

Simulation of Laminar Pipe Flows

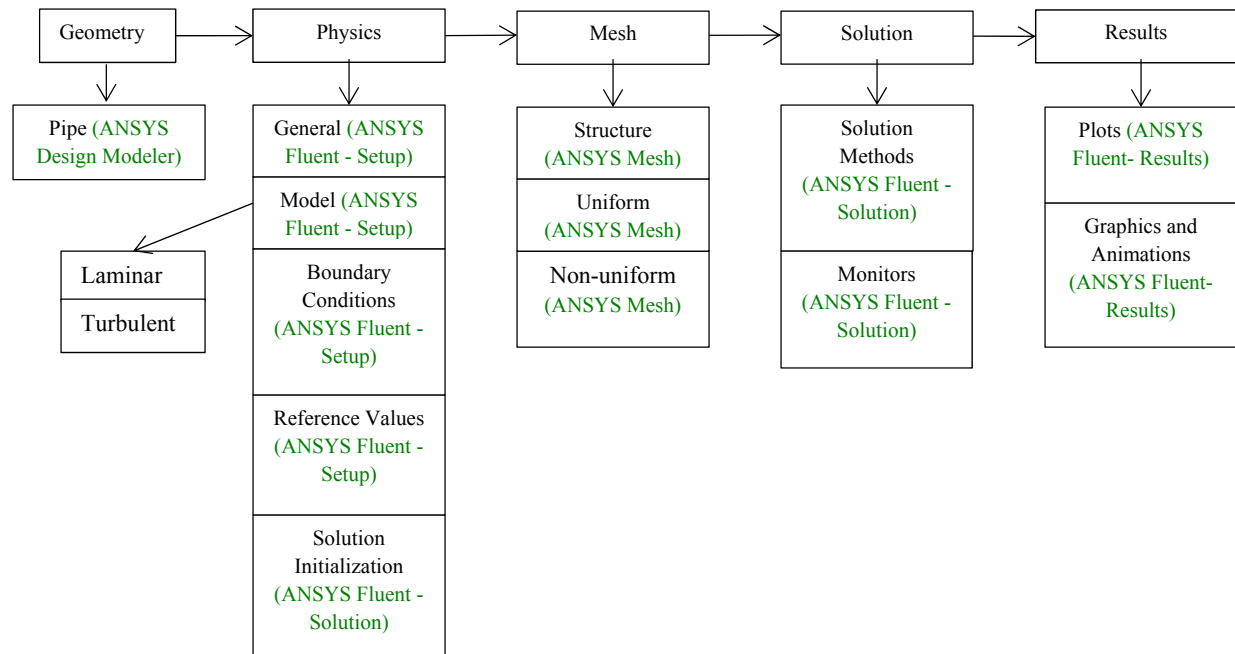
ENGR:2510 Mechanics of Fluids and Transport Processes
CFD PRELAB 1
(ANSYS 17.1; Last Updated: Oct. 7, 2016)

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1. Purpose

The Purpose of CFD PreLab 1 is to teach students how to use the CFD educational interface (ANSYS), be familiar with the options in each step of CFD Process, and relate simulation results to AFD concepts. Students will simulate **laminar** pipe flow following the “CFD process” by an interactive step-by-step approach. Students will have “hands-on” experiences using ANSYS to compute axial velocity profile, centerline velocity, centerline pressure, and wall shear stress. Students will compare simulation results with AFD data, analyze the differences and possible numerical errors, and present results in CFD Lab 1 report.



Flow chart for “CFD Process” for pipe flow

2. Simulation Design

In EFD Lab 2, you conducted experimental study for **turbulent** pipe flow. The data you have measured will be used for CFD Lab 1. In CFD PreLab 1, simulation will be conducted only for **laminar** circular pipe flows, i.e. the Reynolds number is less than 2300. Reynolds number based on pipe diameter and mean inlet velocity is **654.75** in the current simulation. CFD predictions of friction factor and fully developed axial velocity profile will be compared with AFD data.

Table 1 – Geometry dimensions

Parameter	Unit	Value
Radius of Pipe	m	0.02619
Diameter of Pipe	m	0.05238
Length of the Pipe	m	7.62

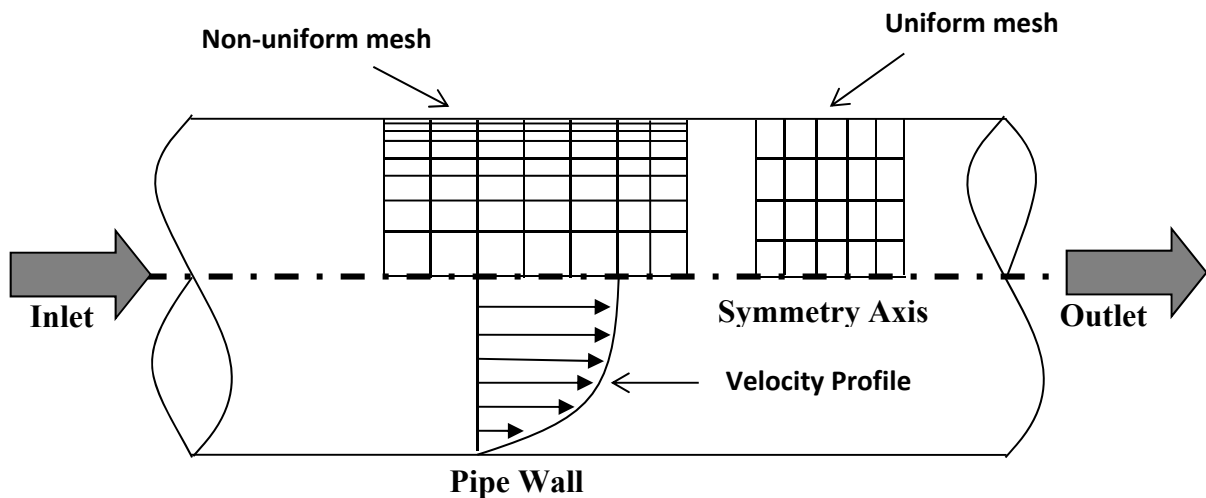


Figure 1 - Geometry

Since the flow is axisymmetric we only need to solve the flow in a single plane from the centerline to the pipe wall. **Boundary conditions** need to be specified include **inlet**, **outlet**, **wall**, and **axis**, as will be described in details later. Uniform flow is specified at inlet, the flow will reach the fully developed regions after a certain distance downstream. No-slip boundary condition will be used on the wall and constant pressure for the outlet. Symmetric boundary condition will be applied on the pipe axis. Since the flow is laminar, turbulence models are not necessary.

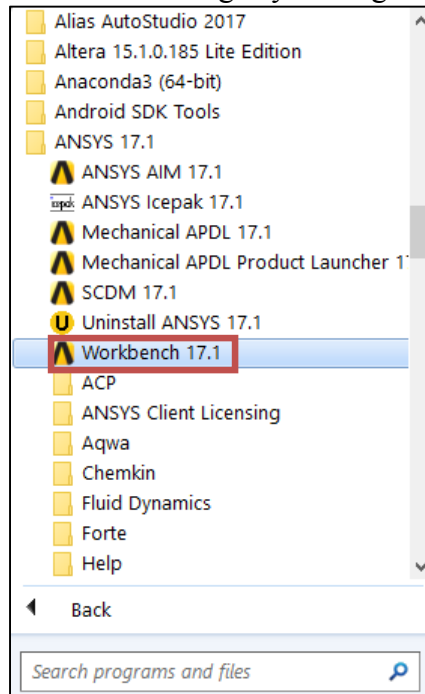
Navigation Tips

- To zoom in and out use the magnifying glass with a plus sign in it and drag, from top left to bottom right over the area you wish to zoom.
- To look at a view plane, simply click on the arrow in the coordinate system identifier in the bottom right of the screen. i.e. if you wish to look at the XYplane, click on the Z Arrow.

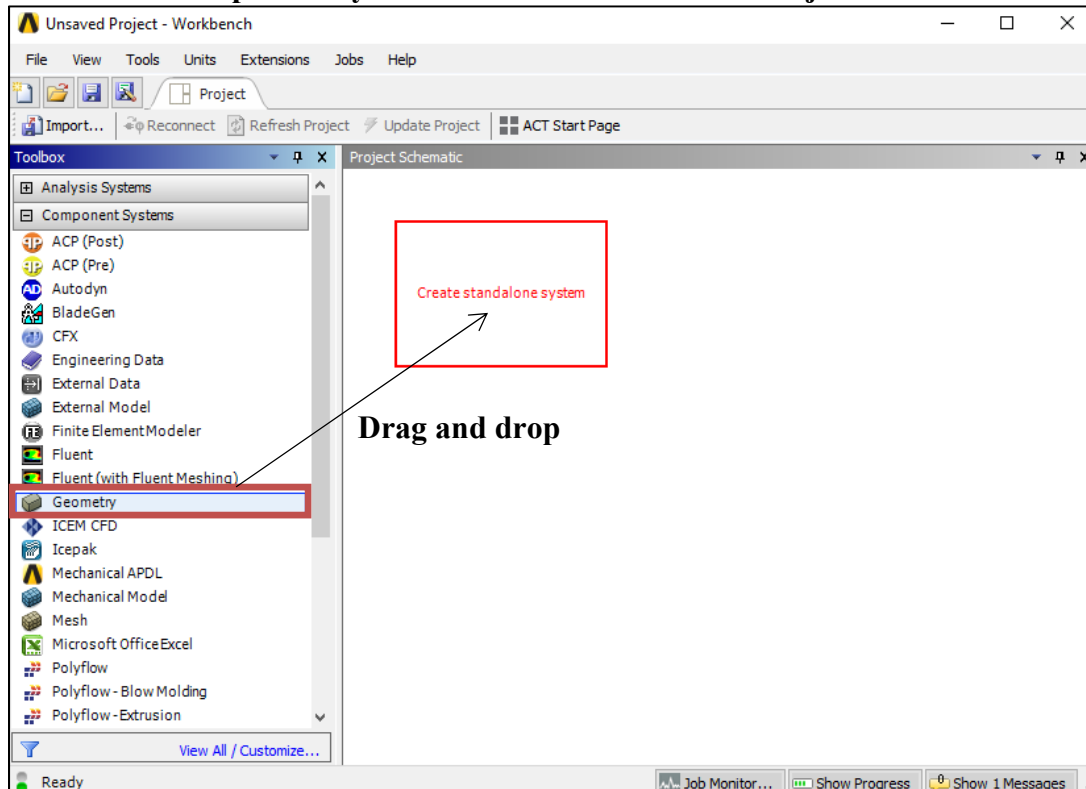
3. Open ANSYS Workbench

3.1. Start > All Programs > ANSYS 17.1 > Workbench 17.1

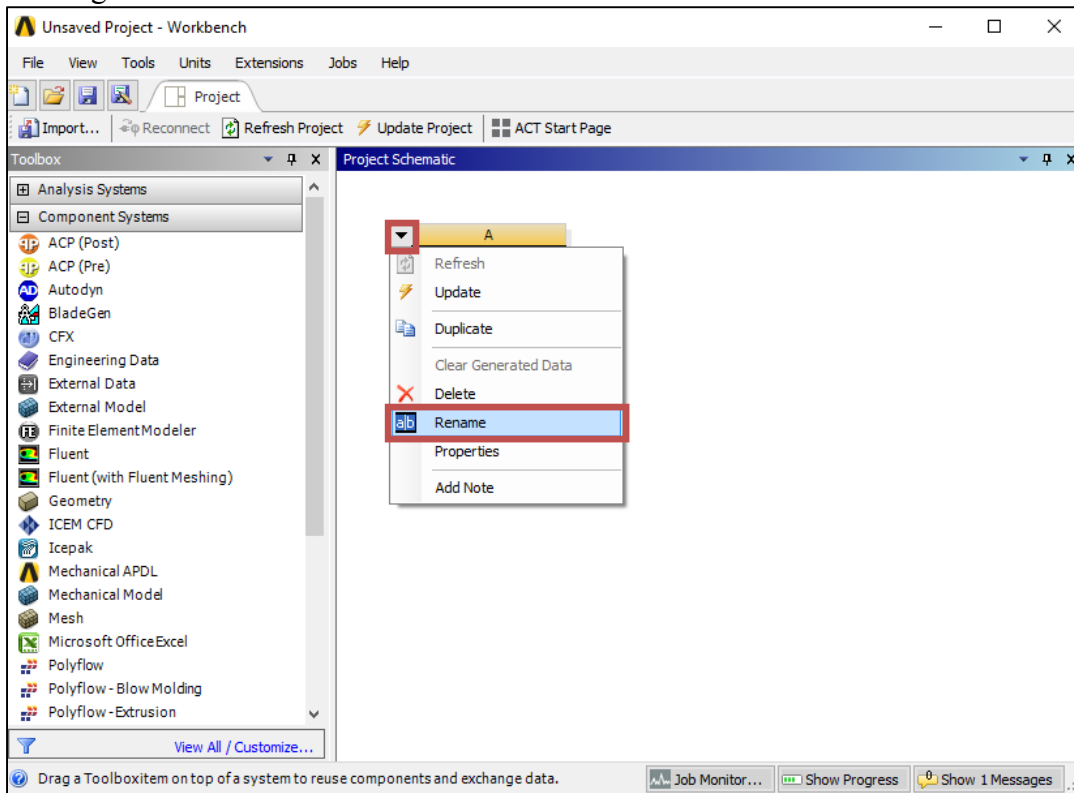
(Note: You may ignore the firewall warnings by closing the pop-up window)



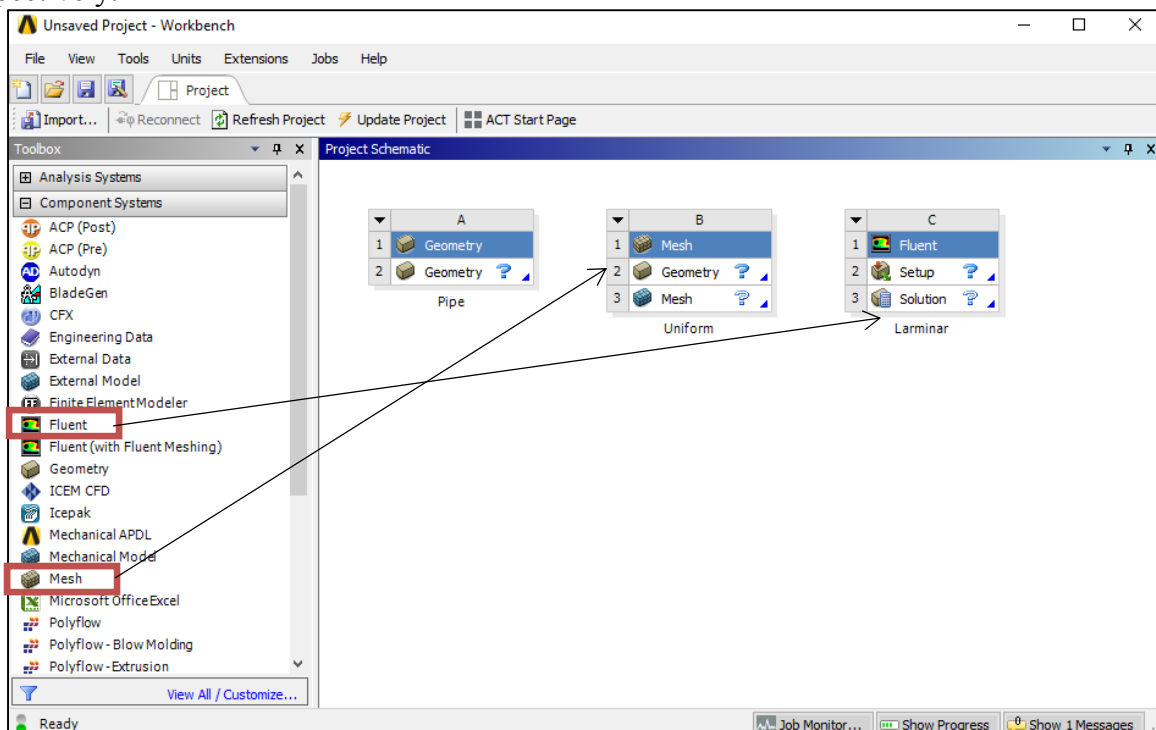
3.2. From the ANSYS Workbench home screen (**Project Schematic**), drag and drop the **Geometry** component for the **Component Systems** in the **Toolbox** into the **Project Schematic**.



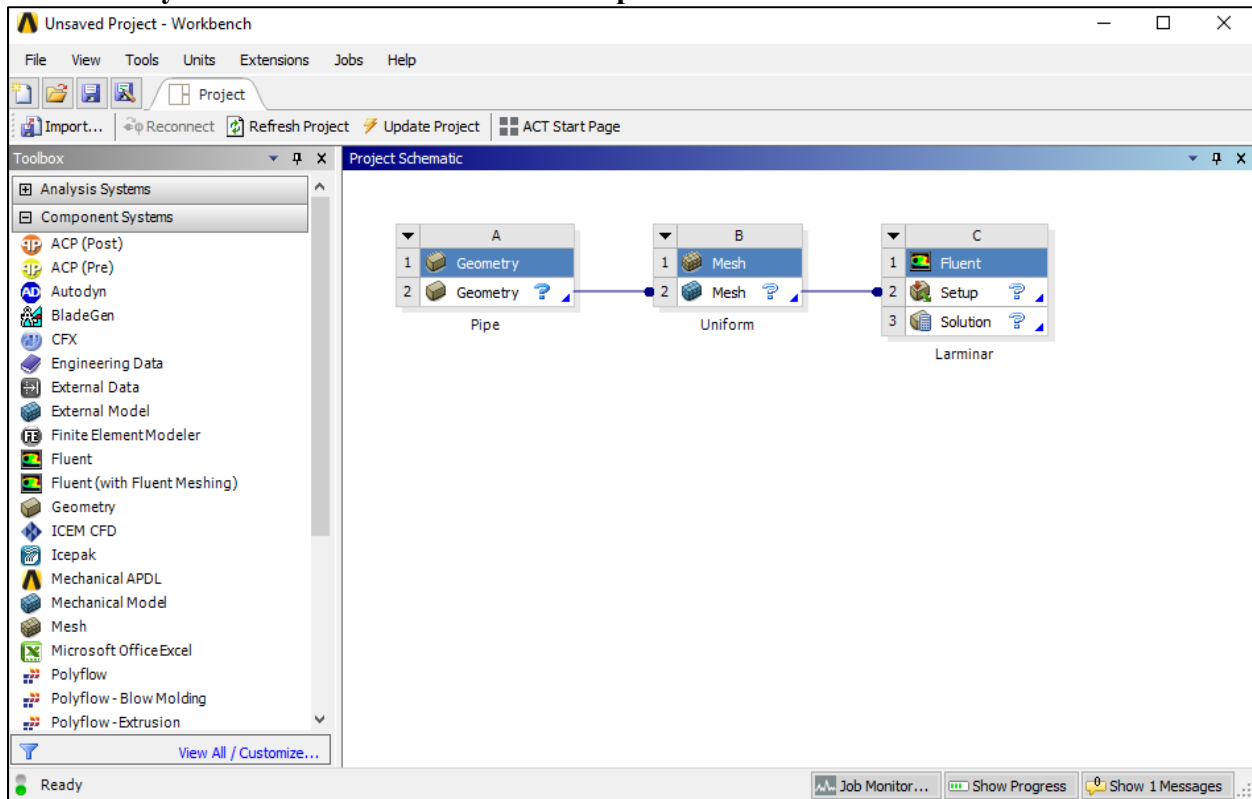
3.3. Rename the geometry “**Pipe**” by right clicking on the down arrow of the Geometry component and selecting **Rename**.



3.4. Drag and drop a **Mesh** component and a **Fluent** component into the **Project Schematic** as shown below. Rename the components as “Uniform” and “Laminar” for **Mesh** and **Fluent** components respectively.



