{h} \right)^2 - \frac{z}{h} \right)$$

$$v = \frac{h^2}{2\mu} p_y \left(\left(\frac{z}{h} \right)^2 - \frac{z}{h} \right)$$

$$w = 0$$

$$p = p(x, y)$$

$$\chi \neq \chi(z) \quad \wedge \quad \frac{dx}{dy} \Big|_{\chi} = \frac{u}{v} = p_x / p_y = f(x, y)$$

\wedge in z direction χ are
unvarying on top of each other

$$\omega_z = -u_y + v_x = [-p_{xy} + p_{yx}] \frac{h^2}{2\mu} \left[\left(\frac{z}{h}\right)^2 - \frac{z}{h} \right] = 0$$

$$\wedge \text{ continuity } u_x + v_y = 0 \quad \text{ie } \nabla \cdot \underline{v} = 0$$

$$\wedge \underline{v} = \nabla \alpha$$

However, p not governed
Bernoulli equation.

$$\text{Such that } \nabla^2 \alpha = 0 \\ = \text{laminar flow}$$

$$u_x = a p_{xx} \quad \wedge \quad v_y = a p_{yy} \quad a = \frac{h^2}{2\mu} \left[\left(\frac{z}{h}\right)^2 - \frac{z}{h} \right]$$

$$\text{or } p_{xx} + p_{yy} = 0 \quad \text{or per Stokes flow}$$

$$\underline{v} = u \hat{i} + v \hat{j}$$

$$u = a p_x$$

$$v = a p_y$$

$$\nabla p = p_x \hat{i} + p_y \hat{j} = a (u \hat{i} + v \hat{j})$$

$$\text{ie } \nabla p \propto \underline{v} \text{ and aligned } \underline{v}$$

$$\underline{v} \times \nabla p = 0$$

$$(u \hat{i} + v \hat{j}) \times (p_x \hat{i} + p_y \hat{j}) = 0 \quad \text{ie } u p_y - v p_x = 0$$

$$u p_y \hat{k} - v p_x \hat{k} = 0$$

$$a p_x p_y - a p_y p_x = 0 = a (p_x p_y - p_y p_x)$$

$$(u, v)_{\max} \text{ @ } z = h/2$$

$$p_x = -\frac{8\mu}{h^2} u_0 \quad p_y = -\frac{8\mu}{h^2} v_0$$

$$\underline{v}_0 = u_0 \hat{i} + v_0 \hat{j}$$

$$\Delta p = \int (p_x dx + p_y dy) = -\frac{8\mu}{h^2} \left[\int u_0 dx + \int v_0 dy \right]$$

