

- 5.80 A prototype ship is 35 m long and designed to cruise at 11 m/s (about 21 kn). Its drag is to be simulated by a 1-m-long model pulled in a tow tank. For Froude scaling find (a) the tow speed, (b) the ratio of prototype to model drag, and (c) the ratio of prototype to model power.

$$Fr_m = \frac{V_m}{\sqrt{gL_m}} = Fr_p = \frac{V_p}{\sqrt{gL_p}}$$

$$\alpha = \frac{L_m}{L_p} = \frac{1}{35}$$

$$\frac{V_m}{V_p} = \sqrt{\frac{L_m}{L_p}} = \sqrt{\alpha} \Rightarrow V_m = V_p \sqrt{\alpha} = 1.86 \frac{m}{s}$$

$$\frac{D_m}{\frac{1}{2} \rho V_m^2 A_m} = \frac{D_p}{\frac{1}{2} \rho V_p^2 A_p}$$

$$\frac{D_m}{D_p} = \frac{V_m^2}{V_p^2} \frac{A_m}{A_p} = \alpha \frac{L_m^2}{L_p^2} = \alpha^3 = \frac{1}{35^3} = \frac{1}{42,900}$$

$$\frac{P_m}{P_p} = \frac{D_m}{D_p} \times \frac{V_m}{V_p} = \alpha^3 \sqrt{\alpha}$$

$$= \alpha^{3.5} = \frac{1}{254,000}$$

Solution: Given $\alpha = 1/35$, then Froude scaling determines everything:

$$Fr_m = \frac{V_m^2}{gL_m} = \frac{V_p^2}{gL_p} = Fr_p \text{ or } \frac{V_m}{V_p} = \left(\frac{L_m}{L_p} \right)^{1/2} = \sqrt{\alpha}$$

$$V_{tow} = V_m = V_p \sqrt{\alpha} = 11 \sqrt{35} \approx 1.86 \text{ m/s}$$

$$F_m/F_p = (V_m/V_p)^2 (L_m/L_p)^2 = (\sqrt{\alpha})^2 (\alpha)^2 = \alpha^3 = (1/35)^3 = \frac{1}{42,900} \quad \text{Ans}$$

$$P_m/P_p = (F_m/F_p)(V_m/V_p) = \alpha^3 (\sqrt{\alpha}) = \alpha^{3.5} = 1/35^{3.5} \approx \frac{1}{254,000}$$

Model Ship Testing:

$$C_T(Re, Fr) = C_w(Fr) + C_v(Re) \quad \text{Frond Scale}$$

$$C_{w_p} = C_{w_m} = C_{T_m} - C_{v_m}(1+\epsilon)$$

$$C_{T_p} = C_{w_p} + C_{v_p}(1+\epsilon)$$

C_{T_m} measured in tank tank & $C_v(Re)$
 Based on model-ship correlation line,
 i.e., data for flat plate skin friction
 for $Re_m \leq Re \leq Re_p$. k = form factor,
 which accounts for 3D effects on C_v
 and usually assumed independent Fr
 and Re. Note $C = \text{drag coefficient} = \frac{F}{\frac{1}{2} \rho V^2 S}$

Example : $L_p = 100 \text{ m}$, $V_p = 10 \text{ m/s}$, $S_p = 300 \text{ m}^2$
 $\alpha = 1/25$, $F_{T_m}(V_m) = 60 \text{ N}$

Find V_m using Fr scaling:

$$Fr_m = \frac{V_m}{\sqrt{gL_m}} = Fr_p = \frac{V_p}{\sqrt{gL_p}} = \frac{10}{\underbrace{\sqrt{9.81 \times 100}}_{31.32}} = .32$$

$$V_m = V_p \sqrt{\frac{L_m}{L_p}} = V_p \sqrt{\alpha} \\ = 10 \sqrt{1/25} = 2 \text{ m/s}$$

$$Re_m = \frac{U_m L_m}{V} = \frac{2 \times 100/25}{10^{-6}} = 8 \times 10^6$$

$$V = 10^{-6} \text{ m}^2/\text{s}$$

$$Re_p = \frac{\bar{U}_p L_p}{V} = \frac{10 \times 100}{10^{-6}} = 10^9$$

$$\rho = 1000 \text{ kg/m}^3$$

Use ITTC model-8 hypothesis condition
on flat plate friction line, as per text

$$C_{v_m} = C_v(Re_m) = .003 \quad \text{take } k=0$$

$$C_{v_p} = C_v(Re_p) = .0015$$

$$F_{v_m} = \frac{1}{2} \rho U_m^2 S_m C_{v_m} \quad S_m/S_p = \alpha^2 \\ = \frac{1}{2} \times 1000 \times 2^2 \times (300/25^2) \times .003 \\ = 2.88 \text{ N}$$