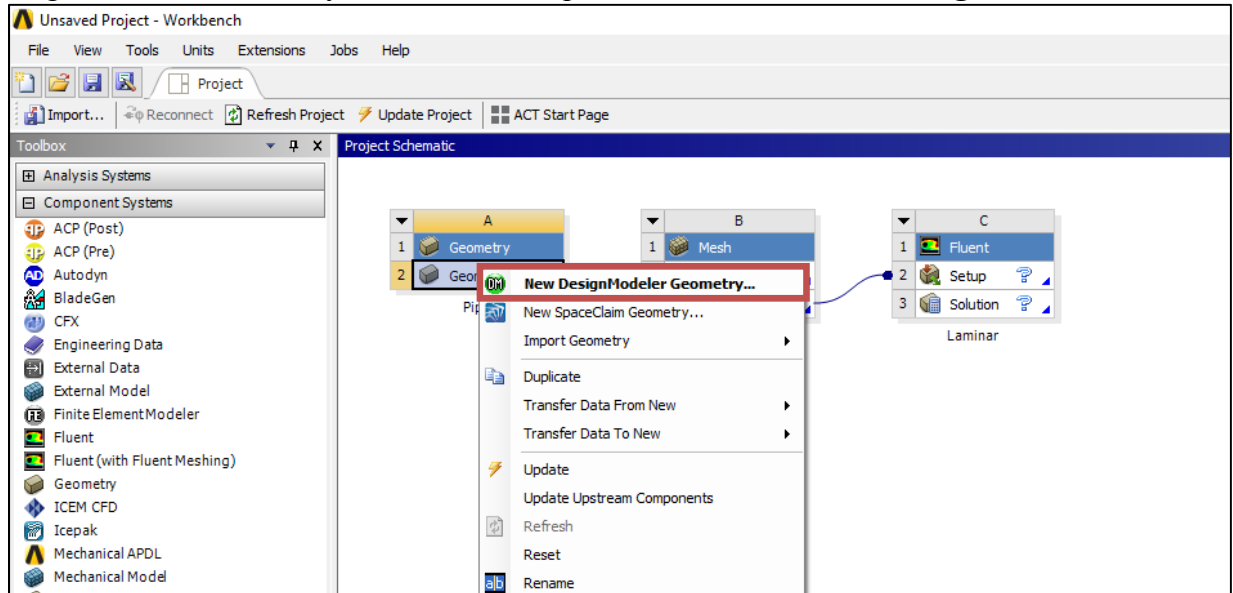
3.5. Make the connections as per below by dragging connections exactly as per below. Drag and drop “Geometry” to “Mesh” and “Mesh” to “Setup”.



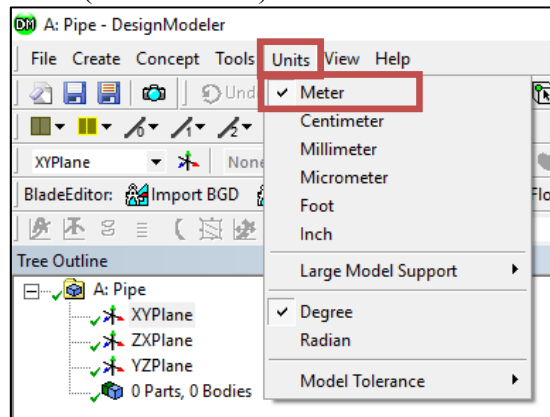
- 3.6. Create a folder on the network drive (home.iowa.uiowa.edu) called “*CFD Pre-Lab and Lab 1*”.
- 3.7. Save the project file by clicking **File > Save As...**
- 3.8. Save the project onto the folder you just created and name it “*CFD Pre-Lab and Lab 1 Pipe Flow*”. (This project file will be used for both Pre-Lab 1 and Lab 1.)

4. Geometry Creation

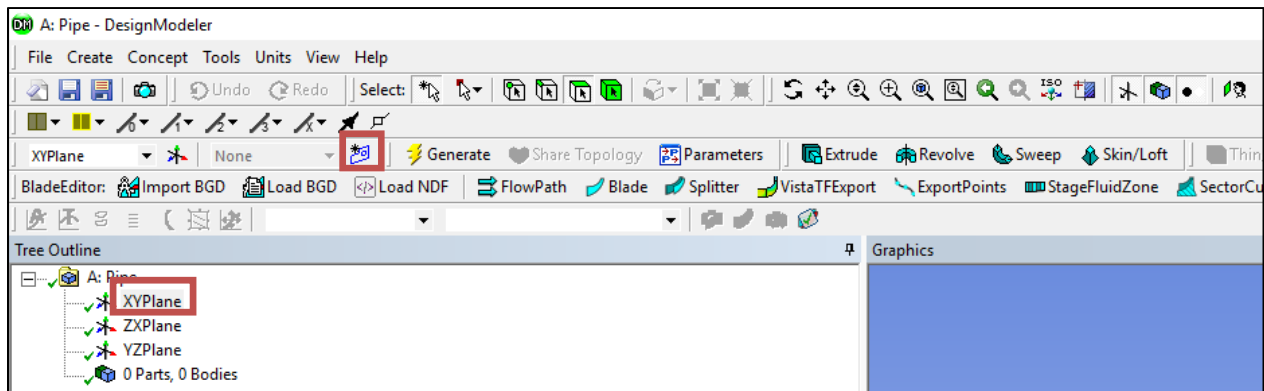
4.1. Right click on **Geometry** and from the drop down menu select **New DesignModeler Geometry...**



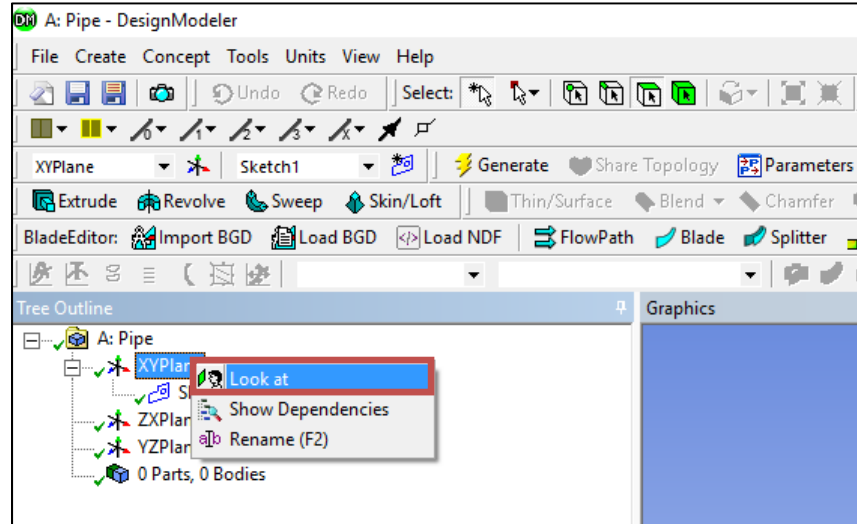
4.2. Make sure that Unit is set to Meter (default value).



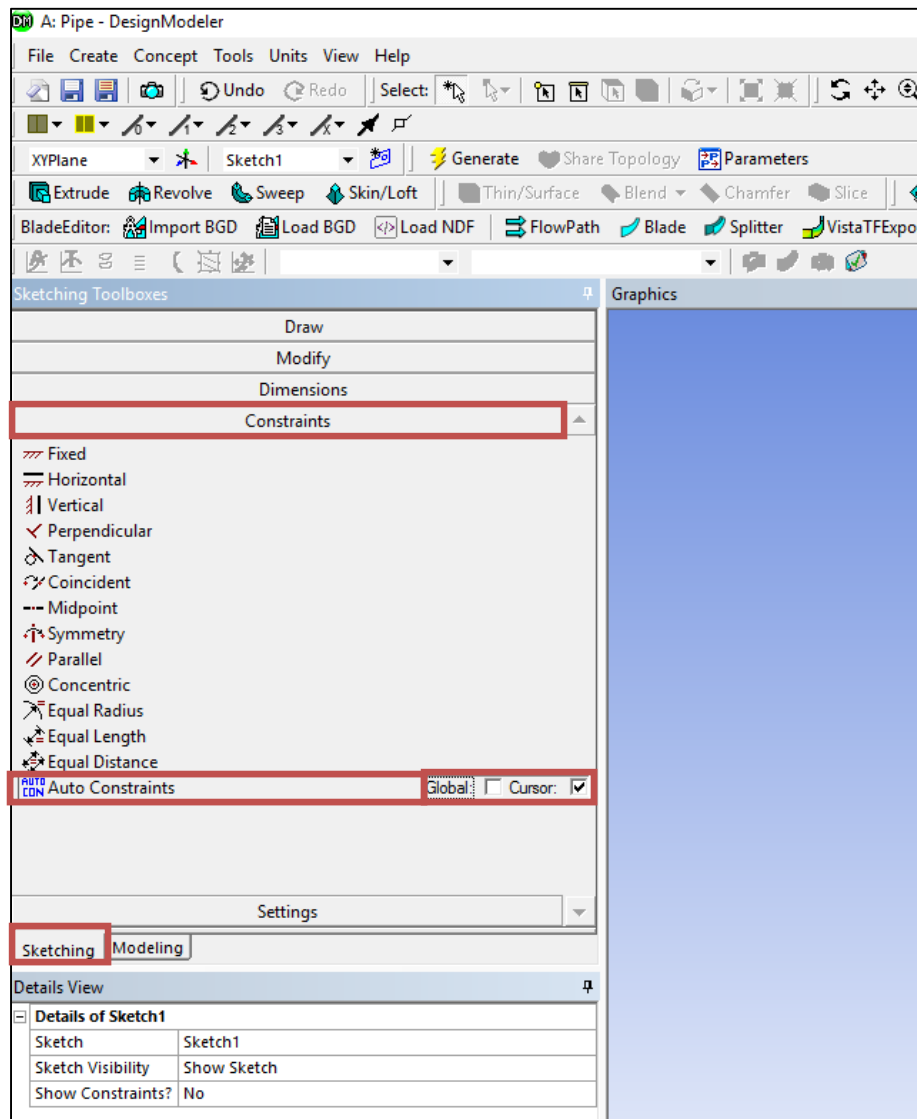
4.3. Select the **XYPlane** under the **Tree Outline** and click **New Sketch** button.



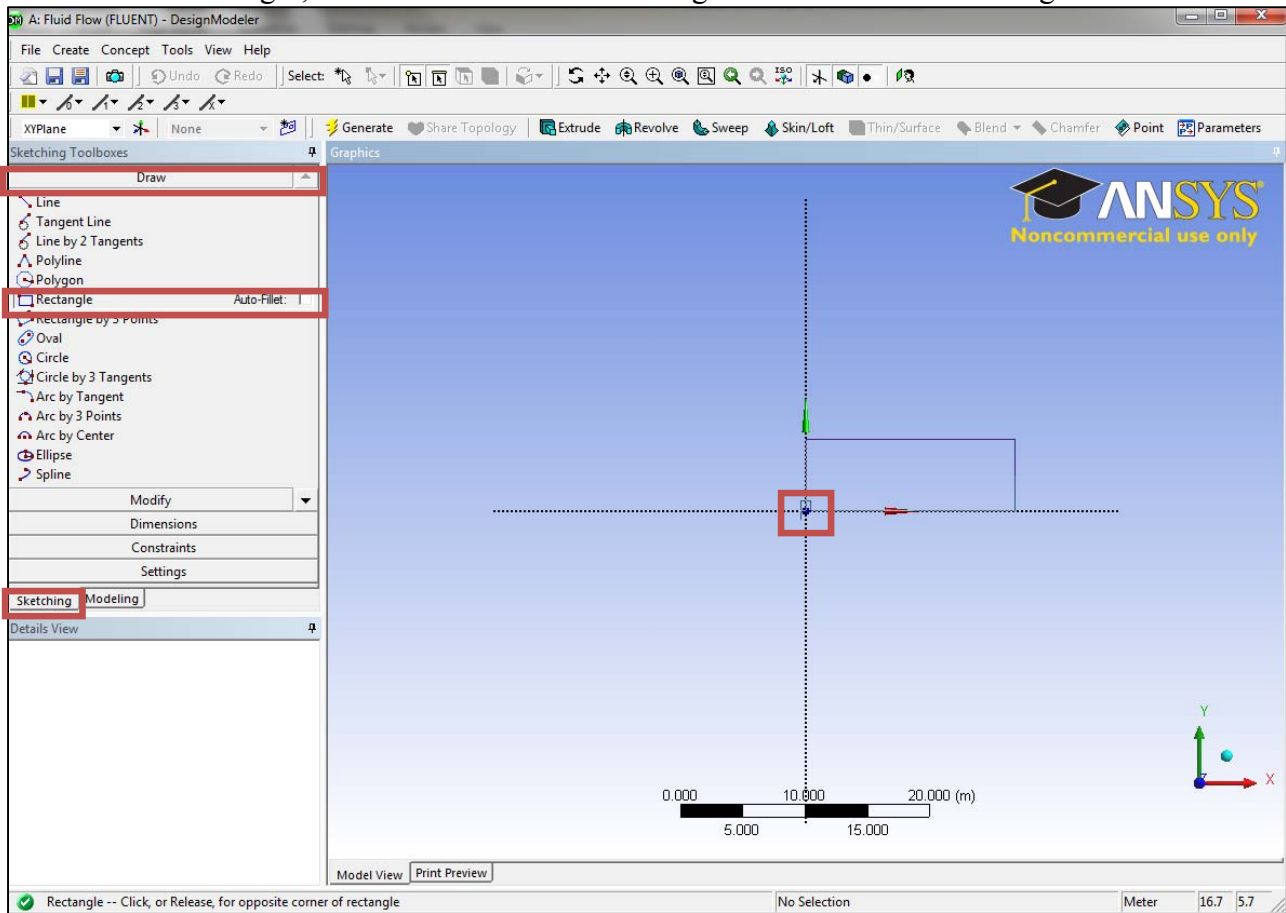
4.4. Right click **XYPlane** and select **Look at**.



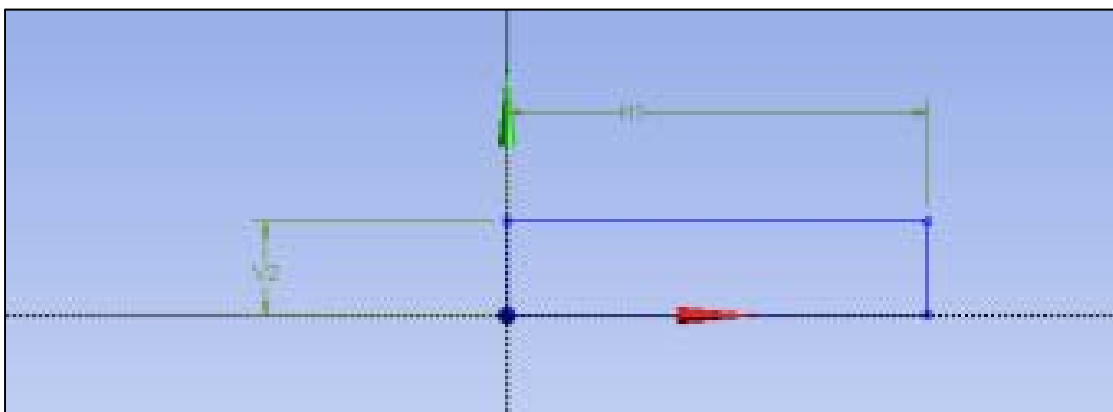
4.5. Select **Sketching > Constraints > Auto Constraints**. Enable the auto constraints option to pick the exact point as below.



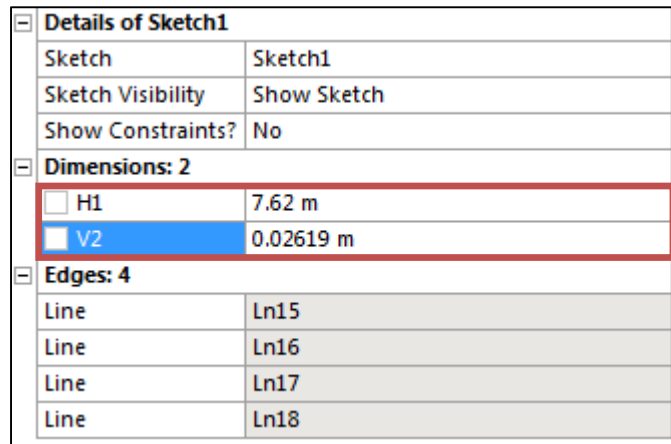
4.6. Select **Sketching** > **Draw** > **Rectangle**. Create a rectangle geometry as per below, make sure to start from the origin, the mouse arrow should change to a “P” when on the origin.



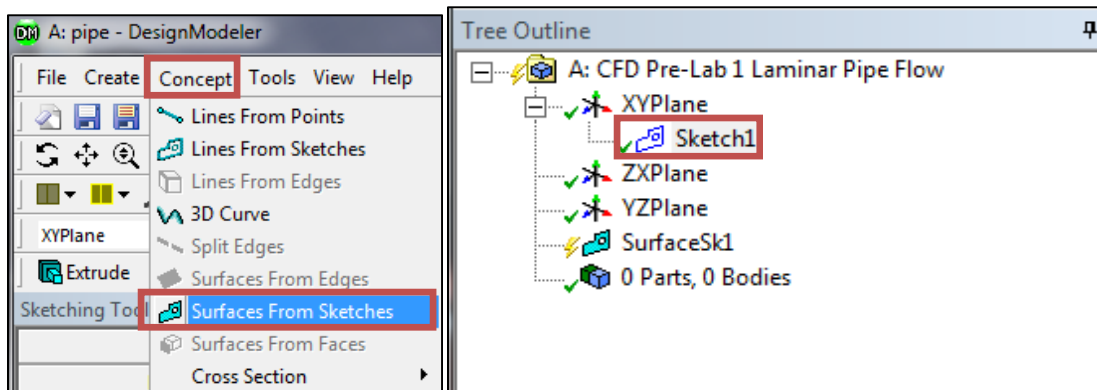
4.7. Select **Dimensions** > **General**. Click on top edge then click above the geometry to place the dimension. Repeat the same thing for one of the vertical edges. You should have a similar figure as per below.



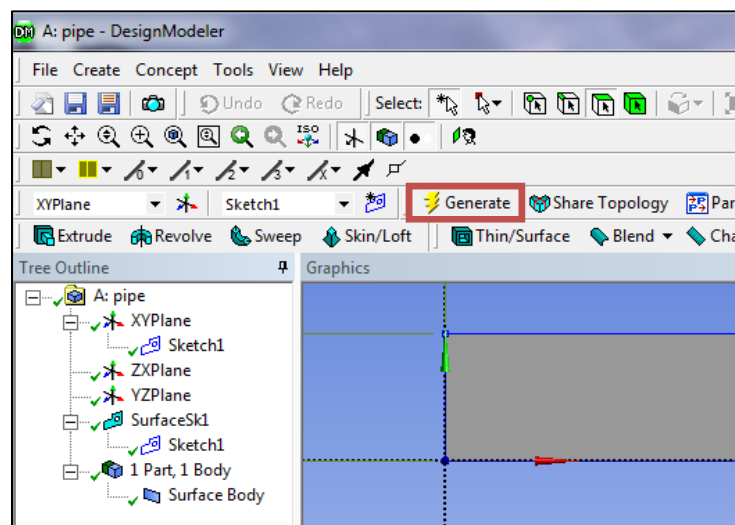
4.8. Click on **H1** under **Details View**, in the bottom left of the screen, and change **H1** to **7.62m**. Click on **V2** and change it to **0.02619m**. (Don't include unit "m" when put in the values)



4.9. **Concept > Surfaces From Sketches**, select the sketch by left clicking on **Sketch1** in the **Tree Outline** and hit **Apply** in the **Details View**.



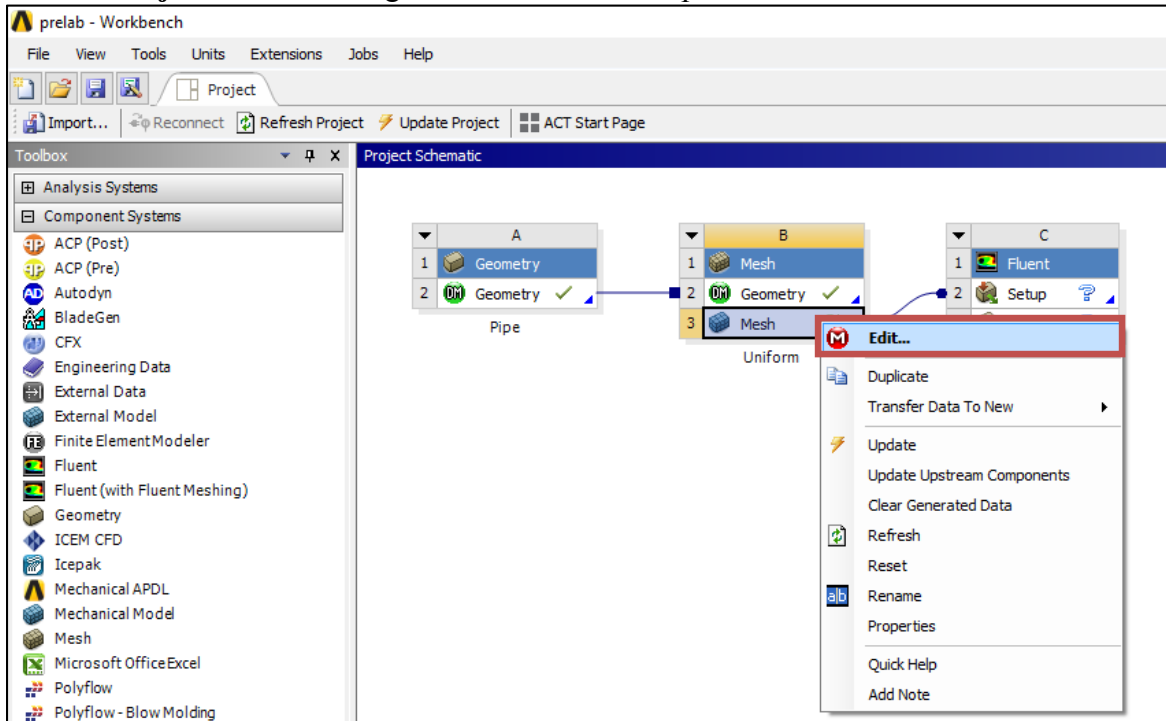
4.10. Click **Generate**. This will create a surface.