$$p(x, y) - p_0 = -\frac{8\mu}{h^2} \int_{\chi \text{ midplane}} v_0 ds$$

$$\hat{e}_s = dx \hat{i} + dy \hat{j} = ds$$

$$\underline{v}_0 = u_0 \hat{i} + v_0 \hat{j} = v_0 \underline{ds}$$

$$\underline{v}_0 \cdot \underline{ds} = u_0 dx + v_0 dy$$

$$\frac{1}{4} - \frac{1}{2} \\ = -\frac{1}{4}$$

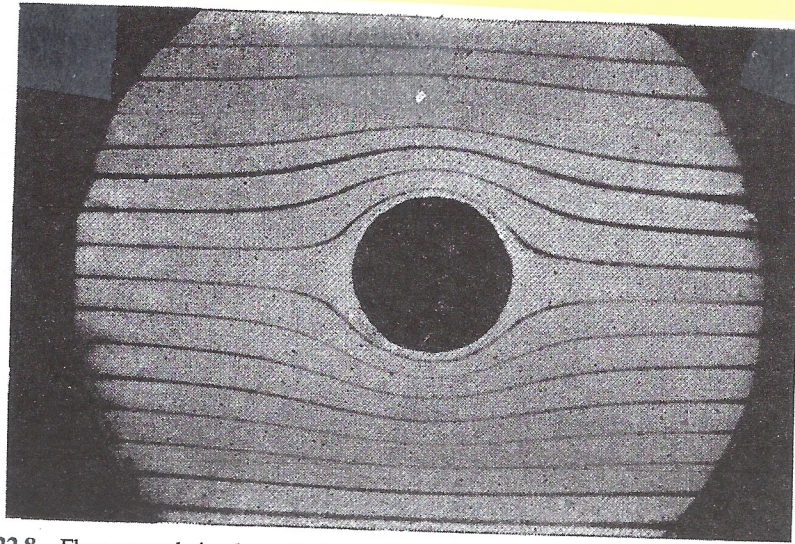


Figure 22.8 Flow around circular cylinder model in a Hele-Shaw apparatus. Streamlines form the potential flow pattern. Photograph courtesy of Professor Bloor, University of Edinburgh, is attributed to Institute of Naval Architects in Bloor (2008).

Example : circular cylinder

potential flow solution $\underline{v} = u_r \hat{e}_r + u_\theta \hat{e}_\theta$

$$u_r = U \left(1 - \frac{r_0^2}{r^2}\right) \cos \theta \quad u_\theta = -U \left(1 + \frac{r_0^2}{r^2}\right) \sin \theta$$

$U =$ midplane velocity far from cylinder along

Stagnation streamline intersects body at $r=R$ & $\theta = \pi$ where $v_0 = u_r$
 \leftarrow far from cylinder

$$p(r, \pi) - p(R, \pi) = -\frac{\rho U^2}{4} \left(R + \frac{r_0^2}{R} - r - \frac{r_0^2}{r} \right)$$

Then from $r=r_0$ to $\theta = \pi$ $v_0 = u_\theta$

$$p(r_0, \theta) - p(r_0, \pi) = +\frac{\rho U^2}{4} (1 - \cos \theta)$$

Along the stagnation point χ & around the cylinder the pressure decreases

$$p(x, y) - p_0 = -\frac{\rho M}{h^2} \int_x^y v_0 ds$$

$$p(r, \pi) - p(R, \pi) = -\frac{\rho M}{h^2} \int_R^r \sigma \left(1 - \frac{v_0^2}{v^2}\right) \cos \theta dr$$

$$p(r, \theta) - p(r, \pi)$$

$$= +\frac{\rho M}{h^2} \int_{\theta=\pi}^{\theta} \sigma \left(1 + \frac{v_0^2}{v^2}\right) \sin \theta d\theta$$

$$= \frac{16\mu U}{h^2} \left(-\cos \theta \right) \Big|_{\pi}^{\theta}$$

$$= +\frac{16\mu U}{h^2} (-\cos \theta + 1)$$

$$= -\frac{16\mu U}{h^2} (\cos \theta - 1)$$

$$= \frac{\rho M \sigma}{h^2} \int_R^r \left(1 - \frac{v_0^2}{v^2}\right) dr$$

$$= \frac{\rho M U}{h^2} \left[r + \frac{v_0^2}{v} \right] \Big|_R^r$$

$$= -\frac{\rho M U}{h^2} \left[R + \frac{v_0^2}{R} - r - \frac{v_0^2}{r} \right]$$

The **Saffman–Taylor instability**, also known as **viscous fingering**, is the formation of patterns in a morphologically unstable interface between two fluids in a porous medium, described mathematically by Philip Saffman and G. I. Taylor in a paper of 1958.^{[1][2]} This situation is most often encountered during drainage processes through media such as soils.^[3] It occurs when a less viscous fluid is injected, displacing a more viscous fluid; in the inverse situation, with the more viscous displacing the other, the interface is stable and no instability is seen. Essentially the same effect occurs driven by gravity (without injection) if the interface is horizontal and separates two fluids of different densities, the heavier one being above the other: this is known as the Rayleigh–Taylor instability. In the rectangular configuration the system evolves until a single finger (the Saffman–Taylor finger) forms, whilst in the radial configuration the pattern grows forming fingers by successive tip-splitting.^[4]



An example of viscous fingering in a Hele-Shaw cell.

Most experimental research on viscous fingering has been performed on Hele-Shaw cells, which consist of two closely spaced, parallel sheets of glass containing a viscous fluid. The two most common set-ups are the channel configuration, in which the less viscous fluid is injected at one end of the channel, and the radial configuration, in which the less viscous fluid is injected at the centre of the cell. Instabilities analogous to viscous fingering can also be self-generated in biological systems.^[5]