$$F_{w_m} = F_{T_m} - F_{v_m} = 60 - 2.88 = 57.12 \text{ N}$$

$$F_{w_p} = \frac{1}{2} \rho U_p^2 S_p \times \frac{F_{w_m}}{\frac{1}{2} \rho U_m^2 S_m} = \alpha^{-3} F_{w_m} \\ = 25^3 F_{w_m} \\ = 8.92 \times 10^5 \text{ N}$$

$$F_{v_p} = \frac{1}{2} \rho U_p^2 S_p C_{v_p} = \frac{1}{2} \times 1000 \times 10^2 \times 300 \times .0015 \\ = .225 \times 10^5 \text{ N}$$

$$F_{T_p} = 9.14 \times 10^5 \text{ N}$$

$$\text{If neglect } C_v(Re): F_{T_p} = \frac{1}{2} \rho U_p^2 S_p \times \frac{F_{T_m}}{\frac{1}{2} \rho U_m^2 S_m} = \alpha^{-3} F_{T_m} = 9.37 \times 10^5 \text{ N}$$

2.5% over estimate!

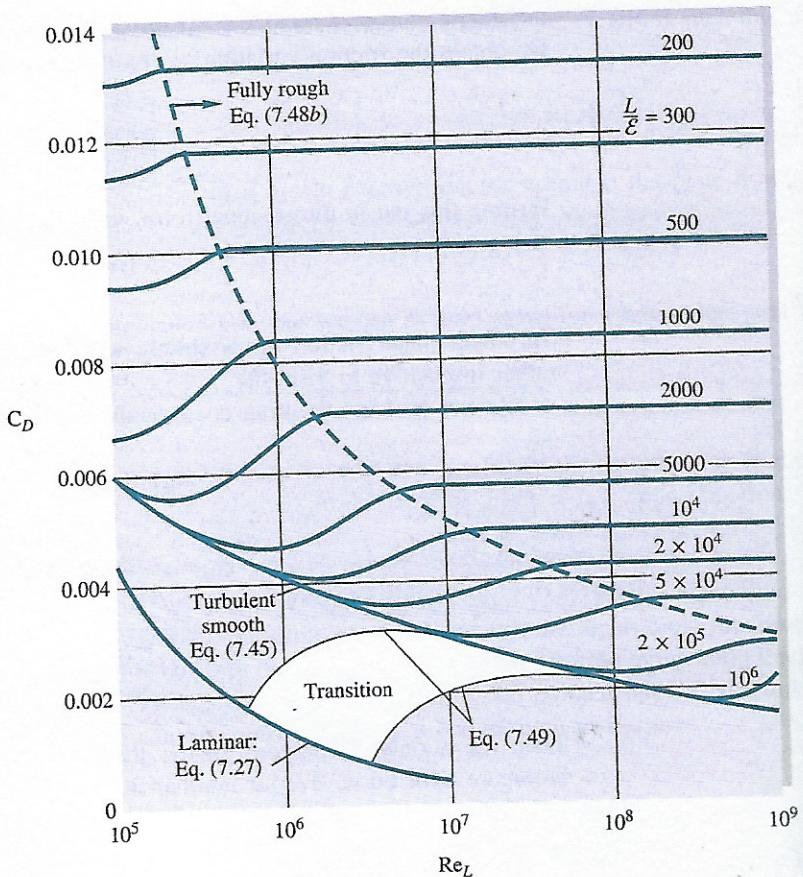


Fig. 7.6 Drag coefficient of laminar and turbulent boundary layers on smooth and rough flat plates. This chart is the flat-plate analog of the Moody diagram of Fig. 6.13.

Equation (7.48b) is plotted to the right of the dashed line in Fig. 7.6. The figure shows the behavior of the drag coefficient in the transition region $5 \times 10^5 < Re_L < 8 \times 10^7$, where the laminar drag at the leading edge is an appreciable fraction of the total drag. Schlichting [1] suggests the following curve fits for these transition curves, depending on the Reynolds number Re_{trans} where transition begins:

$$C_D \approx \begin{cases} \frac{0.031}{Re_L^{1/7}} - \frac{1440}{Re_L} & Re_{trans} = 5 \times 10^5 \\ \frac{0.031}{Re_L^{1/7}} - \frac{8700}{Re_L} & Re_{trans} = 3 \times 10^6 \end{cases}$$

EXAMPLE 7.4

A hydrofoil 1.2 ft long and 6 ft wide is placed in a seawater flow of 40 ft/s, with $\rho = 1.025 \text{ slugs/ft}^3$ and $\nu = 0.000011 \text{ ft}^2/\text{s}$. (a) Estimate the boundary layer thickness at the trailing edge. Estimate the friction drag for (b) turbulent smooth-wall flow from the