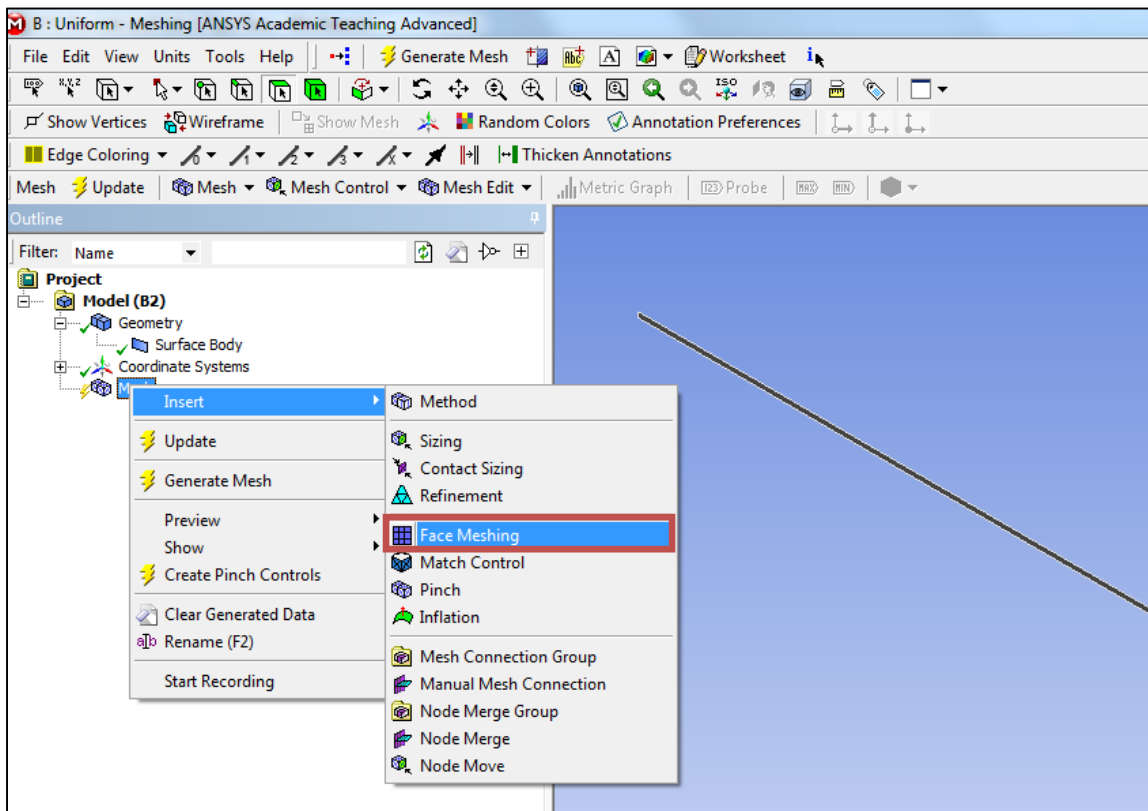
4.11. **File > Save Project**. Save project and close the **Design Modeler** window.

5. Mesh Generation

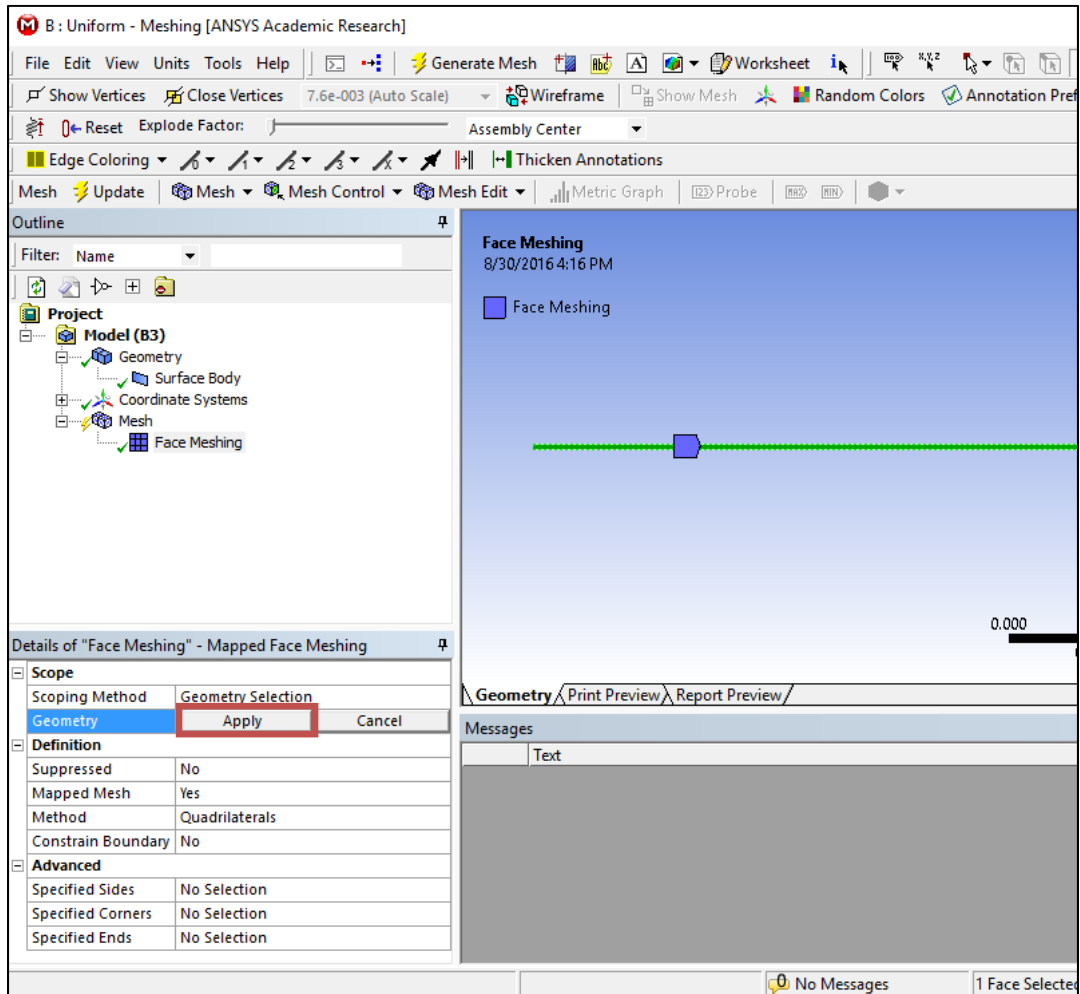
5.1. From the **Project Schematic** right click on **Mesh** component and select **Edit...**



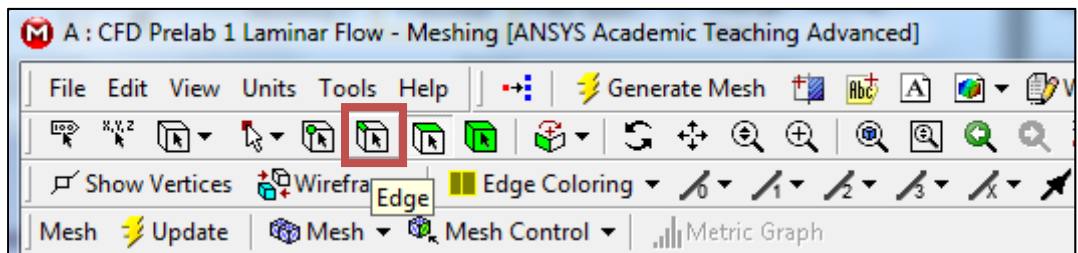
5.2. Right click on **Mesh** then select **Insert > Face Meshing**.



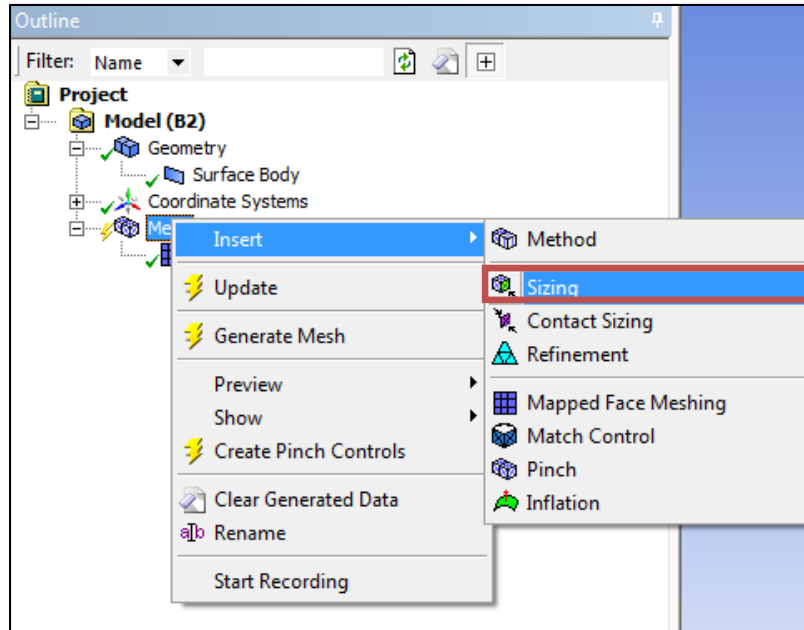
5.3. Select your geometry by clicking the yellow box which says **No Selection**, then click on the geometry surface, and click **Apply**. (Note: You can change orientation of your view to xy plane by clicking the z-axis figure on the lower right corner. Press “F7” on your keyboard to restore to the “whole view”. Zoom in by holding the right mouse button and selecting a region.)



5.4. Click on the edge button. This will allow you to select edges of your geometry.



5.5. Right click on **Mesh** then select **Insert > Sizing**.



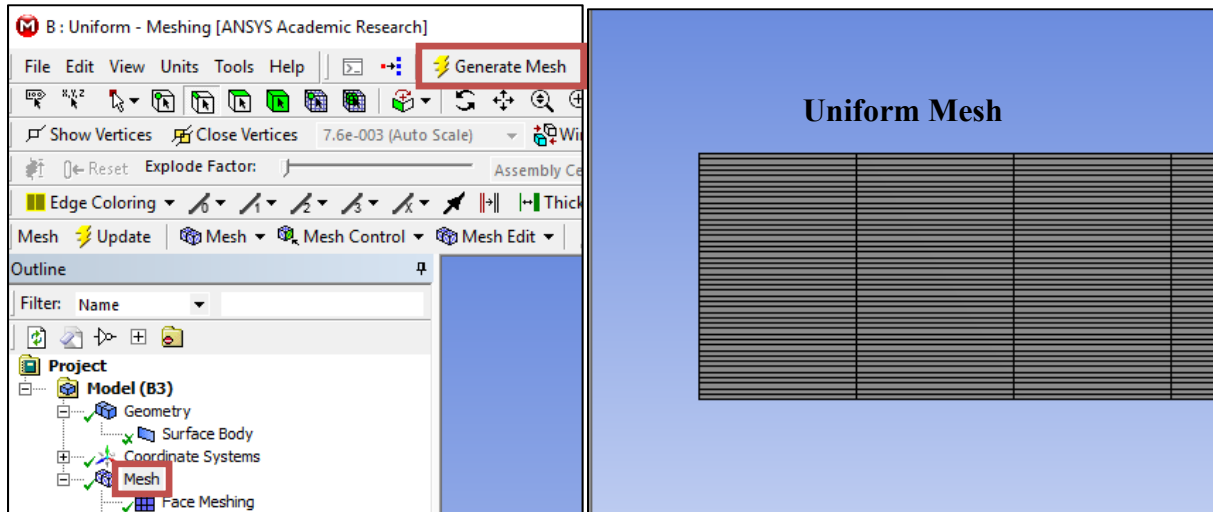
5.6. Hold **Ctrl** button and select the top and bottom edge of the rectangle then click **Apply** on **Geometry**. Specify details of sizing as per below in the **Details of "Edge Sizing" – Sizing** window.

Details of "Edge Sizing" - Sizing	
Scope	
Scoping Method	Geometry Selection
Geometry	2 Edges
Definition	
Suppressed	No
Type	Number of Divisions
<input type="checkbox"/> Number of Divisions	453
Behavior	Hard
Bias Type	No Bias

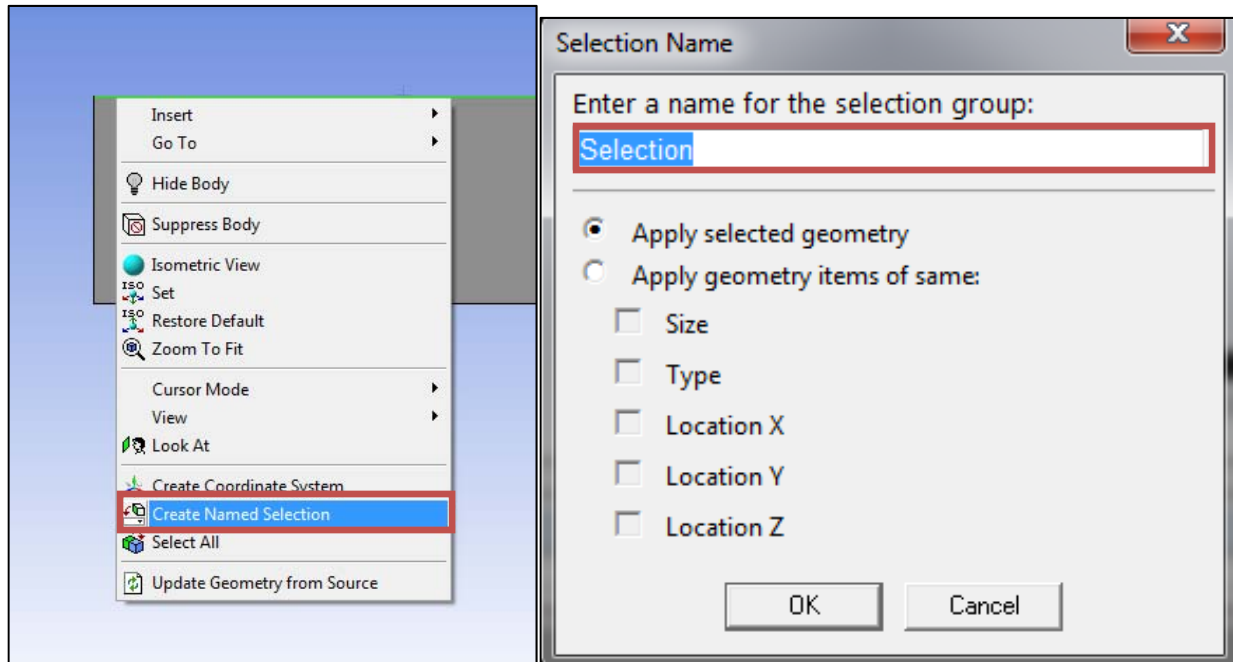
5.7. Repeat step 5.5. to insert **Sizing**. Select the left and right edge of the rectangle and click **Apply** then change sizing parameters as per below (Please see 5.3. for view restoring and zooming in).

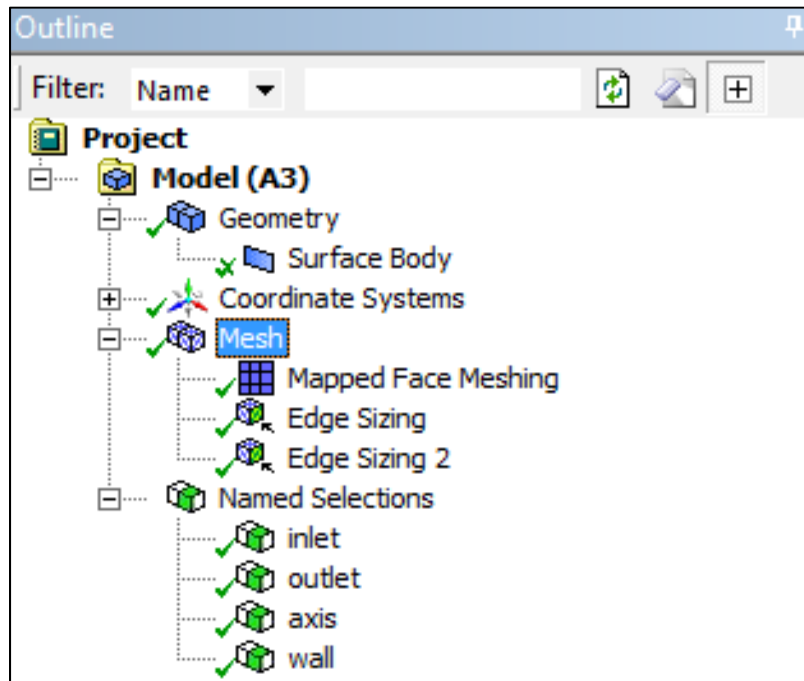
Details of "Edge Sizing 2" - Sizing	
Scope	
Scoping Method	Geometry Selection
Geometry	2 Edges
Definition	
Suppressed	No
Type	Number of Divisions
<input type="checkbox"/> Number of Divisions	45
Behavior	Hard
Bias Type	No Bias

5.8. Click on **Generate Mesh** button. Click **Mesh** under **Outline**. The mesh should look like the mesh pictured below.

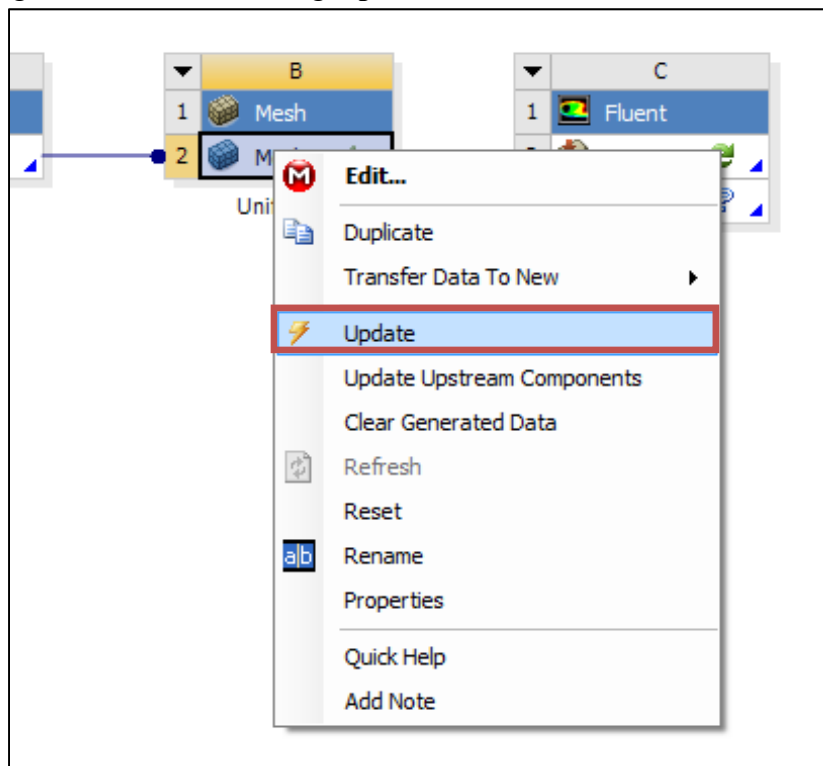


5.9. Change the edge names by selecting the edge, then right click on the edge and select **Create Named Selection**. Name left, right, bottom and top edges as *inlet*, *outlet*, *axis* and *wall* respectively. Your outline should look same as the figure below (next page).



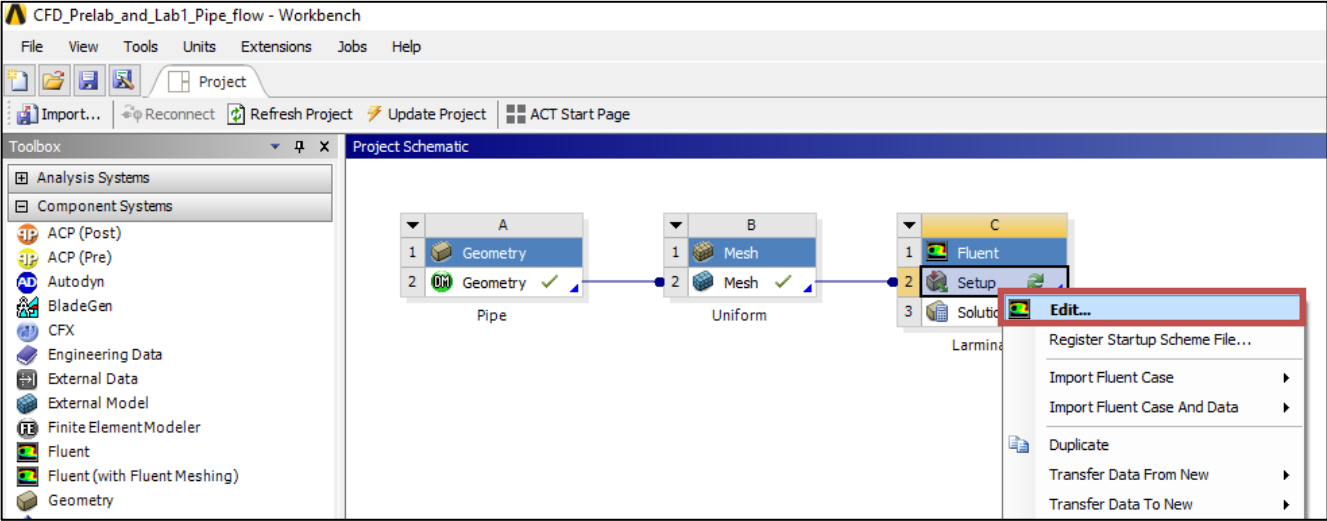


5.10. **File > Save Project.** Save the project and close the window. Update mesh on **Project Schematic** by right clicking on **Mesh** and selecting **Update**.

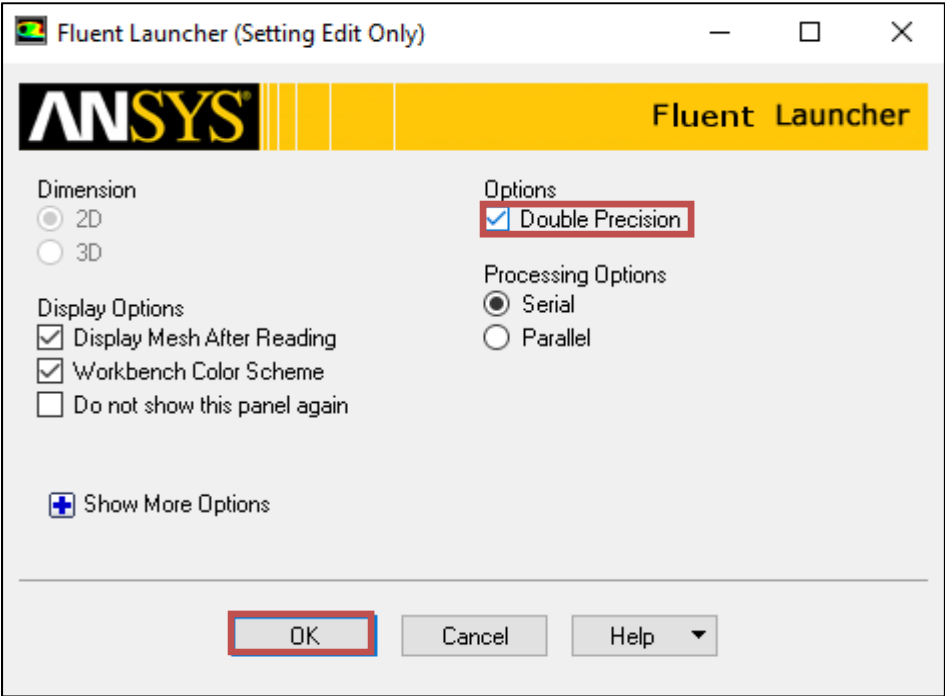


6. Solve (Physics)

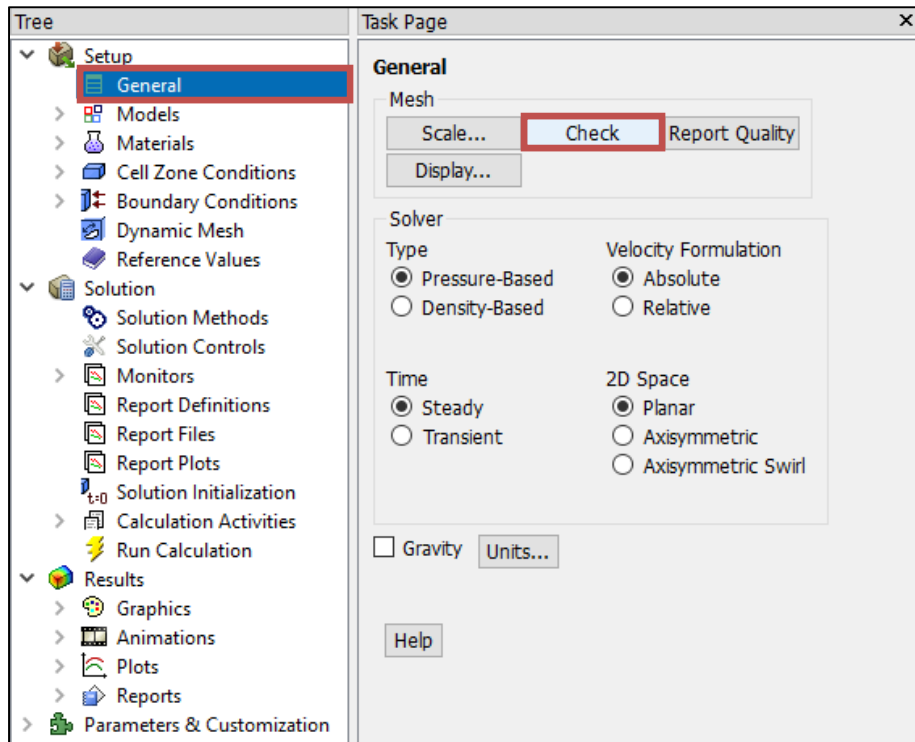
6.1. Right click Setup and select Edit...



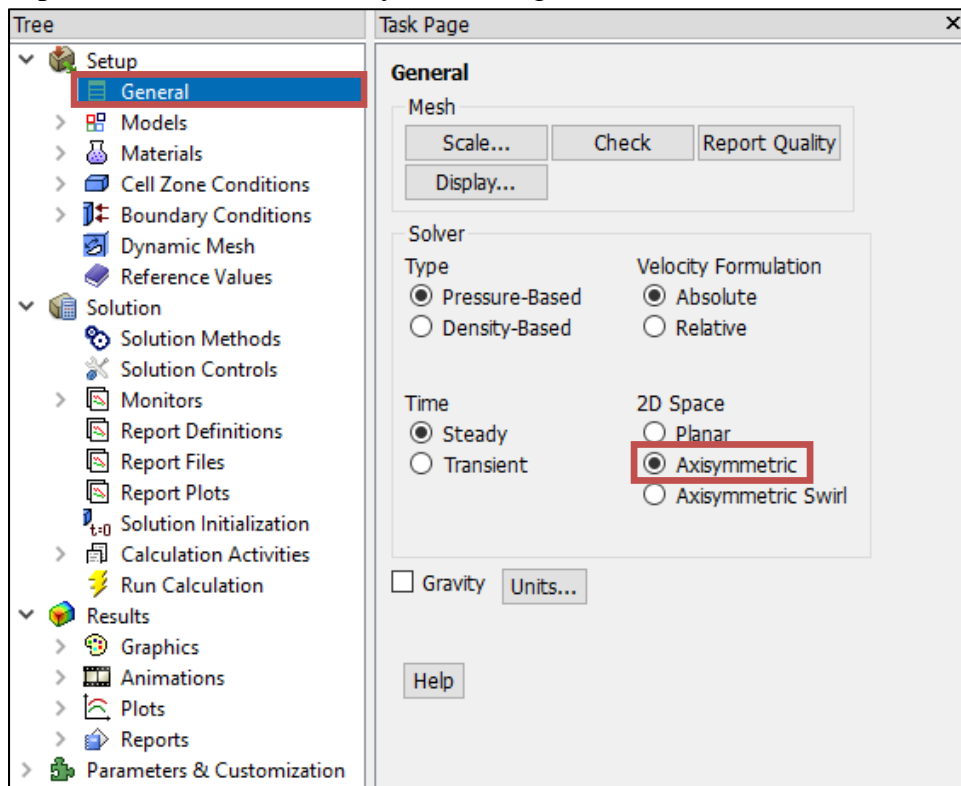
6.2. Check Double Precision and click OK.



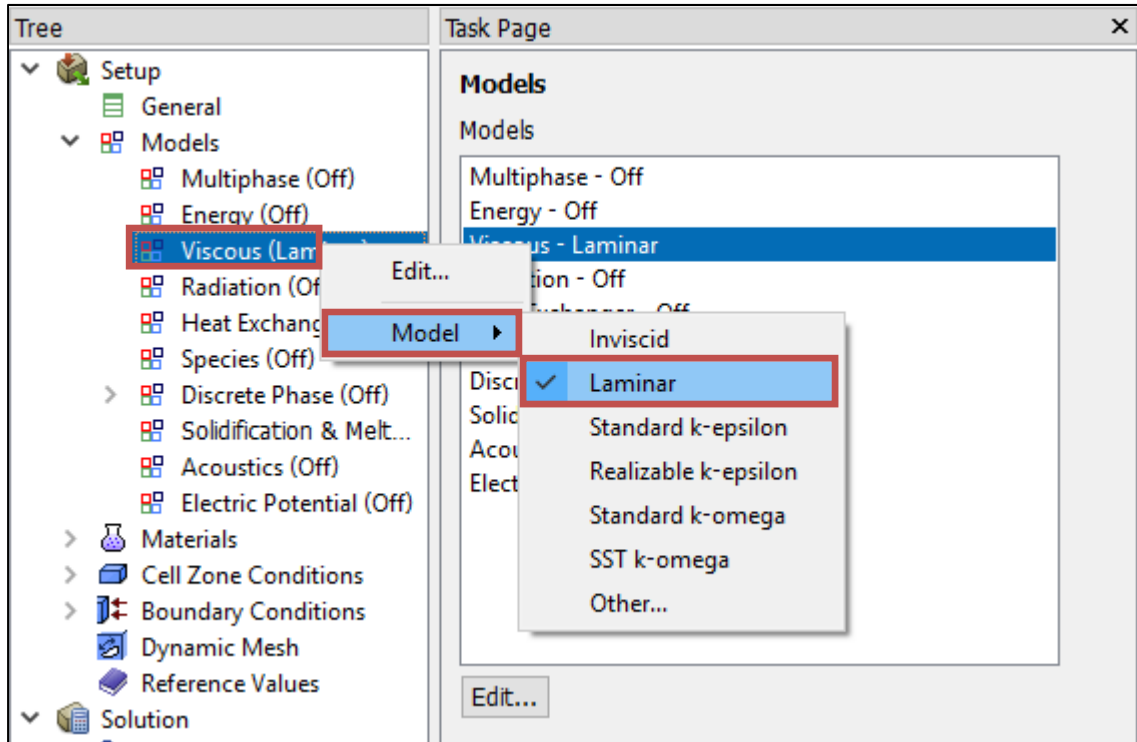
6.3. **Tree > Setup > General > Check.** (Note: If you get an error message you may have made a mistake while creating your mesh. Review steps in mesh generation and make changes.)



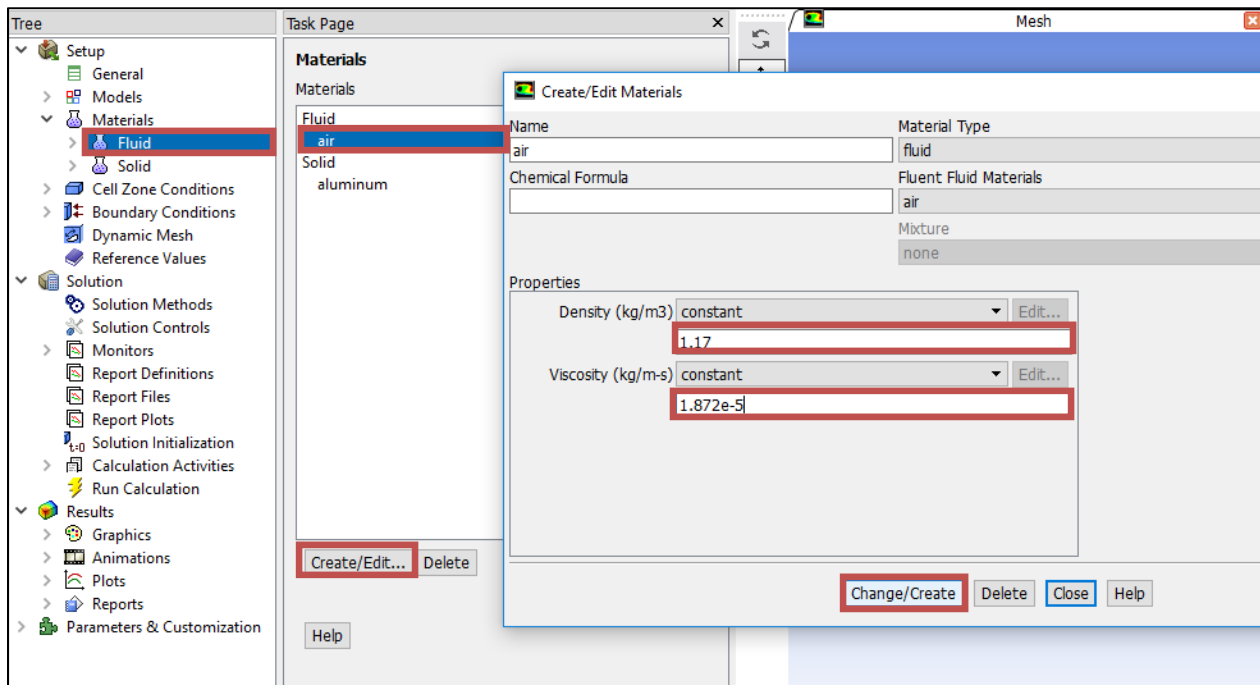
6.4. **Tree > Setup > General.** Choose **Axisymmetric** option shown below.



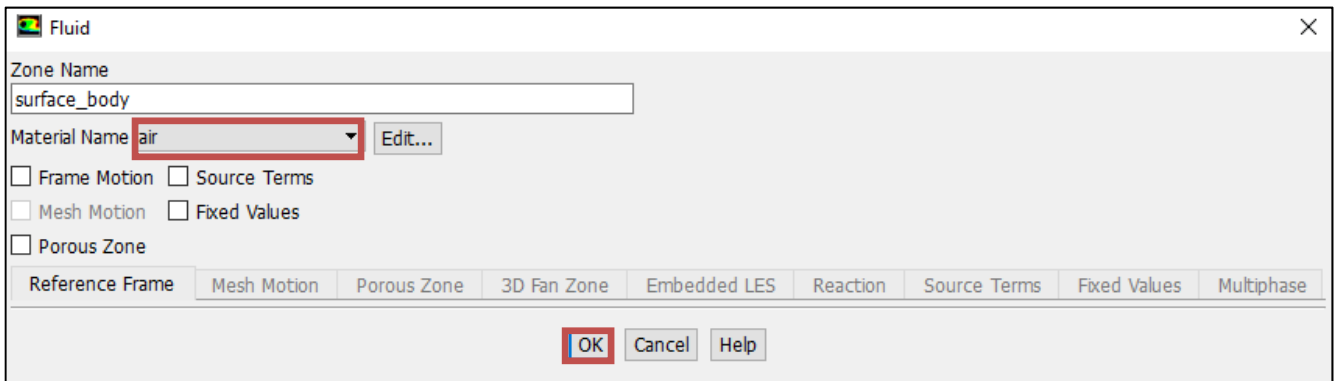
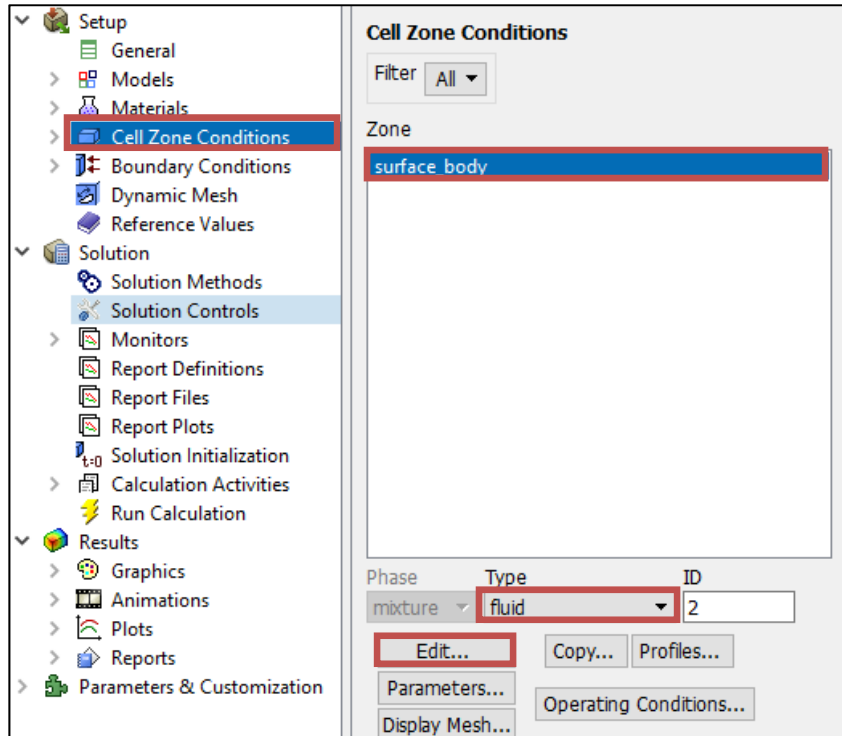
6.5. **Tree > Setup > Models > Viscous (RMB click) > Model.** Select **Laminar**.



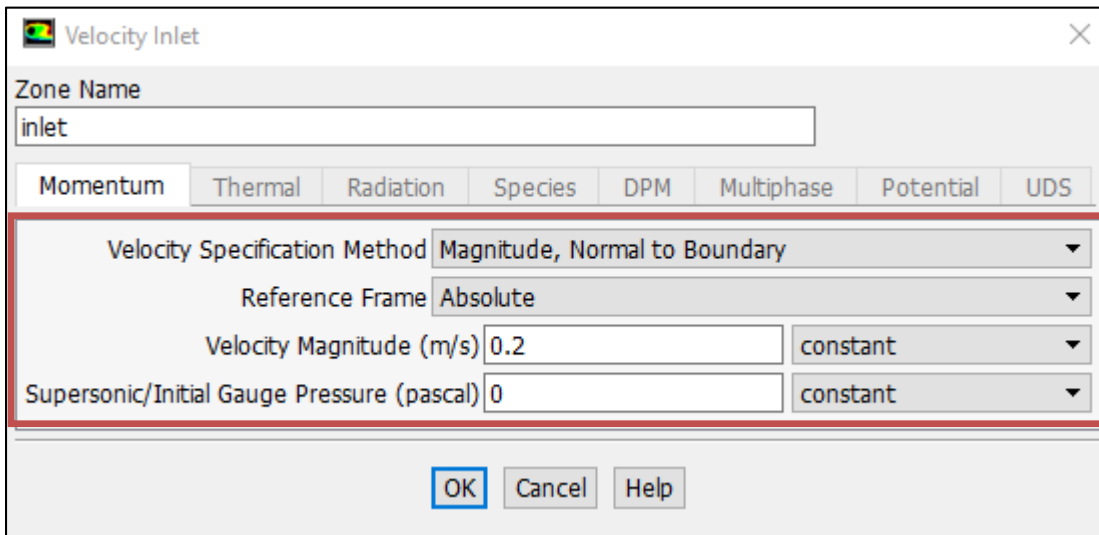
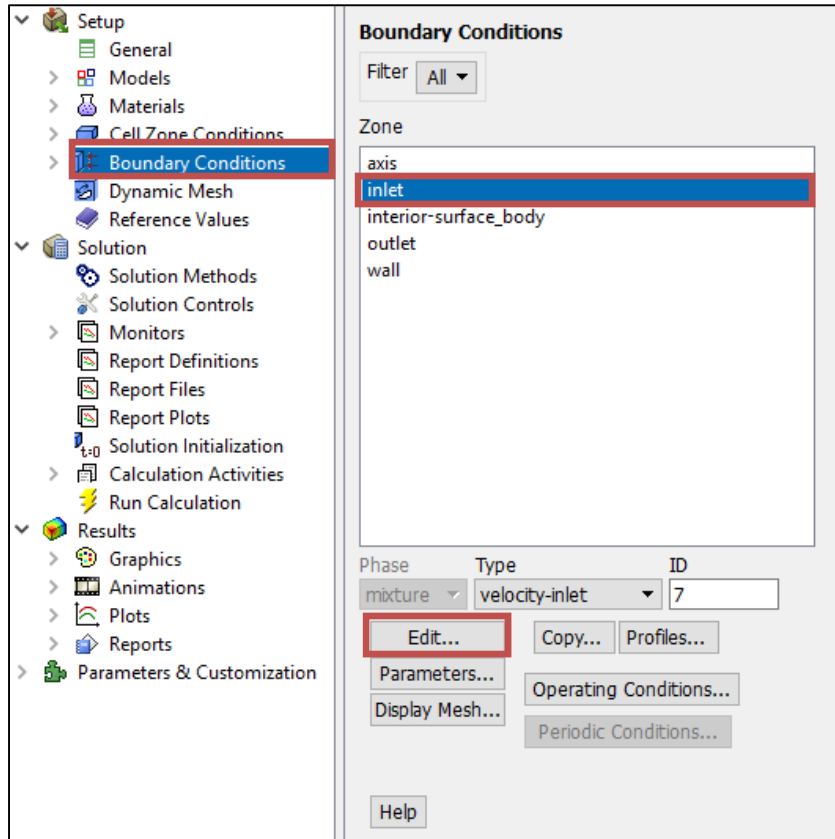
6.6. **Tree > Setup > Materials.** Right click on **air** and left click on **Create/edit...**. Change the **Density** and **Viscosity** as per below and click **Change/Create**. **Close** the dialog box when finished.



6.7. Tree > Setup > Cell Zone Conditions > Zone > surface_body. Change type to **fluid**. Select **Material Name** as **air** and click **OK**. This should be defaulted to fluid.

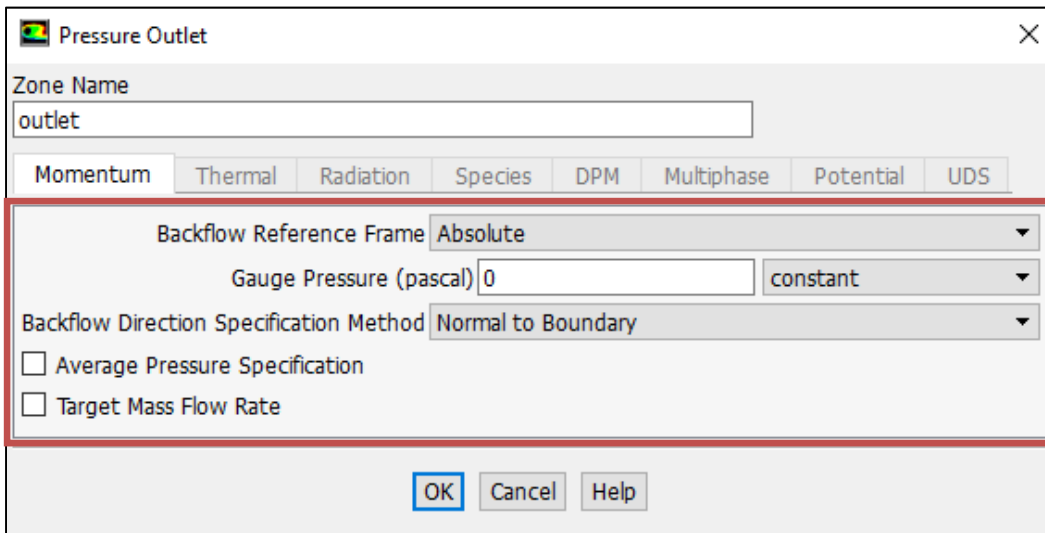
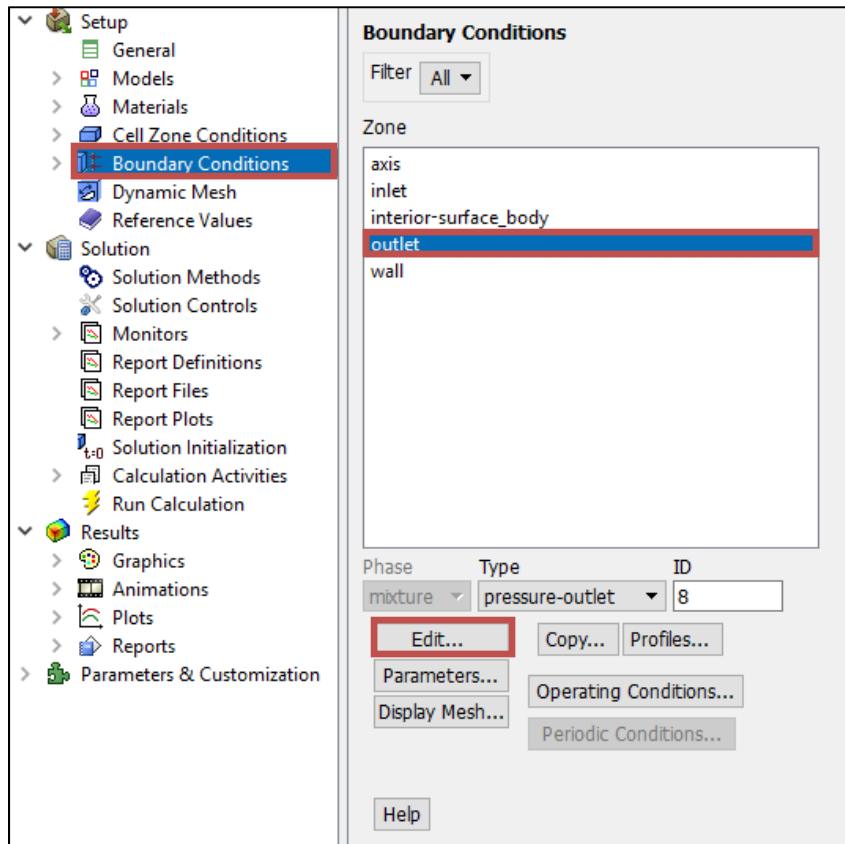


6.8. Tree > Setup > **Boundary Conditions** > inlet > **Edit...** Change parameters as per below and click **OK**. Table below shows the details of the boundary conditions.



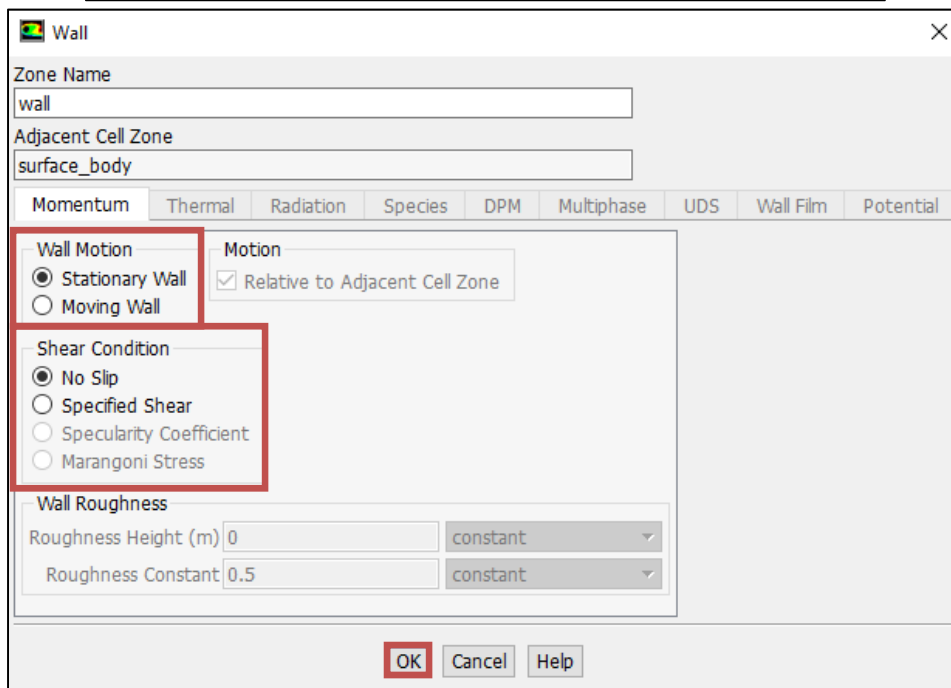
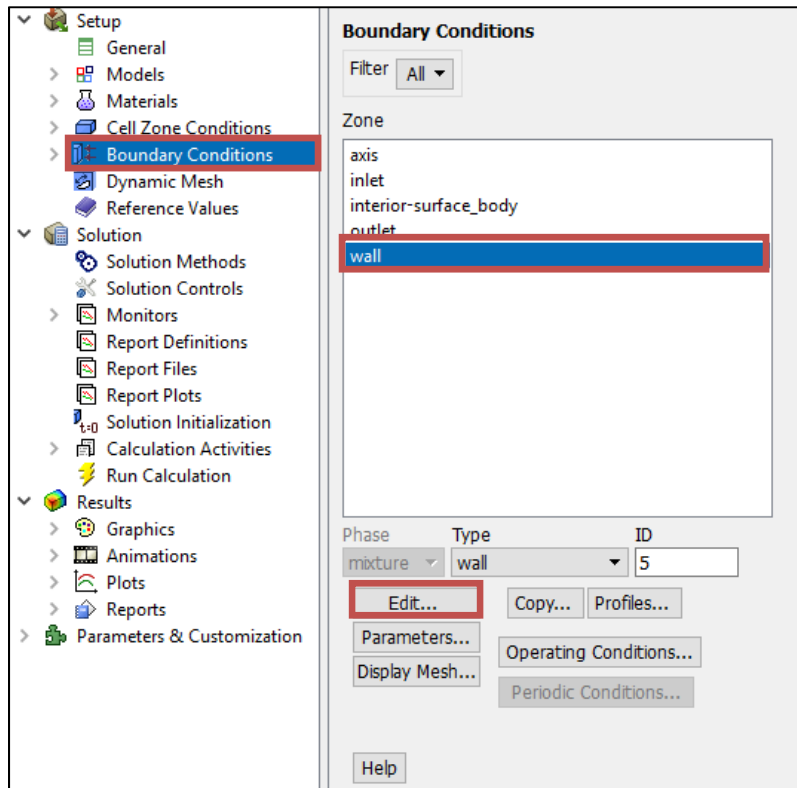
Inlet Boundary Condition			
Variable	u (m/s)	v (m/s)	P (Pa)
Magnitude	0.2	0	-
Zero Gradient	N	N	Y

6.9. **Tree > Setup > Boundary Conditions > outlet > Edit...** Change parameters as per below and click **OK**. Table below shows the details of the boundary conditions.



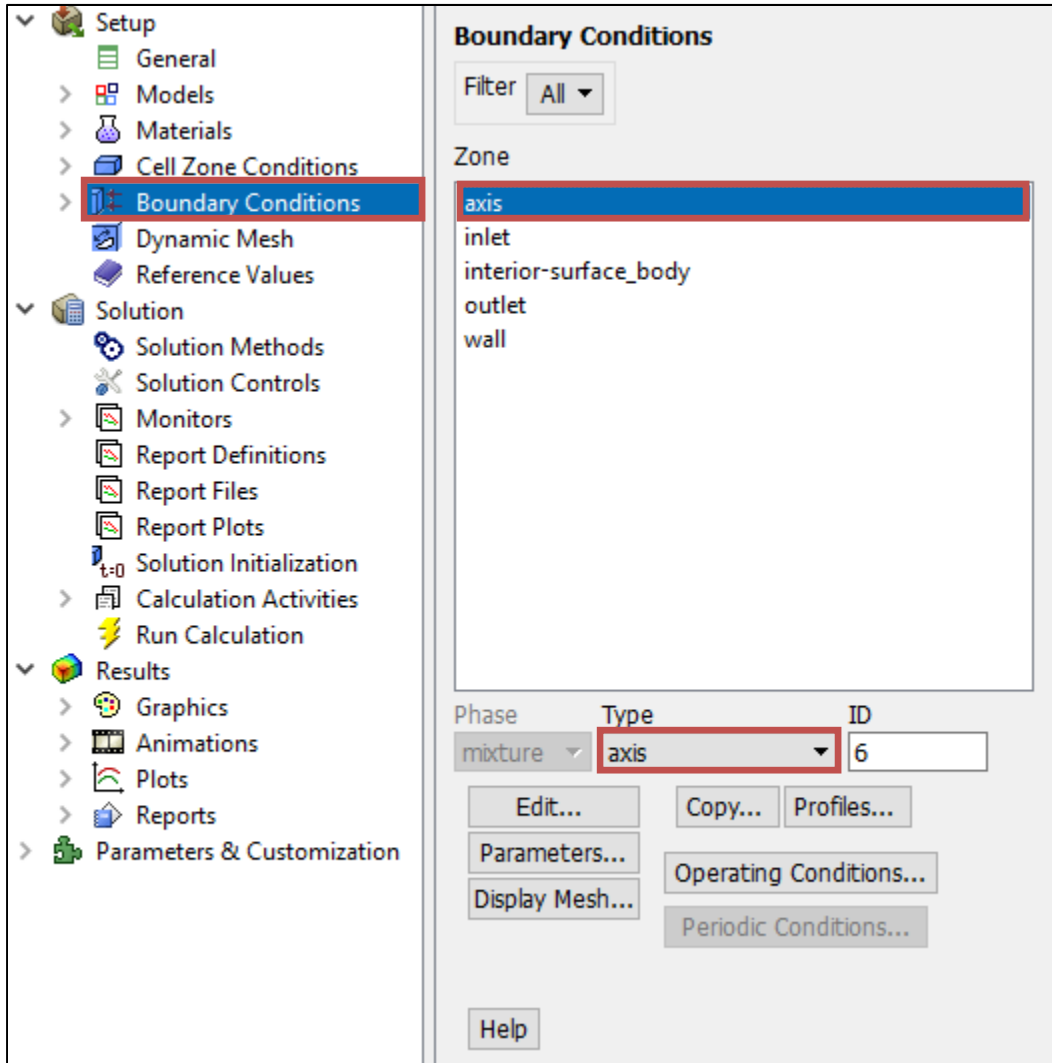
Outlet Boundary Condition			
Variable	u (m/s)	v (m/s)	P (Pa)
Magnitude	-	-	0
Zero Gradient	Y	Y	N

6.10. **Tree > Setup > Boundary Conditions > wall > Edit...** Change parameters as per below and click **OK**. Table below shows the details of the boundary conditions.



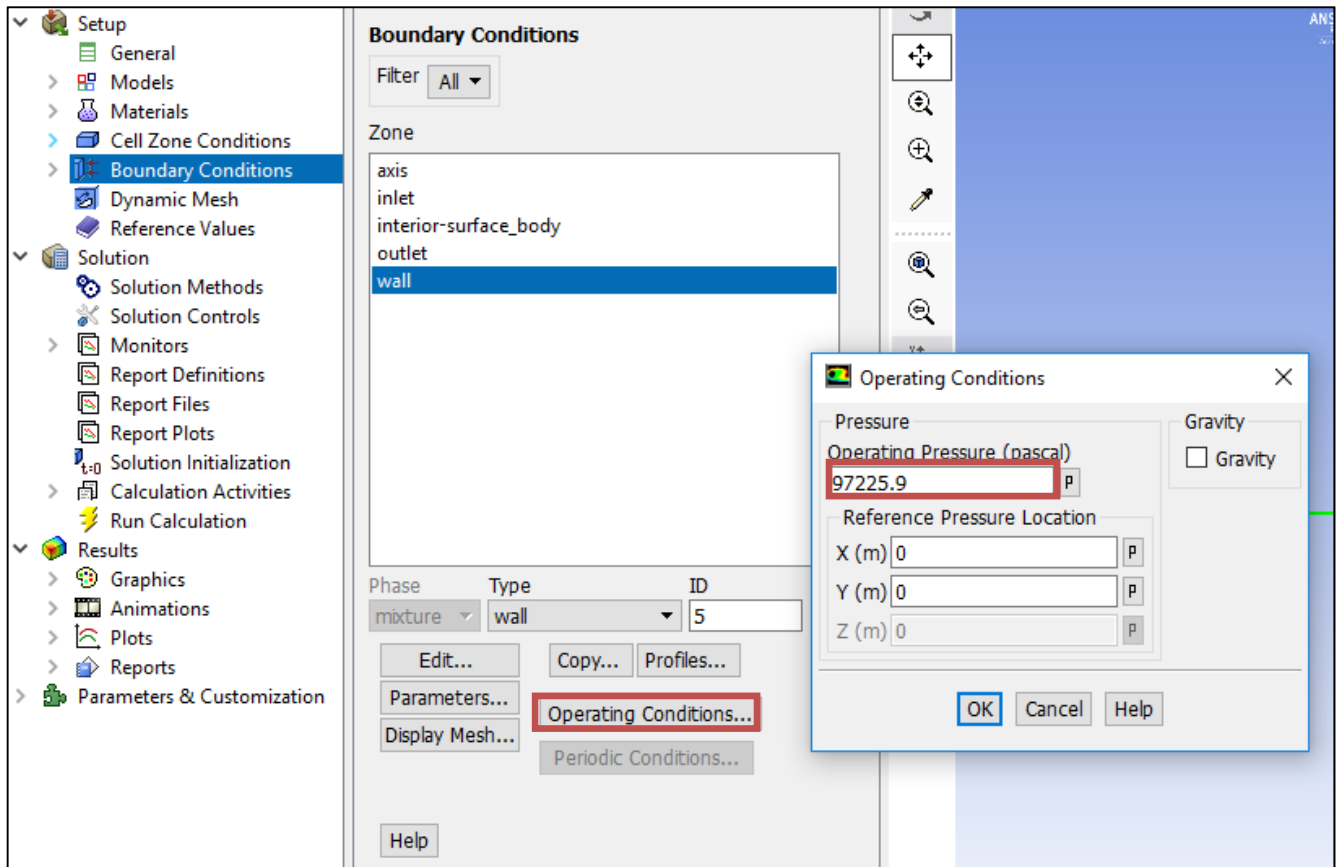
Wall Boundary Condition			
Variable	u (m/s)	v (m/s)	P (Pa)
Magnitude	0	0	-
Zero Gradient	N	N	Y

6.11. **Tree > Setup > Boundary Conditions > axis.** Make sure that axis is selected as per below.



Axis Boundary Condition			
Variable	u (m/s)	v (m/s)	P (Pa)
Magnitude	-	0	-
Zero Gradient	Y	N	Y

6.12. **Tree > Setup > Boundary Conditions > Operating Condition.** Change parameters as per below and click OK.



6.13. **Tree > Setup > Reference Values.** Change parameters as per below.

The screenshot displays the 'Reference Values' configuration panel in a software application. The left sidebar shows a tree view with 'Reference Values' selected. The right panel contains the following settings:

Property	Value
Area (m ²)	0.002154869
Density (kg/m ³)	1.17
Enthalpy (j/kg)	0
Length (m)	0.05238
Pressure (pascal)	0
Temperature (k)	298.16
Velocity (m/s)	0.2
Viscosity (kg/m-s)	1.872e-05
Ratio of Specific Heats	1.4

Reference Zone: surface_body

Help

6.14. **Tree > Solution > Solution Methods.** Change parameters as per below.

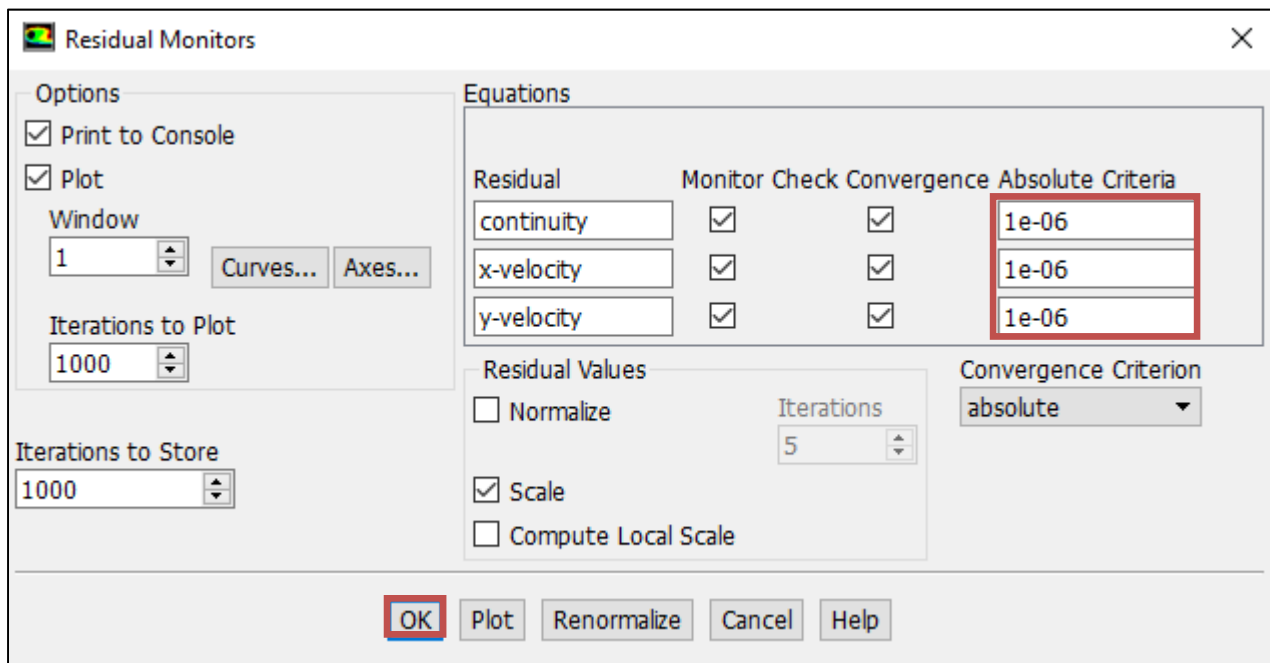
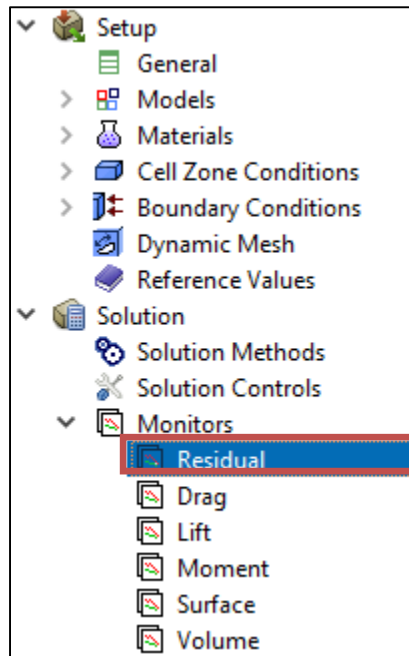
The screenshot displays the software's configuration interface. On the left, a tree view shows the following structure:

- Setup
 - General
 - Models
 - Materials
 - Cell Zone Conditions
 - Boundary Conditions
 - Dynamic Mesh
 - Reference Values
- Solution
 - Solution Methods** (highlighted)
 - Solution Controls
 - Monitors
 - Report Definitions
 - Report Files
 - Report Plots
 - Solution Initialization
 - Calculation Activities
 - Run Calculation
- Results
 - Graphics
 - Animations
 - Plots
 - Reports
 - Parameters & Customization

The right-hand panel, titled "Solution Methods", contains the following settings:

- Pressure-Velocity Coupling: [Dropdown]
- Scheme: SIMPLE [Dropdown]
- Spatial Discretization
 - Gradient: Green-Gauss Cell Based [Dropdown]
 - Pressure: Second Order [Dropdown]
 - Momentum: Second Order Upwind [Dropdown]
- Transient Formulation: [Dropdown]
- Non-Iterative Time Advancement
- Frozen Flux Formulation
- Pseudo Transient
- Warped-Face Gradient Correction
- High Order Term Relaxation [Options...]
- [Default]
- [Help]

6.15. **Tree > Solution > Monitors > Residual.** Right click **Residual** and select **Edit...** Change convergence criterion to **1e-06** for all three equations as per below and click **OK**.



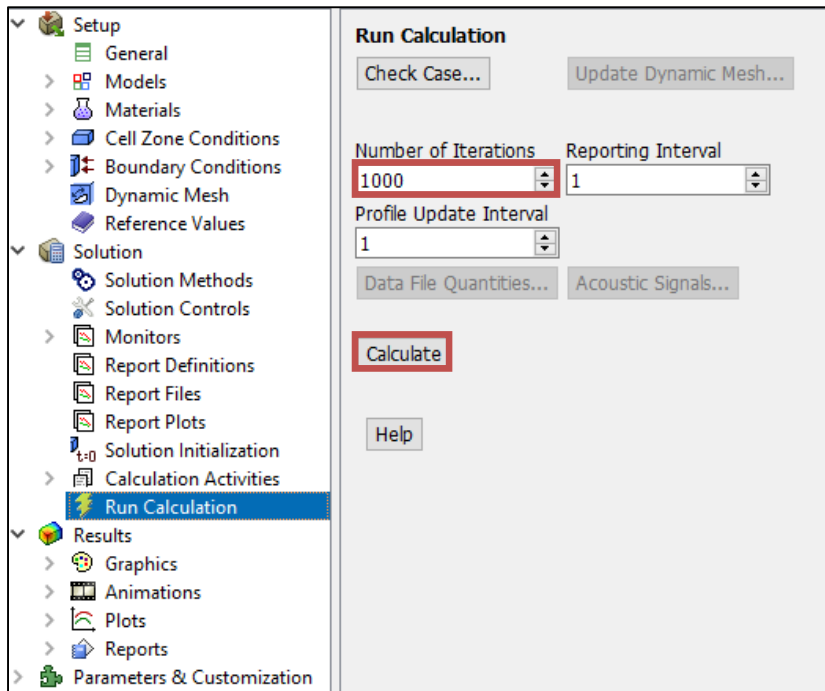
6.16. **Tree > Solution > Solution Initialization.** Change parameters as per below and click **Initialize**.

The screenshot displays the ANSYS Fluent software interface for the 'Solution Initialization' task. On the left, a tree view shows the 'Solution Initialization' option selected and highlighted in blue. The main panel on the right is titled 'Solution Initialization' and contains the following settings:

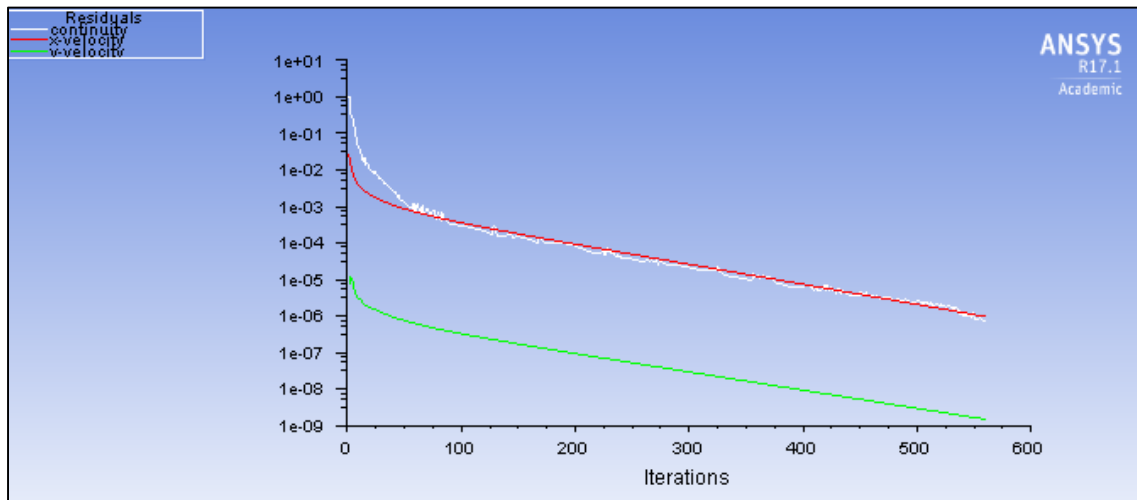
- Initialization Methods:** Standard Initialization (highlighted with a red box), Hybrid Initialization.
- Compute from:** A dropdown menu.
- Reference Frame:** Relative to Cell Zone, Absolute.
- Initial Values:** A section with three input fields:
 - Gauge Pressure (pascal): 0
 - Axial Velocity (m/s): 0.2
 - Radial Velocity (m/s): 0

At the bottom of the panel, there are several buttons: 'Initialize' (highlighted with a red box), 'Reset', 'Patch...', 'Reset DPM Sources', 'Reset Statistics', and 'Help'.

6.17. **Tree > Solution > Run Calculation.** Change **Number of Iterations** to **1000** and click **Calculate**.



6.18. Once the solution converges, click **OK**. (The residuals should be comparable to the ones below.)



NOTE: ANSYS determines when to stop a calculation based on the iteration number and convergent limit you specified. If: 1. the maximum iteration number is reached, but convergent limit is not reached, or 2. convergent limit is satisfied, but maximum iteration number is not reached, ANSYS will terminate the computation.

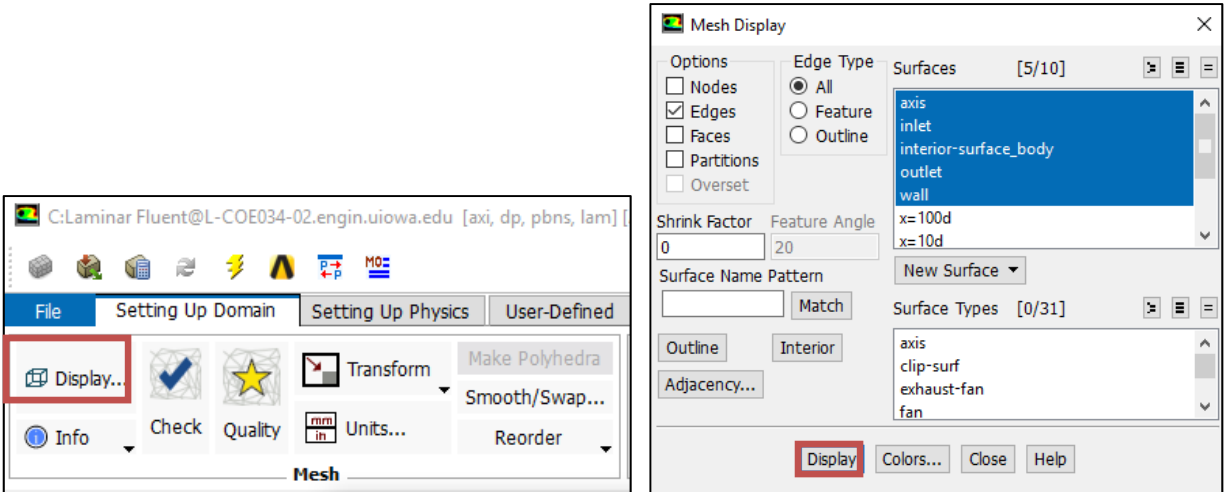
6.19. **File > Save Project.** Save the project

7. Results

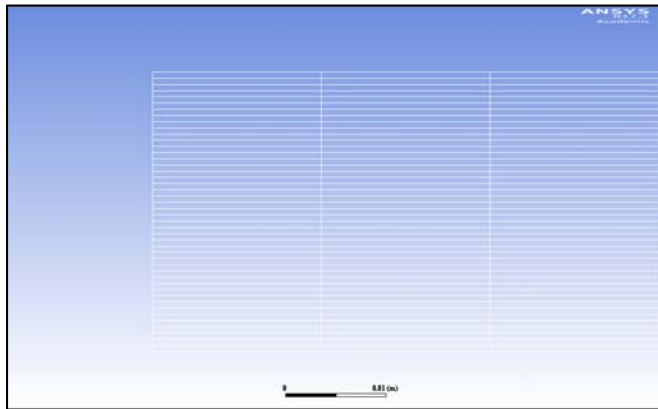
Please read exercises before continuing. This section tells you how to post-process the results.

7.1. Displaying Mesh

Setting Up Domain > Display > Mesh. Select all **Surfaces** you wish to be visible and select **Display** then click **Close**.

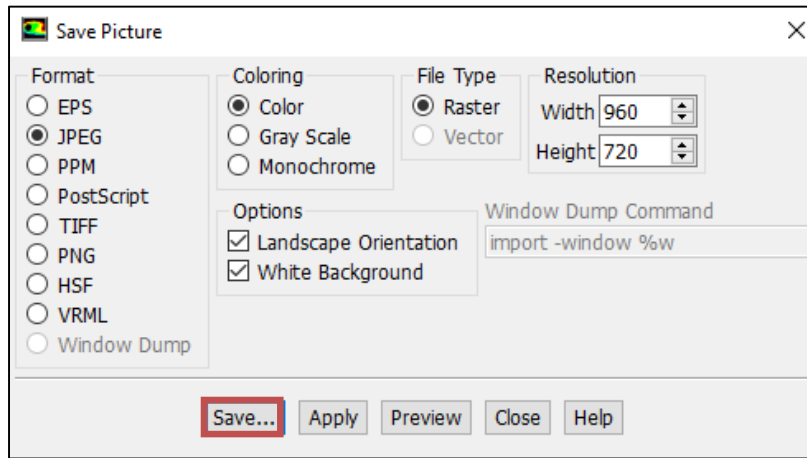


Zoom in to the inlet by using the magnifying glass with a plus sign in the middle of it. The mesh should look like the one below.

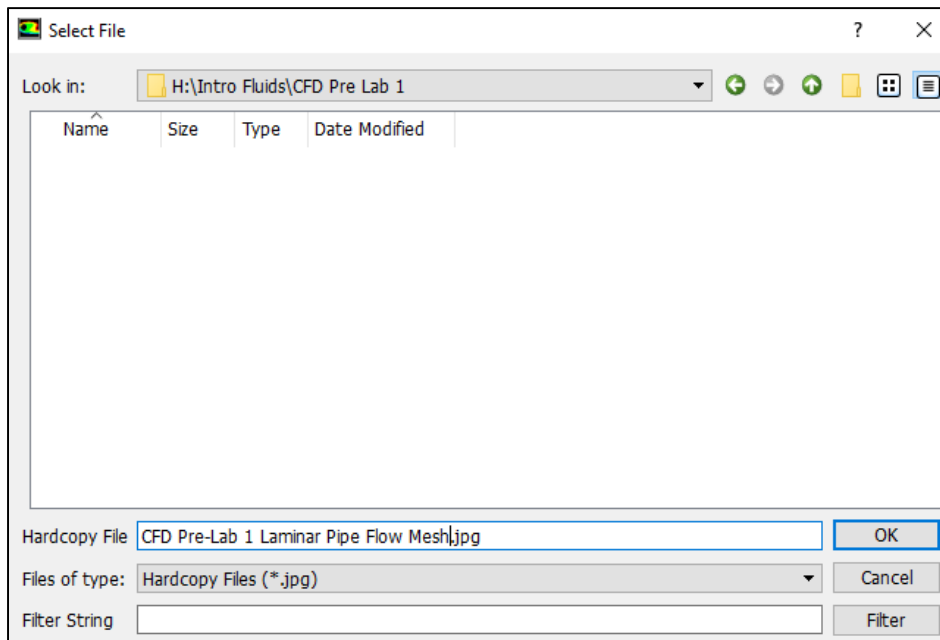


7.2. Saving Pictures

To save a picture of the screen, select **File > Save Picture**. Make sure all the parameters are set similar to the ones below and click **Save** (To preview the picture, before you save click **Preview** in the **Save Picture** window)

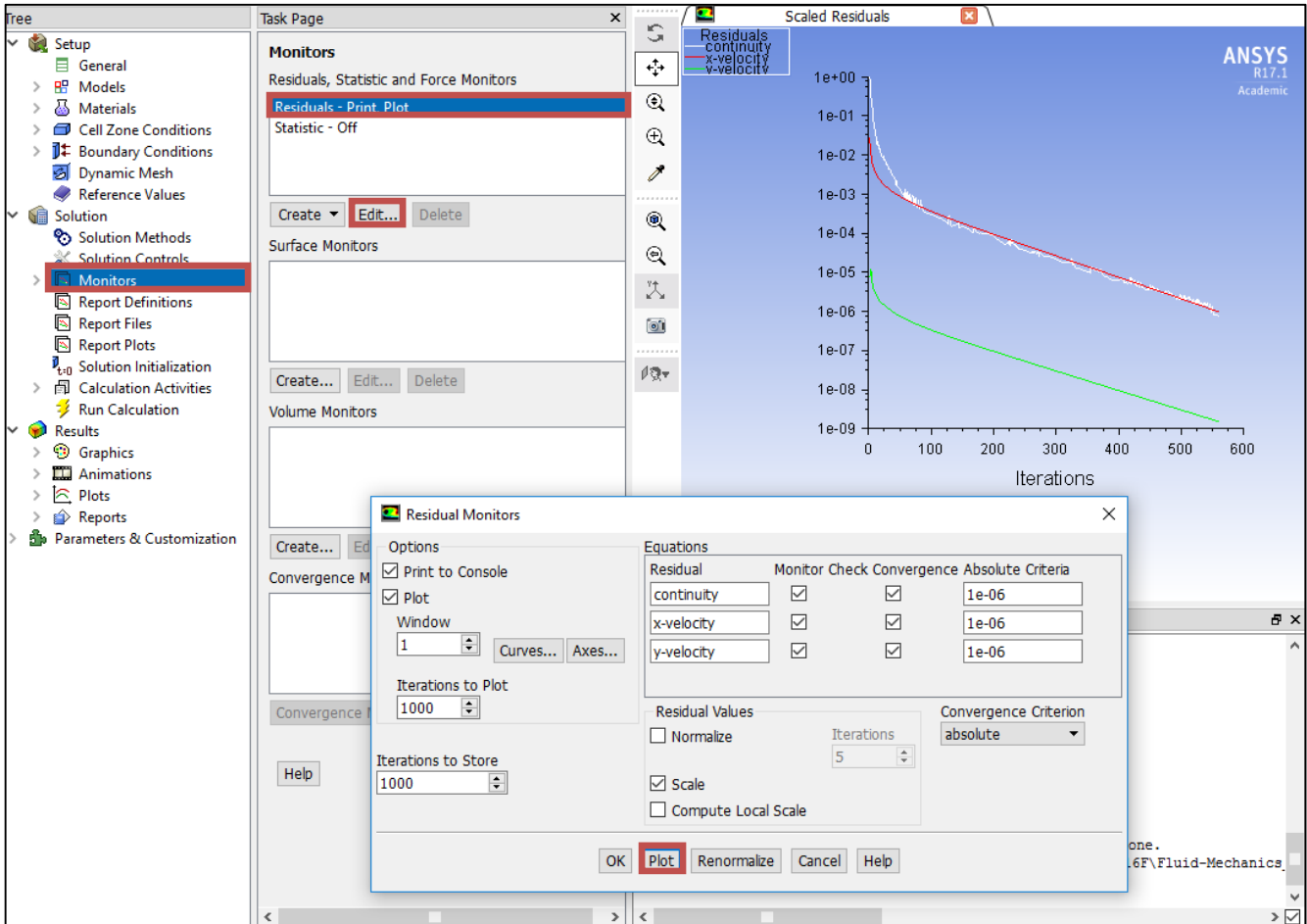


Name the File, navigate to the CFD Pre-Lab 1 file you created and save it in that file. Then close the **Save Picture** window.



7.3. Plotting Residuals

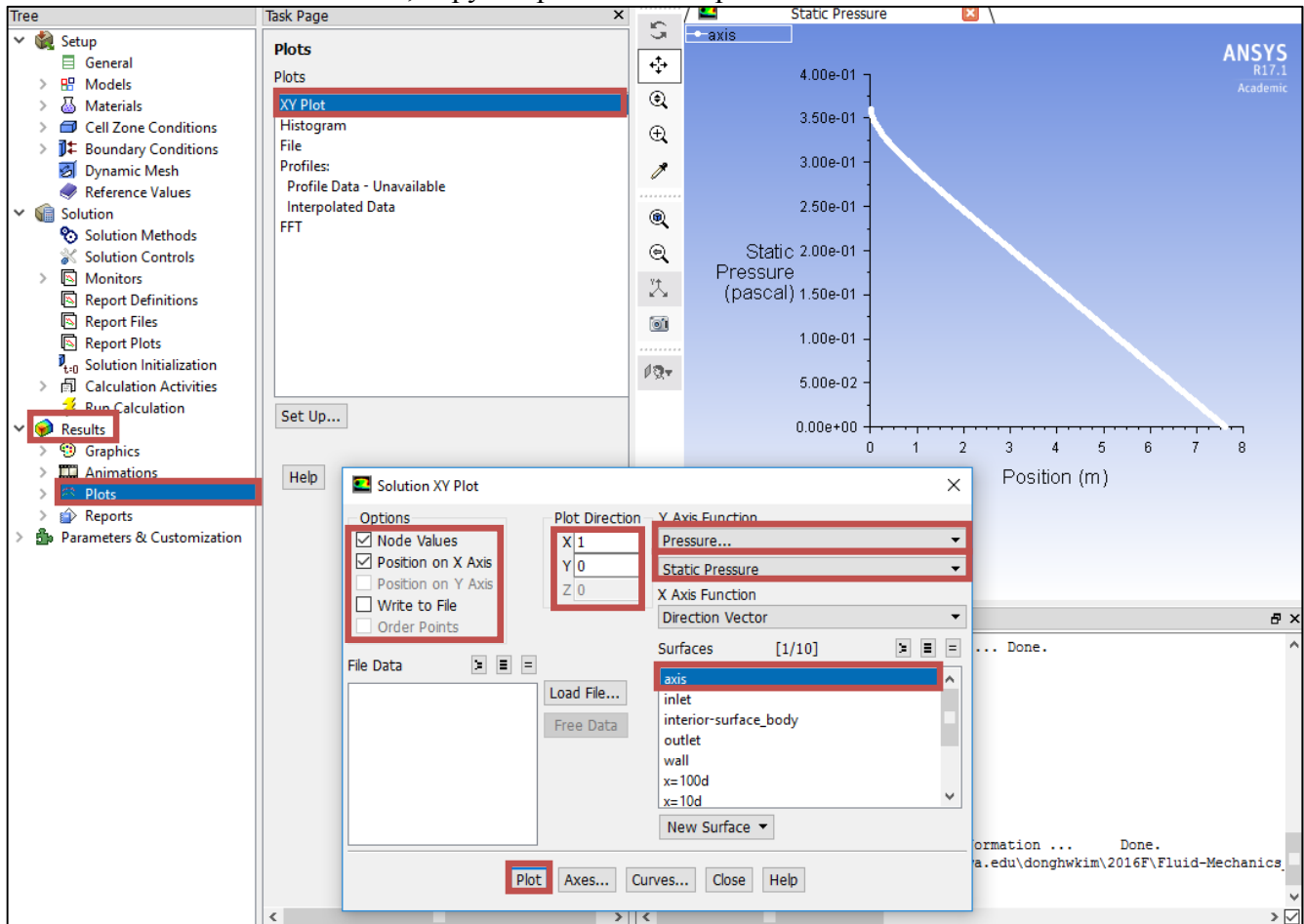
To display the residuals click **Tree > Solution > Monitors**. Right click **Residuals** select **Edit**, click **Plot** then click **Cancel**.



You can save this picture the same way you saved the mesh. Name it “*CFD Pre-Lab 1 Laminar Pipe Flow Residuals History*” and save it to the folder you created on the network drive.

7.4. Plotting Centerline Pressure Distribution

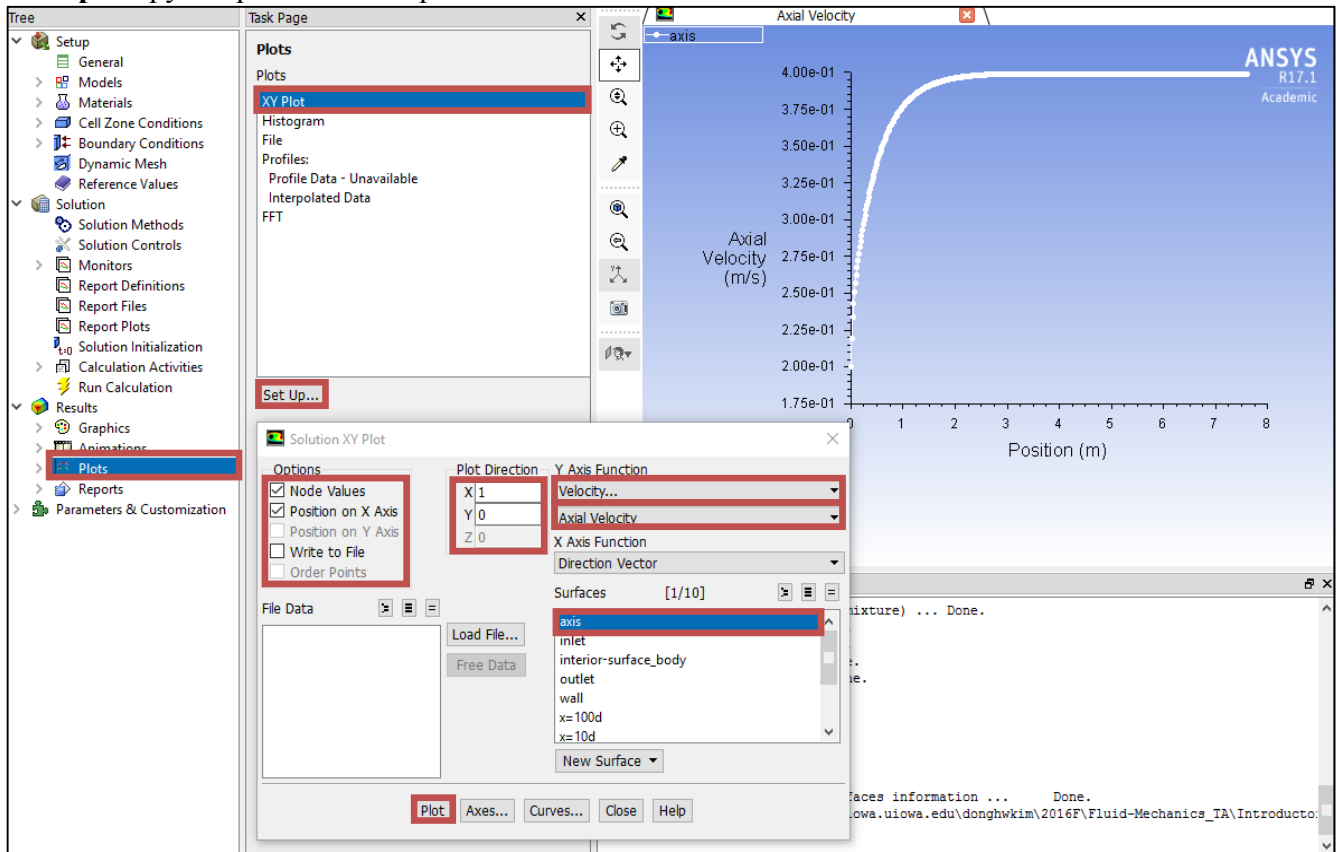
To plot results, click **Tree > Results > Plots**. Right click **XY Plot**, then click **Set Up**. To plot the Centerline Pressure Distribution, copy the parameters as per below and click **Plot**.



Save the picture as you did for the mesh and call it “*CFD Pre-Lab 1 Laminar Pipe Flow Centerline Pressure Distribution*” and save it in the folder you created.

7.5. Plotting Centerline Velocity Distribution

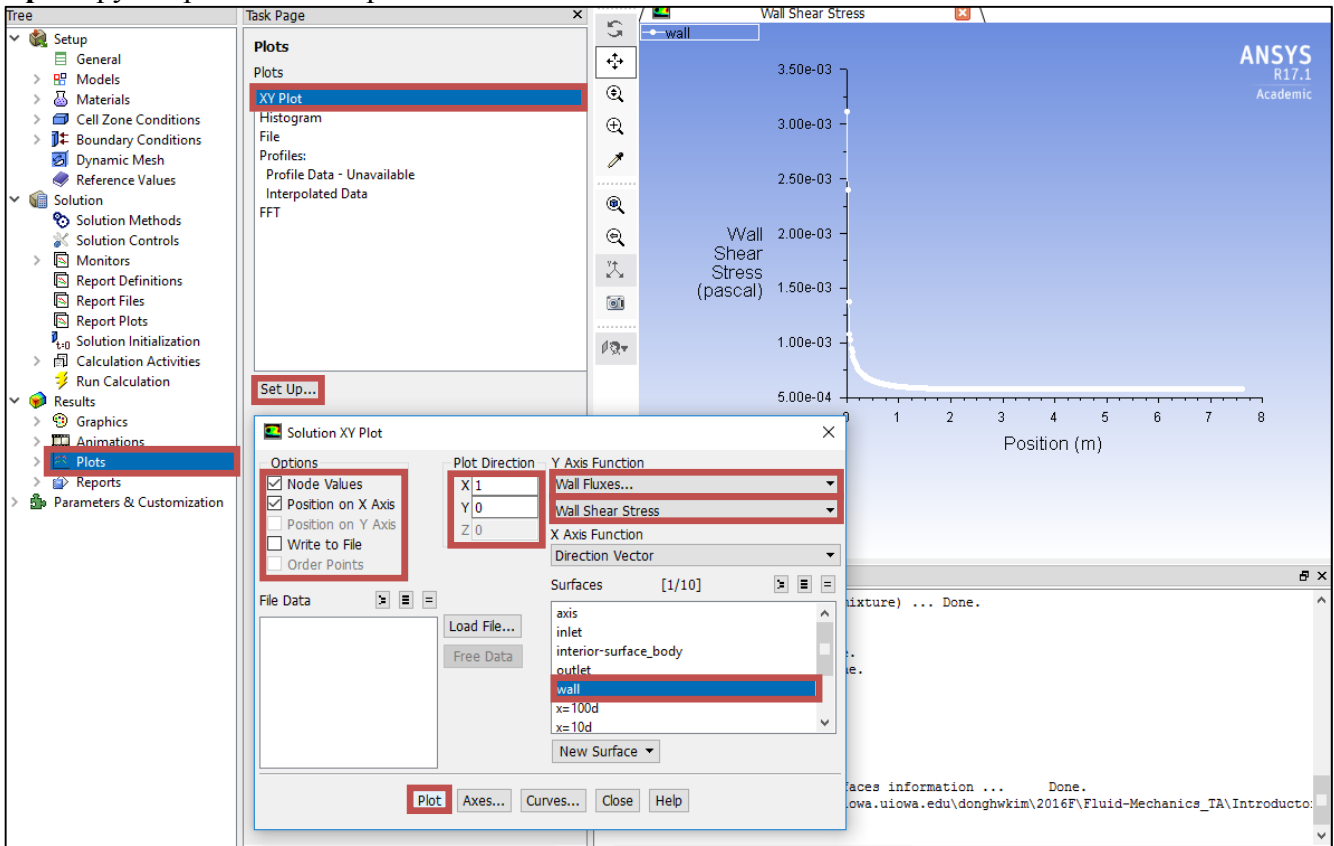
To plot Centerline Velocity Distribution, click **Tree > Results > Plots**. Right click **XY Plot**, then click **Set Up**. Copy the parameters as per below and click **Plot**.



Save the picture as you did for the mesh and call it “*CFD Pre-Lab 1 Laminar Pipe Flow Centerline Velocity Distribution*” and save it in the folder you created.

7.6. Plotting Wall Shear Stress Distribution

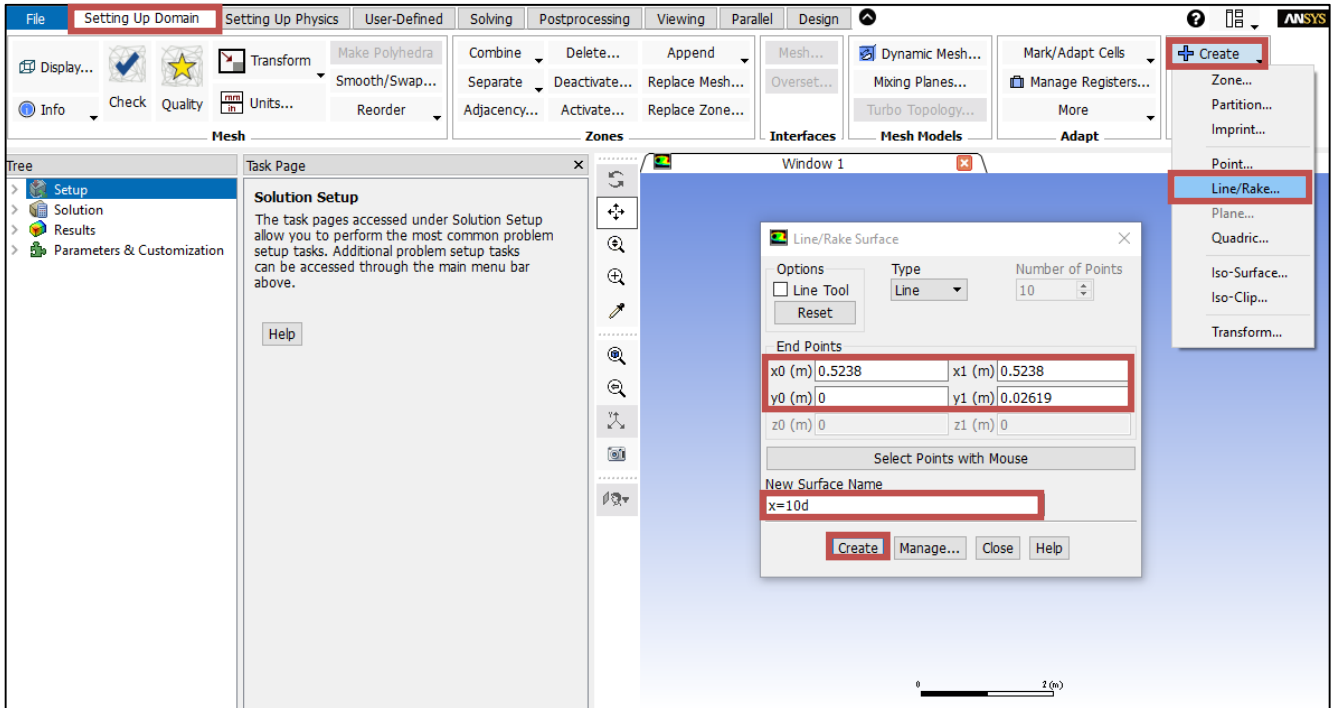
To plot the Wall Shear Stress Distribution, click **Results** > **Plots**. Right click **XY Plot**, then click **Set Up**. Copy the parameters as per below and click **Plot**.



Save the picture as you did for the mesh and call it “*CFD Pre-Lab 1 Laminar Pipe Flow Wall Shear Stress Distribution*” and save it in the folder you created.

7.7. Plotting Profiles of Axial Velocity at All Axial Locations

To plot Profiles of Axial Velocity at All Axial Locations with AFD Data, click **Setting Up Domain > Create > Line/Rake**.

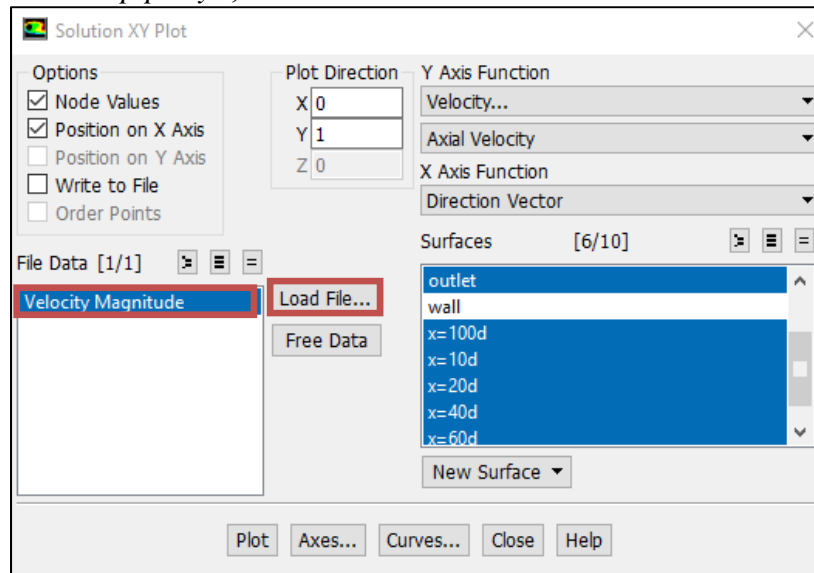


Change x and y values as per above, name the surface, and click **Create**. Repeat this for all lines shown in the table.

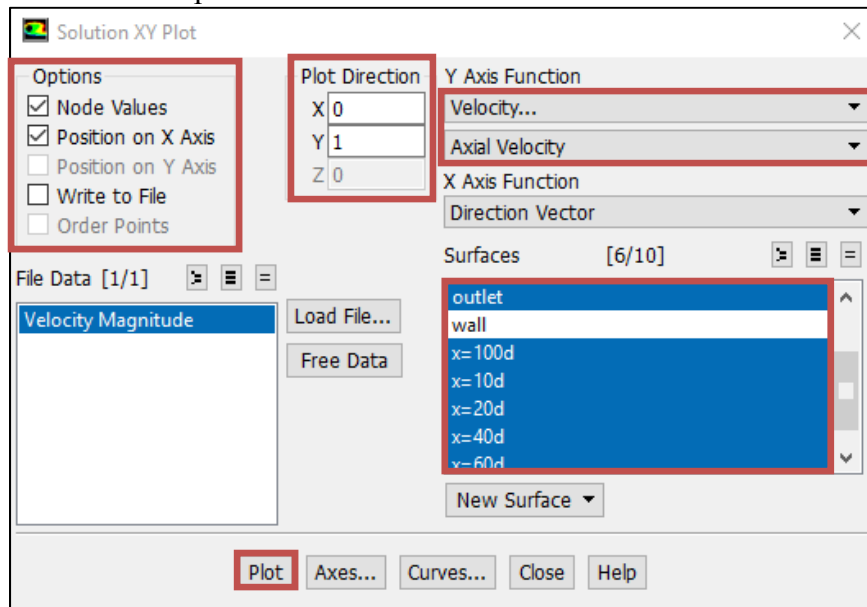
Surface Name	X0	Y0	X1	Y1
x=10d	0.5238	0	0.5238	0.02619
x=20d	1.0476	0	1.0476	0.02619
x=40d	2.0952	0	2.0952	0.02619
x=60d	3.1428	0	3.1428	0.02619
x=100d	5.238	0	5.238	0.02619

When all lines are created, click **Close**.

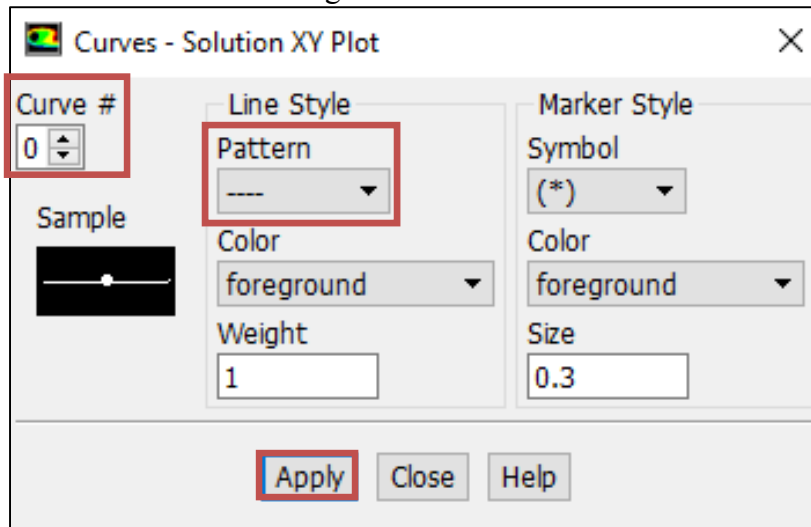
Click **Tree > Results > Plots**. Right click **XY Plot** and click **edit...** Click **Load File** and select “*axialvelocityAFD-laminar-pipe.xy*”, which is downloadable from the class website. Click **OK**.



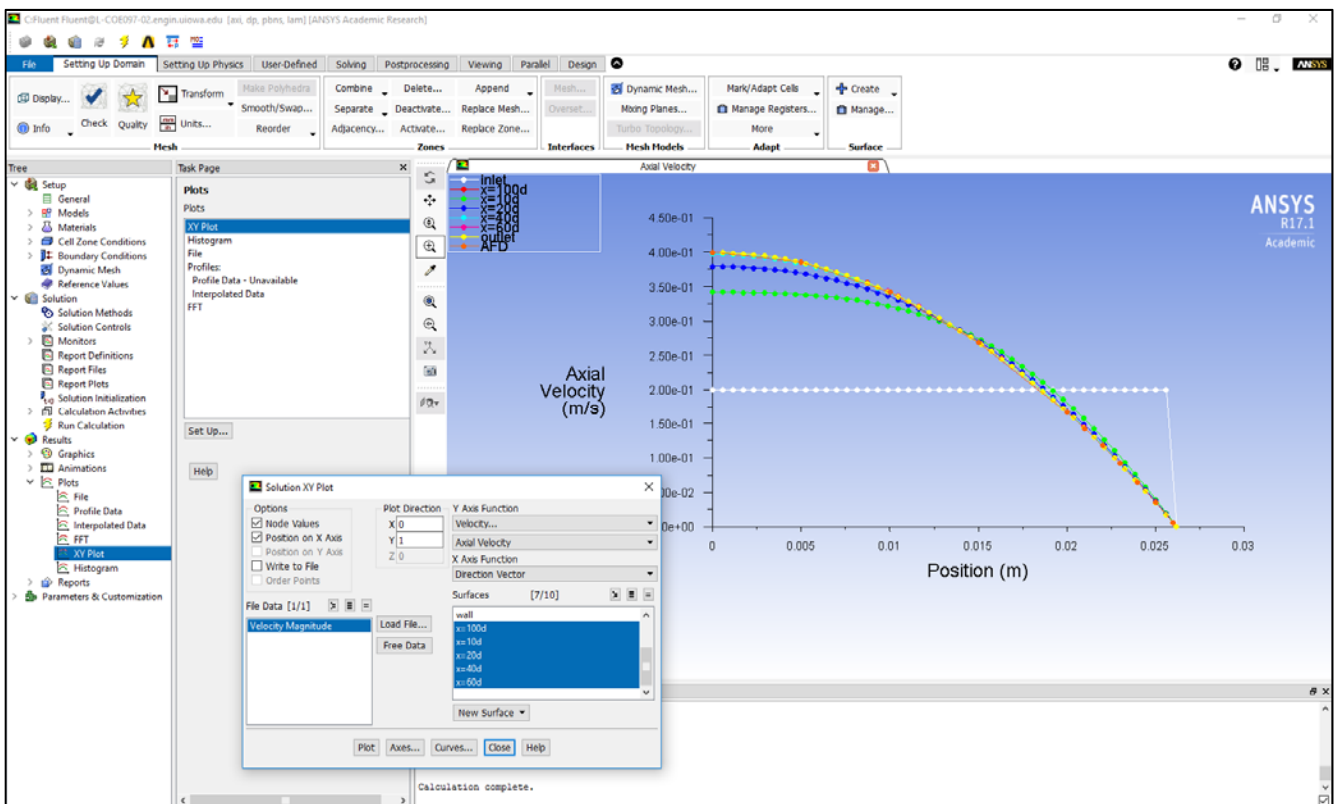
Change Parameters as per below. Make sure to select the inlet and outlet as well.



Click **Curves...** > Change the **Pattern** to the pattern seen below and click **Apply**. Increment the **Curve #** by one and repeat. Do this for curves 0 through 7 then click **Close**.



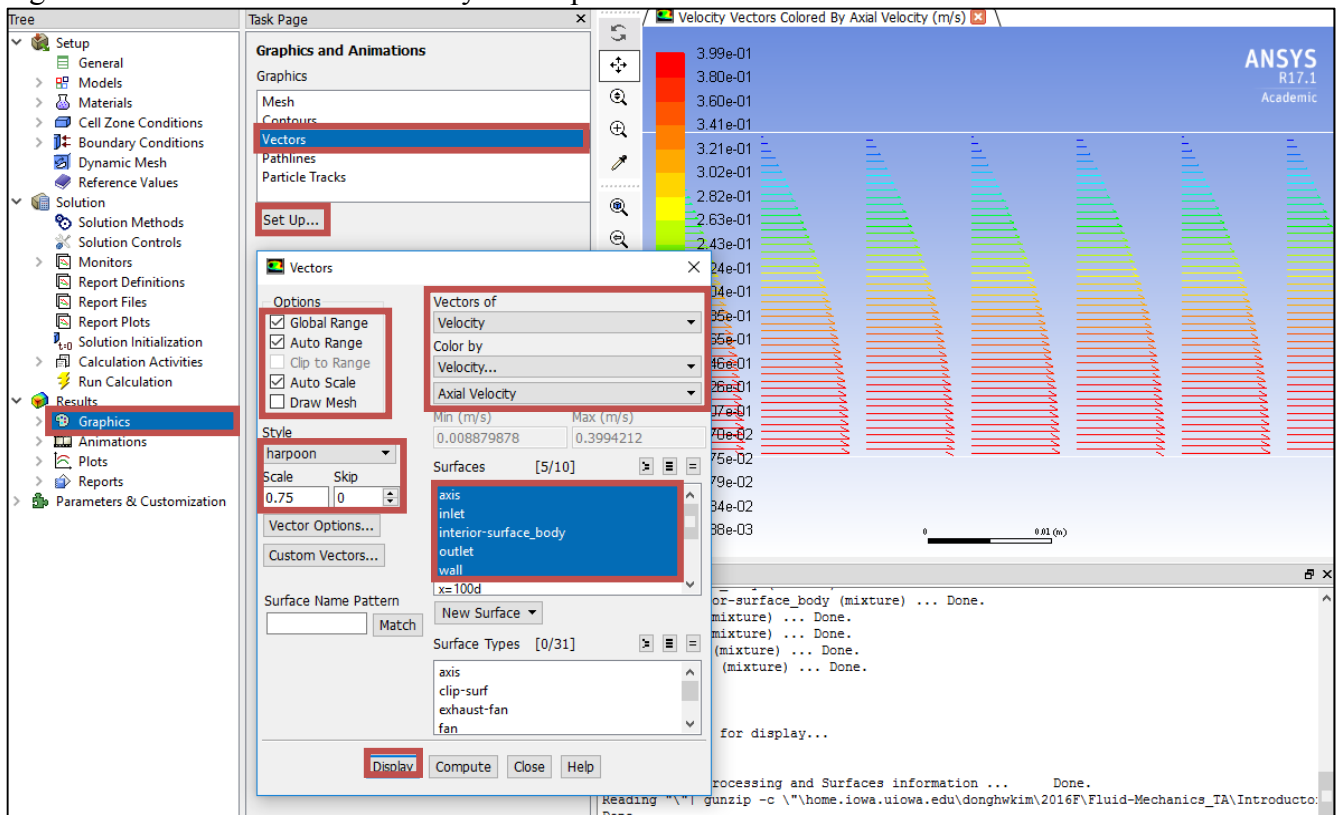
Click **Plot**.



Save the picture as you did for the mesh and call it “*CFD Pre-Lab 1 Laminar Pipe Flow Axial Velocity at All Axial Locations with AFD Data*” and save it in the folder you created. Close the **Solution XY Plot** window.

7.8. Plotting Velocity Vectors

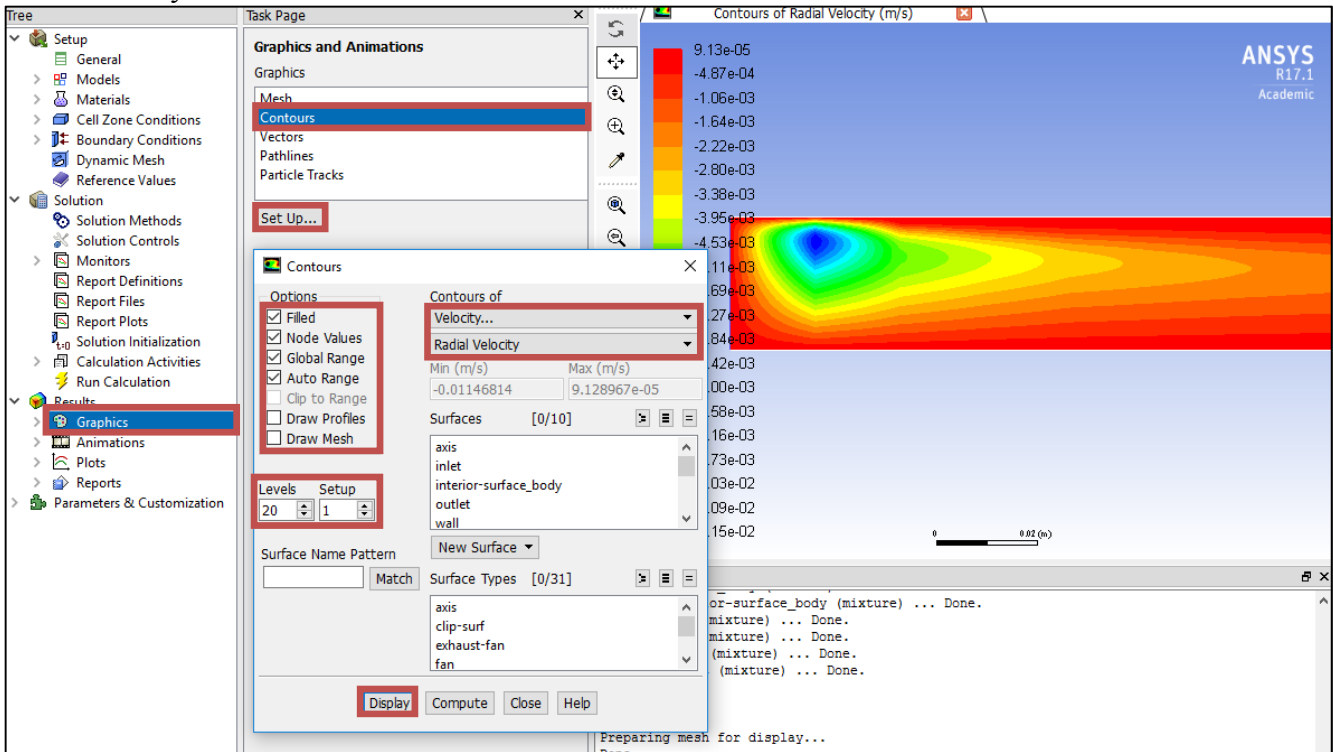
Click **Tree > Results > Graphics > Vectors > Set Up**. To plot the velocity vectors at the region flow begin to becomes fully developed, copy the parameters as per below and click **Display**. Zoom into the region where the flow is almost fully developed.



Save the picture as you did for the mesh and call it “*CFD Pre-Lab 1 Laminar Pipe Flow Velocity Vectors at the Region Flow Begins to Become Fully Developed*” and save it in the folder you created. Close the **Vectors** window.

7.9. Plotting Contours of Radial Velocity

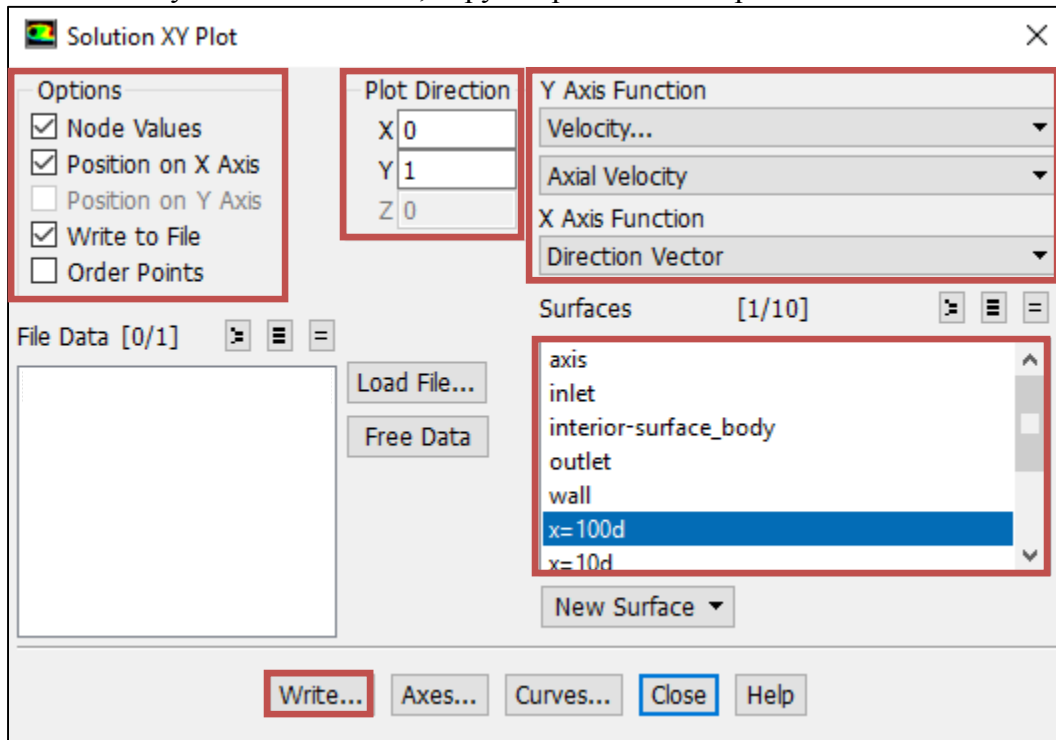
Click **Tree > Results > Graphics > Contours > Set Up...** To plot the Contours of Radial Velocity, copy the parameters as per below and click **Display**. Zoom in to the pipe inlet to see the contours of radial velocity.



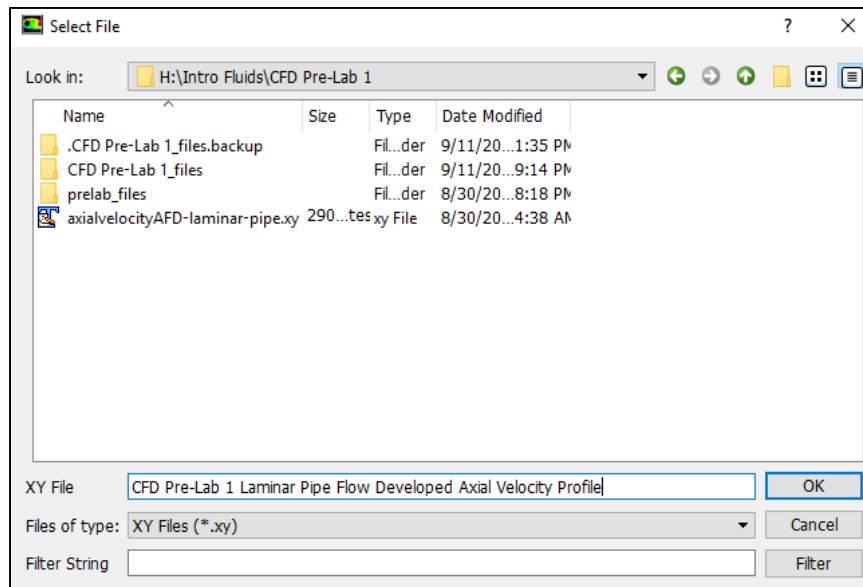
Save the picture as you did for the mesh and call it “*CFD Pre-Lab 1 Contours of Radial Velocity*” and save it in the folder you created. Close the **Contours** window.

7.10. Exporting Axial Velocity Profile at x=100d Location

To export solution value, click **Tree > Results > Plots**. Double click **XY Plot**. To export the Developed Axial Velocity Profile at x=100d, copy the parameters as per below and click **Write...**

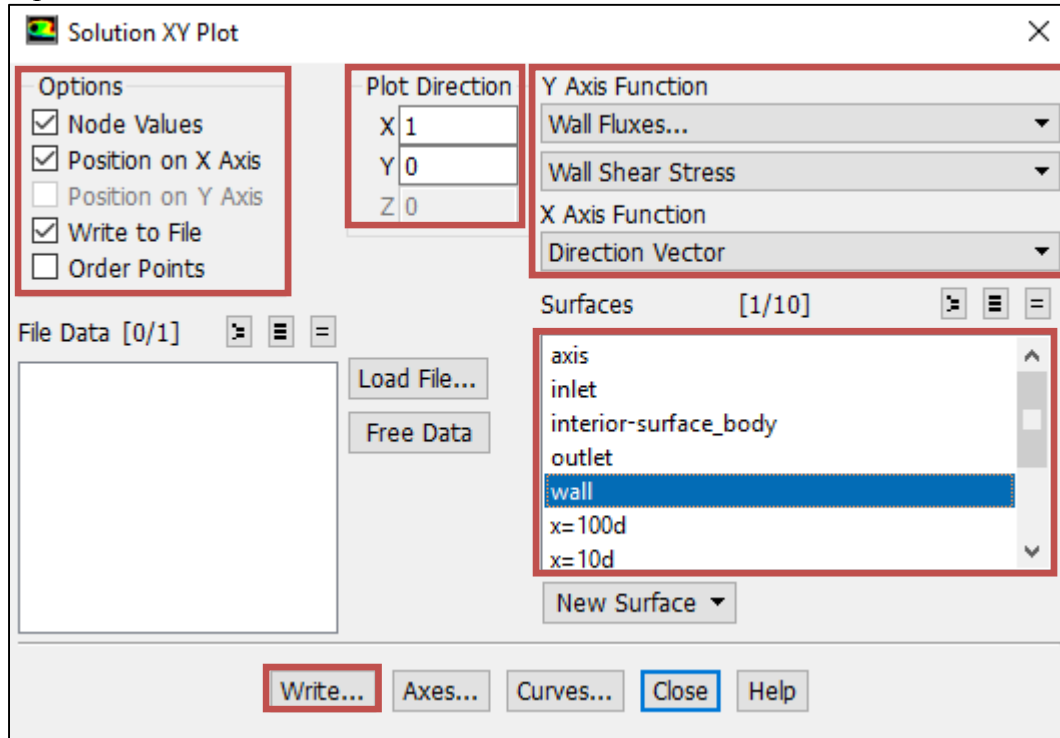


Name the file “*CFD Pre-Lab 1 Laminar Pipe Flow Developed Axial Velocity Profile*” and leave the Files of Type: as XY Files. Click **OK**.



7.11. Exporting Wall Shear Stress Distribution

To export the wall shear stress distribution, click **Tree > Results > Plots**. Double click **XY Plot**. Copy the parameters as per below and click **Write...**

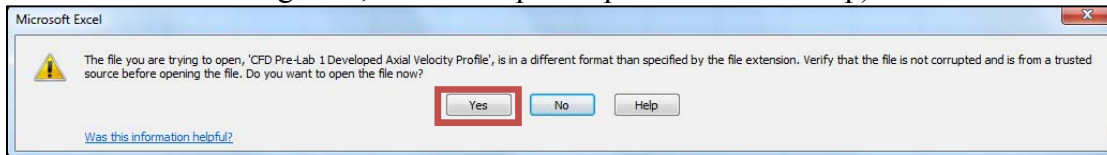


Name the file “*CFD Pre-Lab 1 Laminar Pipe Flow Wall Shear Stress Distribution*” and leave the Files of Type: as XY Files. Click **OK**. Close the Solution XY Plot.

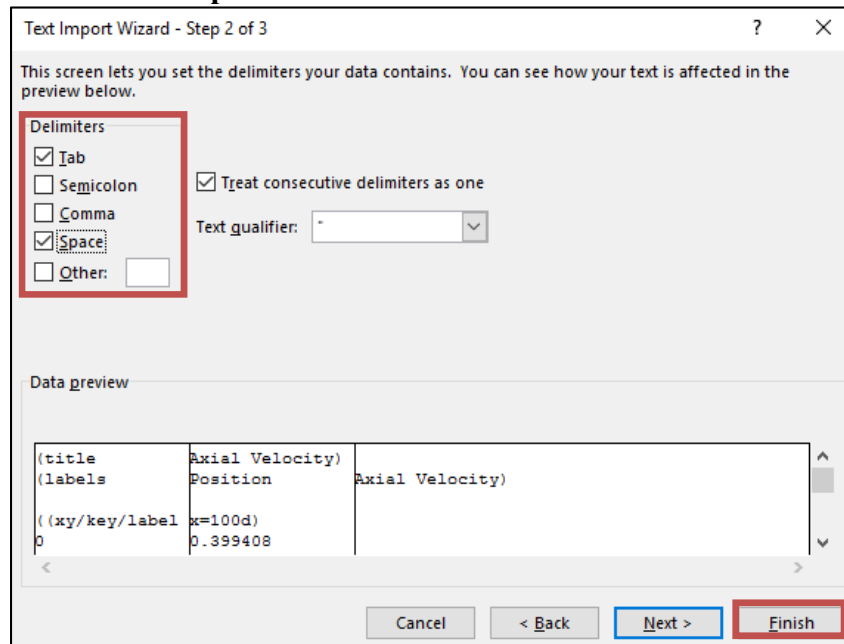
File > Save Project. Save the project and close the **Fluent** window.

7.12. Normalizing Velocity Profile

- Open excel from Start Menu.
- Click **File > Open**, navigate to your folder you created on the network drive.
- Change the file type to all files.
- Select the file CFD Pre-Lab 1 Laminar Pipe Flow Developed Axial Velocity Profile and click **Open**.
- Select **Yes** on the **Microsoft Excel** message. (You might not see this pop-up depending on the version of Excel being used, if not then please proceed to next step)



- Make sure delimited is selected and click **Next**.
- Make sure that **Tab** and **Space Delimiters** are checked and hit **Finish**.



- In Cell C5 enter the formula as seen in the Formula Bar below. Then take the fill handle and drag to the end of the data. This normalizes the velocity profile from the max velocity.

	A	B	C	D	E
1	(title	Axial Velocity)			
2	(labels	Position	Axial Velocity)		
3					
4	((xy/key/1	x=100d)			
5	0	0.399408	1		
6	0.000582	0.399309			
7	0.001164	0.39872			
8	0.001746	0.397739			

- Insert a **Scatter Plot With Smooth Lines and Markers**.
- For the x-axis use the radial position, and for the y-axis use the normalized velocity.
- Name it CFD Velocity Profile (Laminar).
- You can move this plot to a new tab by clicking on the chart **Chart Tools > Design > Move Chart Location > New Sheet > OK**
- Next open the file “*Normalized-velocity-AFD-laminar-pipe.xy*” in TextPad, highlight the data and paste it into your Excel spread sheet next to the CFD velocity profile data.
- Plot this in the same way as the other set on the existing plot and call this “*AFD Velocity Profile (Laminar)*”.
- Create axis titles and make sure the legend is shown. You should move the legend to the bottom of the chart. Call the axes *Normalized Velocity [-]* and *Radial Position [m]*.
- Save this Sheet by selecting File > Save As, name it “*CFD Pre-Lab 1 Developed Axial Velocity Profile*”.

8. Exercises

You must complete all the following assignments and present results in your CFD Lab 1 reports following the CFD Lab Report Instructions.

Simulation of Laminar Pipe Flow

You need to use CFD Lab1 Report Template.doc to save all the figures and data

8.1. Compare CFD with AFD on friction factor

Use the instructions to generate the mesh and setup then iterate the simulation until it converges. Find the **relative error** between AFD friction factor ($Factor_{AFD}=0.097747231$) and friction factor computed by CFD, which is computed by:

$$\frac{Factor_{CFD} - Factor_{AFD}}{Factor_{AFD}} \times 100\%$$

To get the value of $Factor_{CFD}$, you need first write to file the wall Shear Stress Distribution. Then use EXCEL to open the data file and pick the value close to the pipe exit or inside the fully developed region. Next use the equation $C=8*\tau/(\rho*U^2)$ to solve for the Friction Factor. Where C is the friction factor, τ is wall shear stress, ρ is density and U is the inlet velocity.

Figures need to be saved: (1) Residual History (2) Centerline Pressure Distribution (3) Centerline Velocity Distribution (4) Wall Shear Stress Distribution (5) Profiles of Axial Velocity at all Streamwise Locations (inlet, outlet, $x/D=10,20,40,60,100$) with AFD data (6) Contour of Radial Velocity (7) Velocity Vectors in Fully Developed Region.

Data need to be saved: (1) Shear Stress in the Developed Region (2) Developing Length (=Length from the pipe inlet to the start of developed region. Use figure of centerline velocity distribution).

8.2. Normalized developed axial velocity profile

8.2.1. Export the axial velocity profile data at $x=100d$ following the instructions in Step 7.10.

8.2.2. Use EXCEL to open the file you exported and normalize the profile using the centerline velocity magnitude, which is the maximum value on that profile. Plot the normalized velocity profile in EXCEL and paste the figure into your report together with other figures you made in Exercise 8.1.

8.3. Questions need to be answered when writing CFD Lab 1 report

8.3.1. Can you use centerline pressure distribution to determine the “developing length”? Why?

8.3.2. What is the value for radial velocity at developed region?

8.3.3. Summarize your findings in CFD Lab report and try to relate them to your classroom lectures or textbooks.